



TYPE-CERTIFICATE DATA SHEET

No. EASA.A.155

**for
Dassault Falcon 7X**

Type Certificate Holder:

DASSAULT AVIATION
9 Rond Point Marcel Dassault
75008 PARIS
France

Airworthiness Category: Large Aeroplanes

For Model: Falcon 7X

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I. General

- | | |
|---|---|
| 1. Type: | Falcon 7X |
| 1.1 Model: | Falcon 7X |
| 2. Performance Class: | A |
| 3. Certifying Authority: | European Aviation Safety Agency
Post office box 10 12 53
D-50452 Cologne
Germany |
| 4. Manufacturer: | Dassault Aviation
9 Rond Point Marcel Dassault
75008 PARIS
France |
| 5. EASA Certification Application Date: | 26 May 2002 |
| 6. EASA Type Certification Date: | 27 April 2007 |

II. Certification Basis

- | | |
|---|---|
| 1. Reference Date for determining the applicable airworthiness requirements | Same as EASA certification application date |
| 2. Reference Date for determining the applicable operational suitability requirements | 17 February 2015 |
| 3. EASA Airworthiness Requirements | |
| 3.A Airworthiness requirements for S/N 0001 to S/N 0400 (modification M1000 not included) | |
| JAR 1 at change 5 plus orange papers 1/97/1 and 1/99/1 | |
| JAR 25 at change 15, effective 01 August 2000 with the following additions: | |
| a) JAR 25 paragraphs at amendment 16 | |
| JAR 25.331(c)(2) | Symmetric manoeuvring conditions |
| JAR 25.335(b)(2) | Design airspeeds |
| JAR 25.337(d) | Limit manoeuvring load factors |
| JAR 25.391 | Control surface loads: general |
| JAR 25.395(b) | Control system |
| JAR 25.415 | Ground gust conditions |
| JAR 25.491 | Taxi, takeoff and landing roll |
| JAR 25.493(c) | Braked roll conditions |
| JAR 25.605(a) | Fabrication methods |
| JAR 25.731(d)(e) | Wheels |
| JAR 25.735 | Brakes |
| JAR 25.904 | Automatic takeoff thrust control system (ATTCS) |
| JAR 25.933 | Reversing systems |
| JAR 25.939(d) | Turbine engine operating characteristics |
| JAR 25.951(d) | Fuel system - General |
| JAR 25.952 | Fuel system analysis and test |
| JAR 25.954 | Fuel system lightning protection |
| JAR 25.961(a) | Fuel system hot weather operation |
| JAR 25.967 | Fuel tank installations |
| JAR 25.975(a)(5) | Fuel tank vents |

JAR 25.981	Fuel tank temperature
JAR 25.993 (c)	Fuel system lines and fittings
JAR 25.994	Fuel system components
JAR 25.997	Fuel strainer or filter
JAR 25.1013	Oil tanks
JAR 25.1015	Oil tank tests
JAR 25.1019	Oil strainer or filter
JAR 25.1145(c)	Ignition switches
JAR 25.1301(d)	Function and installation
JAR 25.1305(a)(3),(a)(9),(c)(5),(c)(6),(c)(7),(c)(8),(d)(2)	Powerplant instruments
JAR 25.1309	Equipment, systems and installations
JAR 25.1310	Power source capacity and distribution
JAR 25.1323	Airspeed indicating system
JAR 25.1351 (b)(6)	Electrical systems and equipment - General
JAR 25.1435	Hydraulic systems
Appendix H §H25.3	Instruction for Continued Airworthiness

- b) CS 25 paragraphs at amendment 12 for aircraft fitted with emergency exit pictograms (modification M-OPT0760)
- | | |
|------------------------------------|------------------------|
| CS 25.811(g) | Emergency Exit Marking |
| CS 25.812(b)(1)
and related AMC | Emergency Lighting |

JAR AWO at change 2, effective 01 August 1996,

3.B Airworthiness requirements for S/N 0401 and ongoing (modification M1000 included) ^{NOTE 3}

Certification Specification 25, Amendment 11, dated 04 July 2011, except the following paragraphs for which an earlier amendment is accepted:

