















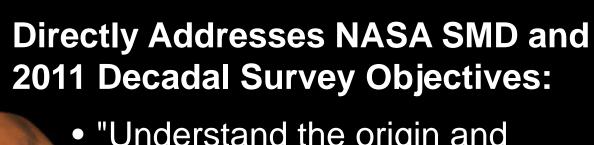








Understand the formation and evolution of terrestrial planets through investigation of the interior structure and processes of Mars.



 "Understand the origin and diversity of terrestrial planets."

 "Understand how the evolution of terrestrial planets enables and limits the origin and evolution of life."



#### InSight Will Measure Interior Structure with Unprecedented Precision

Measurement	Current Uncertainty	InSight Capability	Improvement
Crustal thickness	65±35 km (inferred)	±5 km	7X
Crustal layering	no information	resolve 5-km layers	New
Mantle velocity	8±1 km/s (inferred)	±0.13 km/s	7.5X
Core liquid or solid	"likely" liquid	positive determination	New
Core radius	1700±300 km	±75 km	4X
Core density	6.4±1.0 gm/cc	±0.3 gm/cc	3X
Heat flow	30±25 mW/m <sup>2</sup> (inferred)	±3 mW/m <sup>2</sup>	8X
Seismic activity	factor of 100 (inferred)	factor of 10	10X
Seismic distribution	no information	locations ≤10 deg.	New
Meteorite impact rate	factor of 6	factor of 2	3X

## InSight

#### **InSight Mission Summary**

 InSight will fly a near-copy of the successful Phoenix lander

• Launch: March 4-24, 2016 from

Vandembergg AATEB

Fast, type-1 trajectory, 6.5-mo.
 cruise to Mars

Landing: September 28, 2016

67-sol deployment phase

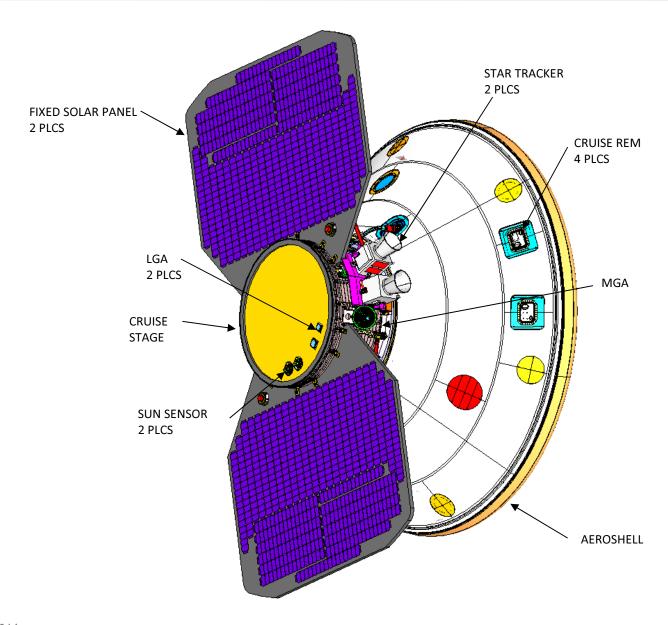
 Two years (one Mars year) science operations on the surface; repetitive operations

Nominal end-of-mission:
 October 6, 2018



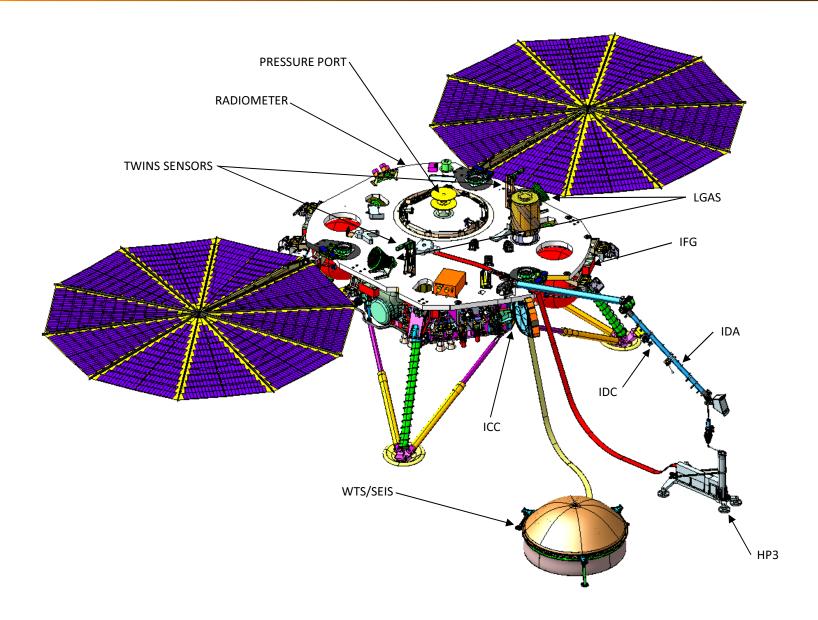


# InSight Spacecraft: Cruise Configuration





## Instruments Deployed





#### InSight Payload



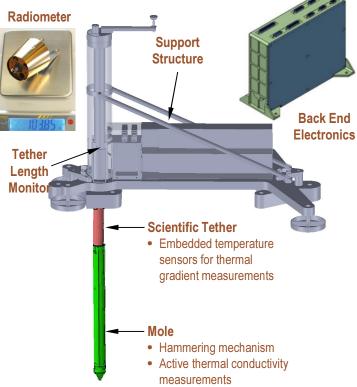
Small Deep Space Transponder

RISE (S/C Telecom)