- a) JAR 25 paragraphs at change 15

25.21(b) (as amended by SC B-01), 25.33, 25.103 (as amended by SC B-01, B-08), 25.171 (as amended by B-03), 25.173 (as amended by B-03), 25.175 (as amended by B-03), 25.177 (as amended by B-03), 25.201 (as amended by SC B-01, B-08), 25.203 (as amended by SC B-01, B-08), 25.207(c)(d)(e) (as amended by SC B-01), 25.251(a)(b) (as amended by ESF C-09), 25.305(e)(f) (as amended by ESF C-09), 25.335(b)(1) (as amended by SC C-03), 25.349(a) (as amended by SC C-01), 25.351(a)(1) (as amended by SC C-01), 25.397(c) (as amended by SC C-02), 25.399, 25.405, 25.427(d) via SC C-09, 25.497, 25.562 (as amended by SC D-26 for seats with inflatable restraints), 25.629(a)(b)(1)(2)(c)(d)(e) (as amended by SC C-09), 25.671 (as amended by SC D-05, D-02), 25.679, 25.681, 25.685, 25.689, 25.693, 25.699 (as amended by ESF D-19), 25X745 (as amended by SC D-07), 25.771, 25.772, 25.773, 25.777 (as amended by SC B-02), 25.779, 25.781, 25.783 (as amended by SC D-22), 25.785 (as amended by Dev. D-18), 25.787, 25.789, 25.791, 25.793, 25.807 (as amended by SC D-22), 25.809 (as amended by SC D-22), 25.810 (as amended by SC D-22), 25.811 (as amended by ESF D-12, D-13), 25.813 (as amended by Dev. D-14), 25.815, 25.817, 25.819, 25.820 (as amended by SC D-22), 25.831(a) (as amended by SC D-09 and ESF D-15), 25.833, 25.851, 25.853 (as amended by SC D-11), 25.854, 25.855 (as amended by SC D-11), 25.857, 25.858, 25.859, 25.865, 25.867, 25.869, 25.871, 25.875, 25X899, 25.901, 25.905, 25.925, 25.929, 25.933(a) (as amended by SC E-04), 25.934, 25.937, 25.941, 25.943, 25.945, 25.953, 25.977, 25.979, 25.991, 25.1017, 25.1021, 25.1023, 25.1025, 25.1027, 25.1043, 25.1045, 25.1093 (as amended by ESF E-08), 25.1103, 25.1121, 25.1123, 25.1141, 25.1149, 25.1153, 25.1155, 25.1161, 25.1163, 25.1165, 25.1167, 25.1181, 25.1182, 25.1183, 25.1185, 25.1187, 25.1189, 25.1191, 25.1195 (as amended by ESF E-02), 25.1197, 25.1199, 25.1201, 25.1203, 25.1303, 25.1307, 25.1321, 25.1322 (as amended by ESF F-41), 25.1325, 25.1326, 25.1327, 25X1328, 25.1329 (as amended by ESF F-37), 25.1331, 25.1333, 25.1335 (as amended by ESF F-37), 25.1355, 25.1357 (as amended by ESF F-22), 25X1360, 25X1362, 25.1363, 25.1381, 25.1383, 25.1411, 25.1415, 25.1421, 25.1423, 25.1433, 25.1439, 25.1443, 25.1447, 25.1449, 25.1450, 25.1453, 25.1455, 25.1457, 25.1459 (as amended by ESF F-35), 25.1461, 25X1499, 25X1516, 25.1522, 25.1523, 25.1531, 25.1545, 25.1547, 25.1549 (as amended by ESF E-10), 15.1551, 25.1553, 25.1557, 25.1561, 25.1563

Subpart J 25A901 to 25A1583, Appendix A, Appendix D, Appendix F, Appendix I, Appendix J

- b) JAR 25 paragraphs at amendment 16
25.904, 25.907, 25.933(b), 25.981 (as amended by SC E-01), 25.1013, 25.1015, 25.1019, 25.1145, 25.1305, 25.1309, 25.1310, 25.1323 (as amended by SC B-01)
- c) CS 25 paragraphs at amendment 2
25.105(a), 25.111(c), 25.121(b)(c)
- d) CS 25 paragraphs at amendment 4
25.611, 25.1301, 25.1353, 25.1529, Appendix H
- e) Amended CS 25 amendment 11 paragraphs
25.143 (supplemented by SC B-02, B-04, B-05), 25.145(a)(b) (as amended by SC B-01), 25.207 (as amended by SC B-08), 25.331 (as amended by SC C-01), 25.831 (as amended by SC D-09), 25.841 (as amended by SC D-09), 25.903(c) (as amended by SC E-05), 25.963 (as amended by SC C-06), 25.1431 (as amended by SC F-06)
- f) CS 25 paragraphs that are not applicable
25.795, 25.1302, 25.1365, 25.1535, Subpart H 25.1701 to 25.1731, Appendix M, Appendix N

All Weather Operations: JAR AWO change 2 (as amended by ESF K-01, K-02)

3.1. Special Conditions

- B-01 Stalling and scheduled operating speeds
- B-02 Motion and effects of cockpit controls
- B-03 Static directional, lateral and longitudinal stability and low energy awareness
- B-04 Flight envelope protection
- B-05 Normal load factor limiting system
- C-01 Design maneuver requirements
- C-02 Limit forces and torque
- C-03 Design dive speed Vd
- C-05 Interaction of systems and structure (superseded by CS 25 amdt 11 in case of modification M1000 installation)
- C-06 Fuel tank crashworthiness
- D-02 Electronic flight control unusual features
- D-05 Flight controls - Harmonised 25.671
- D-07 Nose wheel steering - Towbarless towing
- D-09 Airworthiness standards for subsonic aeroplanes to be operated above 41 000 ft
- D-11 Fire protection of thermal and acoustic insulation material
- D-22 Fuselage doors
- E-01 Fuel tank safety
- E-04 Reversing system requirements
- E-05 Sustained engine imbalance
- F-06 Protection from effects from HIRF
- F-24 Human factors aspects of flight deck design

Special Conditions related to certain installed modifications

Steep approach capability modification M0194

B(SAL)-06 Steep Approach and Landing

Shower installation modification M-OPT0459

D-25 Shower/bathroom Installation

Inflatable Restraint modification M-OPT0686

D-26 Seats with Inflatable Restraints

Head-up Guidance System modification M-OPT0002

F-36 Head-up Guidance System

Crew Rest Area modification M-OPT0359

F-44 Installation of Crew rest area

EFVS System modification M-OPT0017

F-47 Enhanced Flight Vision System (EFVS) with Ops credit

Avionics EASy II modification M1122

F-51 Data Link Services for the Single European Sky

F-52 Flight Recorders including Data Link Recording

Avionics EASy III modification M1254

F-65 Data Link Services

Fuselage stretch modification M1000 (for S/N 0401 and ongoing)^{NOTE 3}

B-08 Stalling and scheduled operating speeds in Icing Conditions and Flight in Icing Conditions during Take-off

C-06 Fuel tank crashworthiness (updated with new issue)

Inconsistencies clarification for the modification M1000 certification basis

Subpart B certification basis induces the following inconsistencies:

- CRI B-01 refers to paragraph 25.143(g) and 25.207(f) which respectively become 25.143(h) and 25.207(g) at applicable amendment 11
- CRI B-02 refers to paragraph 25.143(c) which becomes 25.143(d) at applicable amendment 11.
- CRI B-02 introduced new paragraph 25.143(j) titled "Pilot Strength" while 25.143(j) has been introduced in CS 25 Amdt 3 with a different purpose dealing with "Flight in icing conditions before activation of the ice protection system"
- CRI B-03 refers to paragraph 25.143(g) which becomes 25.143(h) at applicable amendment 11
- CRI B-04 introduced new paragraph 25.143(h) titled "Flight envelope protection" while 25.143(h) has been introduced in CS 25 Amdt 3 with a different purpose dealing with "Manoeuvring capabilities",
- CRI B-05 introduced new paragraph 25.143(i) titled "Normal load factor limiting system" while 25.143(i) has been introduced in CS 25 Amdt 3 with a different purpose dealing with "Compliance in icing conditions".
- At Amdt 3, § 25.21(g)(1) requires that all Subpart B paragraphs be demonstrated in icing condition with ice accretion using Appendix C. There is no reversion for 25.21(g)(1), but a for §25.105(a), 25.111(c), 25.121(b)(c) for which CRI B-08 criteria are used for ice accretion.

3.2. Reserved

3.3. Deviations

D-14 Door between passenger compartments

D-18 Personal injury criteria of dynamic testing of side facing sofa

Deviations related to certain installed modifications

Fuselage stretch modification M1000

D-31 Width of Aisle

3.4. Equivalent Safety Findings

- C-09 JAR 25.251, 25.305 and 25.629 - Vibration, buffet and aeroelastic stability requirements
- C-12 JAR 25.361 - Engine failure loads (superseded by CS 25 amdt 11 in case of modification M1000 installation)
- C-15 JAR 25.341, 25.343(b), 25.345(c), 25.371, 25.373(a), 25.391, 25.1517 - Gust and continuous turbulence (superseded by CS 25 amdt 11 in case of modification M1000 installation)
- C-16 JAR 25.963(g) - Fuel tank access cover (superseded by CS 25 amdt 11 in case of modification M1000 installation)
- D-12 JAR 25.811(d)(1) and (d)(2) - Emergency exit locator sign used also as marking sign – cabin without divider
- D-13 JAR 25.811(d)(1) and (d)(3) - Emergency exit locator sign used also as marking sign – cabin with divider
- D-15 JAR 25.831(a) - Packs-off take off
- D-19 JAR 25.699(b) - Lift and drag device indicator
- E-02 JAR 25.865, 25.1181, 25.1195, 25.1203 - Engine fire protection in designated fire zones
- E-08 JAR 25.1093(b) - Falling and blowing snow
- E-10 JAR 25.1549 - Powerplant instruments – colour markings
- E-12 JAR 25.971 - Fuel tank sump (superseded by design change in case of modification M1000 installation)
- F-22 JAR 25.1357(e), 25.1309 - Honeywell PRIMUS EPIC Integrated Modular Avionics system (compliance with requirements for individual circuit protection)
- F-35 JAR 1459 (a)(2) - Use of IRS for DFDR vertical acceleration
- F-37 JAR 25.1329, JAR 25.1335 - Revisions to JAR 25.1329 and 25.1335 resulting from Flight Guidance Systems Harmonisation
- F-41 JAR 25.1322 - CAS window red message line space
- G-01 JAR 25X1591 - Operation on contaminated runways (superseded by CS 25 amdt 11 in case of modification M1000 installation)
- K-01 Revisions to JAR AWO resulting from JAR/FAR 25.1329 Harmonisation
- K-02 Revisions to JAR AWO paragraphs resulting from JAA/FAA Harmonisation

Equivalent Safety Finding for aircraft equipped with modification M-OPT0649

D-27 Table Obstruction to Type III Emergency Exits

3.5. Environmental Protection Requirements

Noise level (for further details see TCDSN EASA.A.155):

- Chapter 1 of ICAO Annex 16 Volume 1 Chapter 4 Amdt 8. (S/N 0001 to 0400, modification M1000 not included)
- Chapter 1 of ICAO, Annex 16, Volume 1, amendment 9, Chapter 4 and CS 36, amendment 2 (S/N 0401 and ongoing, modification M1000 included)^{NOTE 3}

Fuel venting and exhaust emissions:

- Chapter 2 of ICAO Annex 16 Volume 2 Part II and Part III Chapter 2, Amdt 4 applicable on November 4, 1999. (S/N 0001 to 0400, modification M1000 not included)
- Chapter 2 of ICAO Annex 16 Volume II, amendment 6, Part II to the Chicago Convention for the prevention of intentional fuel venting and as implemented in Decision No. 2003/3/RM of the Executive Director of the Agency dated 17 October 2003, on certification specifications providing for acceptable means of compliance for aircraft engine emissions and fuel venting « CS-34, initial issue » (S/N 0401 and ongoing, modification M1000 included)^{NOTE 3}

4. Operational Suitability Requirements

4.1 Master Minimum Equipment List (MMEL)

JAR-MMEL/MEL Subpart A (General) and Subpart B (MMEL) at Amdt 1

4.2 Flight Crew Data (FCD)

CS-FCD, Initial Issue dated 31 January 2014

4.3 Cabin Crew Data (CCD)

This OSD is not applicable since the maximum passenger configuration is below 20.

4.4 Simulator Data (SIMD)

CS-SIMD, Initial Issue dated 2 December 2014 (applicable to S/N 0401 and ongoing / modification M1000 included only) ^{NOTE 3}

4.5 Maintenance Certifying Staff Data (MCSD)

Special Condition A-MCSD-01 "Certification Requirements for the OSD element Maintenance Certifying Staff (MCS) of the aircraft model Dassault Falcon 7X fitted with M1000 design change".

III. Technical Characteristics and Operational Limitations

1. Type Design Definition: The Type Design aircraft configuration is the F7TC version stored in an electronic format under the virtual product management tool ENOVIA/VPM©. The repository of the ENOVIA/VPM© database is located in Dassault Aviation facilities.
2. Description: The Falcon 7X is a maximum 22 occupants including a minimum crew of two, tri-jet, long range, large aeroplane category. It has a low positioned, high swept wing, mid-height horizontal stabilizer and tricycle landing gear. Flight controls are fly-by-wire. Three Pratt & Whitney Canada PW307A (PW307D in case of modification M1000 installed) engines are rear mounted, two on side of fuselage and one in center position.
3. Equipment: The F7TC version referenced under III.1. contains also the type design list of equipment.
4. Dimensions

	Without modification M1000 included (S/N 0001 to 0400)	With modification M1000 included (S/N 0401 and ongoing) ^{NOTE 3}
Length	23.38 m	24,46 m
Span	26.21 m	26,28 m
Height	7.93 m	7.93 m
Gross wing area	70.7 m ²	70.7 m ²

5. Engines

		Three Pratt & Whiney Canada Corp. Turbofan Engines refer to EASA Data Sheet IM.E.035	
		Model PW307A	Model PW307D ^{NOTE 3} (with modification M1000 installed)
Engine Limits Static thrust, standard day, sea level	Takeoff (5 min., Normal All Engines Operating)	28.49 kN (6405 lbs)	29,91 kN (6725 lbs)
	Maximum continuous	28.49 kN (6405 lbs)	29,91 kN (6725 lbs)
Engine Limits Maximum permissible engine rotor operating speeds	N1 (Fan) steady state Take-off / Maximum continuous	101% r.p.m. (100% = 11000 r.p.m)	101% r.p.m. (100% = 11000 r.p.m)
	N2 (Gas Gen.) steady state Take-off / Maximum continuous	100% r.p.m. (100% = 28500 r.p.m)	100% r.p.m. (100% = 28500 r.p.m)
Engine Limits Maximum permissible interturbine gas temperatures	Takeoff (5 minutes max)	920°C (1688°F)	920°C (1688°F)
	Max. continuous	920°C (1688°F)	920°C (1688°F)
	Starting	950°C (1742°F)	950°C (1742°F)
	Transient (20 sec.) and starting	930°C (1706°F)	945°C (1733°F)

Note : Engine is approved for operation with thrust reverser p/n F7XC782140020

6. Auxiliary Power Unit (APU):

APU model 36-150 [FN], from Honeywell (Allied Signal), APU is non essential.

APU limitations: according to applicable EASA approved Aircraft Flight Manuals (AFM); AFMs are referenced in Chapter IV.1.

Maximum operating altitude

usable for ground operation only

Maximum Starting Altitude

usable for ground operation only

7. Reserved

8. Fluids (Fuel, Oil, Additives, Hydraulics):

The fluids are defined in the applicable EASA approved Aircraft Flight Manuals (AFM); AFMs are referenced in Chapter IV.1.

9. Fluid capacities

9.1 Fuel Capacity (Density: 0.803 kg/dm³ (6.7 lbs/US gallon))

Usable Fuel	Without modification M1000 included (S/N 0001 to 0400)		With modification M1000 included (S/N 0401 and ongoing) ^{NOTE 3}	
	Volume [dm ³ (gals (US))]	Mass [kg (lbs)]	Volume [dm ³ (gals (US))]	Mass [kg (lbs)]
LH circuit	5944 (1570)	4773 (10522)	6359 (1680)	5106 (11257)
RH circuit	5944 (1570)	4773 (10522)	6383 (1686)	5126 (11301)
Center circuit	6154 (1626)	4942 (10896)	7108 (1878)	5708 (12583)
Total (all tanks)	18042 (4766)	14488 (31940)	19850 (5244)	15940 (35141)
Unusable Fuel				
Drainable	65 (17)	52 (115)	68 (18)	54 (120)
Undrainable	41 (11)	33 (72)	48 (13)	39 (85)
Total unusable (all tanks)	106 (28)	85 (187)	116 (31)	93 (205)

See NOTE 1

9.2 Oil (Density: 0.95 kg/dm³ (7.94 lbs/gal) or (1.99 lbs/qt))

	Volume per engine [dm ³ (gals (US))]	Mass [kg (lbs)]
Max Oil Level (Total)	23.61 (6.24)	23.01 (50.7)
Min Oil Level (Total)	18.69 (4.92)	18.21 (40.14)

See NOTE 1

9.3 Hydraulics (Density: 0.84 kg/dm³ (7.0 lbs/US gallon))

	Volume [dm ³ (gals (US))]	Mass [kg (lbs)]
Hydraulic Fluid - System (Total)	65,19 (17,22)	54,95 (121,15)

See NOTE 1

10. Airspeed Limits:

The airspeed limits are defined in the applicable EASA approved Aircraft Flight Manuals (AFM); the AFMs are referenced in Chapter IV.1.

11. Flight Envelope:

The flight envelope is defined in the applicable EASA approved Aircraft Flight Manuals (AFM); the AFMs are referenced in Chapter IV.1.

Maximum Operating Altitude 15544 m (51000 ft)

12. Operating Limitations

12.1 Approved Operations

The Falcon 7X is eligible for the following kinds of operation when the appropriate equipment and instruments required by the operating requirements are installed, approved, and operating as defined by the MMEL or MEL:

- VFR (Visual)
- IFR (Instrument)

- Day
- Night
- Icing
- Manual or Automatic Category I approaches and non precision approaches
- Manual or Automatic Category I approaches and non precision approaches with EVS operational credit (for S/N 0001 to 0400 only)
- Automatic Cat. II approaches with or without HUD monitoring / EVS M-OPT-0002, Category II requirements provided the airplane is operated in accordance with Airplane Flight Manual Annex 1 and with Supplement 1 revision 1 (or later approved revision) when monitored with HUD (for S/N 0001 to 0400 only)
- LPV approaches
- LPV approaches with EVS Operational Credit (for S/N 0001 to 0400 only)
- Enhanced Surveillance
- RVSM
- RNP RNAV operations, down to RNP 0.3 RNAV (RTCA/DO-236B and DO-283)
- Extended flight over water and uninhabited terrain
- Polar operations (limited 85° North / 85° South)
- Contaminated runways operation (for S/N 0001 to 0400 only)
- Steep approach landing from 4.5 to 6.0 degrees (for S/N 0001 to 0400 only)
- Landing and take off between 8000 ft and 15000 ft.
- Operations with landing gear down (for S/N 0001 to 0400 only)
- ADS-B Out function certified in the frame of EASY II M1122 and subsequent EASy versions compliant with EU 1028-2014 and CS-ACNS

12.2 Other Limitations

Other limitations as defined in the applicable EASA approved Aircraft Flight Manuals (AFM); the AFMs are referenced in Chapter IV.1.

13. Maximum Certified Masses

	Without modification M1000 included (S/N 0001 to 0400)			With modification M1000 included (S/N 0401 and ongoing) ^{NOTE 3}		
	Mass	Fwd limit % MAC	Aft limit % MAC	Mass	Fwd limit % MAC	Aft limit % MAC
Ramp	31842 kg (70200 lbs)	19.5	31.5	33203 kg (73200 lbs)	19.62	27.00
Takeoff	31751 kg (70000 lbs)	19.05	33.65	33112 kg (73000 lbs)	19.54	27.73
Aft CG at 38.5%	25890 kg (57076 lbs)	19.5	38.5	25890 kg (57076 lbs)	19.5	38.5
Landing	28304 kg (62400 lbs)	19.5	37.35	28304 kg (62400 lbs)	19.5	37.35
Zero fuel	18597 kg (41000 lbs)	19.5	38.5	18597 kg (41000 lbs)	19.5	38.5
Minimum flight - FWD	15694 kg (34600 lbs)	26.0	N/A	15694 kg (34600 lbs)	26.6	N/A
Minimum flight - AFT	14696 kg (32400 lbs)	N/A	38.5	14696 kg (32400 lbs)	N/A	38.5

See Note 1 for weight and balance calculation, refer to the Loading Manual in Chapter IV.3.

14. Centre of Gravity Range: The Centre of Gravity Ranges are defined in the applicable EASA approved Aircraft Flight Manuals (AFM); the AFMs are referenced in Chapter IV.1.
15. Datum: 25 % of mean aerodynamic chord (MAC): 12 183 mm from the forward end of the aircraft nose cone (12741 mm in case of modification M1000 installed)

16. Mean Aerodynamic Chord (MAC):

3.34754 m (131.79 in.)

Without modification M1000 included (S/N 0001 to 0400)	With modification M1000 included (S/N 0401 and ongoing) ^{NOTE 3}
0 % MAC is at 11.3461m (446.7 in.) aft of datum. 25 % MAC is at 12.183m (479.65in) aft of datum	0 % MAC is at 11.9049m (468.69 in.) aft of datum 25 % MAC is at 12.7418m (501.646in) aft of datum

17. Levelling Means: Aircraft is levelled in the longitudinal and lateral axis by means of a plumb bob and target in the left main landing gear bay
18. Minimum Flight Crew: For all flights: 2 (pilot and co-pilot)
19. Minimum Cabin Crew: None
- 20 Maximum Seating Capacity: Up to 22: 2 pilots +1 crew (third crew member seat authorized for take-off and landing in the cockpit or crew rest area) + up to 19 passenger seats
see Note 2
21. Baggage / Cargo Compartment: Maximum allowable loads Baggage compartment: not to exceed 300 kg per square meter.
See note 1.
22. Wheels and Tyres: This aircraft is equipped with wheels, brakes, nose wheel single chine radial tubeless tires and main wheels H type radial tubeless tires.
Main wheel tyres are H32×10.5R16.5
Nose wheel tyres are 16×6.0R6
Mixability is not approved.
23. Reserved

IV Operating and Service Instructions

1. Airplane Flight Manual (AFM)

DGT105608, Airplane Flight Manual (AFM) Model Falcon 7X
applicable to S/N 0001 to S/N 0400 (modification M1000 not included)

DGT147681, Airplane Flight Manual (AFM) Model Falcon 7X
applicable to S/N 0401 and ongoing (modification M1000 included) ^{NOTE 3}

2. Instructions for Continued Airworthiness and Airworthiness Limitations that consist of:

- Maintenance Review Board Report DGT 102566
- Chapter 5-40 DGT 107838
- Airplane Maintenance Manual included in FIELD publication in CD format
- Structural Repair Manual included in FIELD publication in CD format

3. Loading Manual (for Weight and Balance calculation)

DGT108840, Loading Manual for Model Falcon 7X
applicable to S/N 0001 to S/N 0400 (modification M1000 not included)

DGT147688, Loading Manual for Model Falcon 7X
applicable to S/N 0401 and ongoing (modification M1000 included) ^{NOTE 3}

V Operational Suitability Data (OSD)

The Operational Suitability Data elements listed below are approved by the European Aviation Safety Agency under the EASA Type Certificate EASA.A.155 as per Commission Regulation (EU) 748/2012 as amended by Commission Regulation (EU) No 69/2014.

1. Master Minimum Equipment List (MMEL)

The MMEL grandfathered as per the defined OSD certification basis in chapter II.4.1, is the Falcon 7X DGT106042 at revision 9 dated 8 October 2013 or later approved revisions.

2. Flight Crew Data

The Flight Crew Data approved as per the defined OSD certification basis in chapter II.4.2, is the Falcon 7X Operational Suitability Manual – Flight Crew (OSM-FC) DGT148654 dated 4 Sept. 2015 or later approved revisions.

Pilot Type Rating: Falcon 7X (with and without M1000 modification included) ^{NOTE 3}

3. Cabin Crew Data

Not applicable

4. Simulator Data

The Simulator Data approved as per the defined OSD certification basis in chapter II.4.4, is the Operational Suitability Manual Simulator (OSM-SIM) Validation Road Map (VDR) DGT152476 original issue or later approved revisions (applicable only to S/N 0401 and ongoing with modification M1000 and M1254 included).

^{NOTE 3}

5. Maintenance Certifying Staff Data

The Maintenance Certifying Staff Data approved as per the defined OSD certification basis in chapter II.4.5, is the Operational Suitability Manual Maintenance Certifying Staff (OSM-MCS) DGT153370 original issue or later approved revisions.

VI. Notes

- NOTE 1: a) The airplane must be loaded according to the appropriate approved Loading Manual (for Weight and Balance calculation). The list of equipment included in certificated empty mass must be provided for each airplane at the time of original certification. A current weight and balance report must be carried in the aircraft at all times from the moment the aircraft is originally certified.
The certified empty mass and corresponding center of gravity location must include the fluids of chapter III.9
- b) Loading of the aircraft must be accomplished in a manner that always maintains the center of gravity within the specified limits considering crew and passenger movements as well as fuel consumption and transfer.
- NOTE 2: Cabin interior and seating configuration must be approved.
- NOTE 3: The introduction of the fuselage stretch modification M1000, for which the avionics EASyIII modification M1254 is a precondition, has the commercial designation "F8X". This modification is applicable for all F7X models from S/N 0401 and ongoing.

VII Abbreviations

APU:	Auxiliary Power Unit
AWO:	All Weather Operation
CCD:	Cabin Crew Data
CRI:	Certificaton Review Item
CS:	Certification Specification
EASA:	European Aviation Safety Agency
ESF:	Equivalent Safety Finding
FCD:	Flight Crew Data
ICAO:	International Civil Aviation Organization
INT/POL:	JAA Interim Policy
JAR:	Joint Aviation Requirement
MCSD:	Maintenance Certifying Staff Data
MMEL:	Master Minimum Equipment List
MEL:	Minimum Equipment List
NPA:	Notice of Proposed Amendment
OSD:	Operational Suitability Data
RVSM:	Reduced Vertical Separation Minima
SB:	Service Bulletin
SC:	Special Condition
SIMD:	Simulator Data
S/N:	Serial Number
TCDS:	Type Certificate Data Sheet
TCDSN:	Type Certificate Data Sheet for Noise

VIII Change Record

Issue	Date	Changes
Issue 1 to 3		No tracking of detailed changes. Changes log implemented only at issue 4.0 according to new EASA procedure.
Issue 4	20 Jan. 2010	<p>Page 5 Section 2.II.6</p> <ul style="list-style-type: none"> - Elect to comply, removal of JAR 25.907 amt16 propeller vibration as not applicable to the Falcon 7X <p>Page 7 Section 2.III.4 and 3.III.5</p> <ul style="list-style-type: none"> - Addition of EASA approved AFM reference DGT 105608 <p>Page 8 Section 2.III.10</p> <ul style="list-style-type: none"> - Maximum Weights – removal of weight table for A/C without M0478 and M0826 as all Falcon 7X fleet has been retrofitted with these modifications and new deliveries are automatically fitted with these modifications. Note that M0826 is only a justification of the maximum ramp and take off weights without design change, therefore M0826 applies to all F7X A/C even if not in the A/C initial RIC. This removal is to ease operator understanding of the F7X TCDS. <p>Page 10 Section 2.IV</p> <ul style="list-style-type: none"> - Correction of Chapter 5-40 reference and removal of operating and service instructions publication format.
Issue 5	15 April 2015	<ul style="list-style-type: none"> - Editorial revision to reflect latest EASA TCDS format - II.3 Addition of paragraphs for modification M-OPT0760 - II.3.1 Addition of SC for modifications M0194, M-OPT0459, M-OPT0686, M-OPT0002, M-OPT0359, M-opt0017 and M1122 - II.3.4 Addition of ESF for modifications M-OPT0649 - II.4 OSD MMEL requirement - III.12.1 Update of approved operations - IV.3 Addition of Loading Manual - V Addition of OSD chapter
Issue 6	15 Dec. 2015	<ul style="list-style-type: none"> - II.2 reference date for OSD added - II.4.3 to II.4.5 OSD added - III.15 Datum corrected - V.2 to V.5 added - VII abbreviations regarding OSD added
Issue 7	24 June 2016	<ul style="list-style-type: none"> - II.3, II.3.1, II.3.3, II.3.4, II.3.5, II.4, III.2, III.4, III.5, III.9.1, III.12, III.13, III.15, III.16, IV.1, IV.3, V, VI NOTE 3 updated to include data for the modification M1000 that is applicable to all F7X models from S/N 401 and ongoing