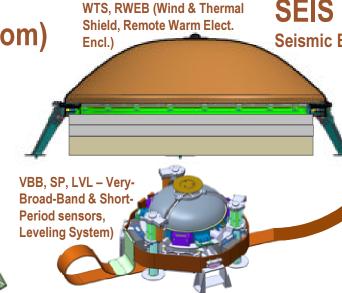
Rotation and Interior Structure Experiment

HP<sup>3</sup> (DLR)

**Heat Flow and Physical Properties Package** 

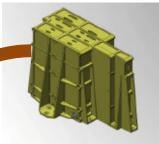


Static tilt sensors



SEIS (CNES) (also IPGP, ETH/SSA, MPS/DLR, IC/Oxford/UKSA, JPL/NASA)
Seismic Experiment for Interior Structure

THR - Tether



**Ebox – Electronics Box** 

Pressure Sensor (Tavis)



TWINS (CAB) – Temp. and Wind for INSight





APSS (JPL)
Auxiliary Payload

Sensor Suite

IDS (JPL)

**Instrument Deployment System** 



**IDA – Instrument Deployment Arm** 



**IDC – Instrument Deployment Camera** 

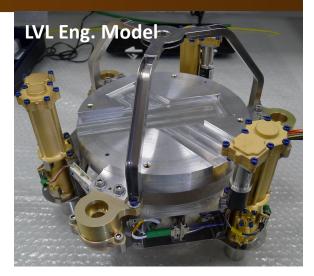
**ICC – Instrument Context Camera** 



#### **SEIS Development Proceeding on Tight Schedule**

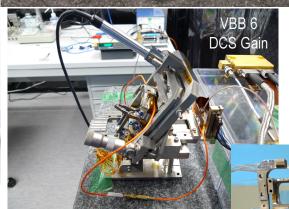








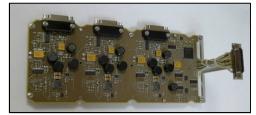




**VBB Qual. Model** 







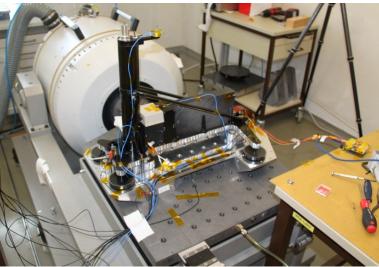
SP EM Sensor and Electronics

**VBB 6 Pivot cabling** 



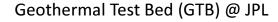
#### HP<sup>3</sup> Back on Track After Mole Redesign





Support Structure EM in Vibration Testing

Mole Pre-Protoflight Model

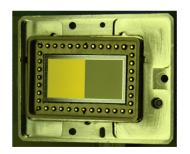


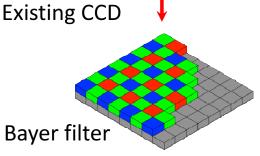
Pre-PFM assembled

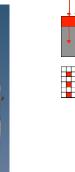


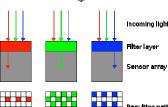
#### **Adding Color to Instrument Deployment Camera**

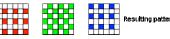
- Construct a color IDC camera by upgrading the CCDs in existing InSight spares.
- Continue with the existing BW cameras in the test bed and integration flow, and replace with color unit if this program is successful and on time.

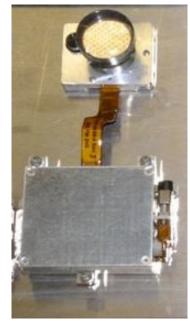


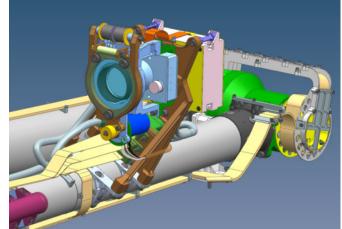












## InSight

#### Status and Schedule – Key Milestones

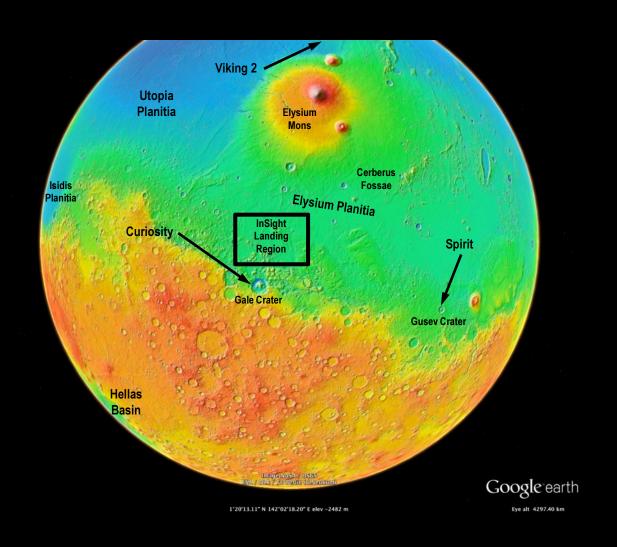
- Passed PDR and Confirmation Review
  - In Development and Fabrication
- On Budget Reserves Exceed NASA Guidelines
- On Schedule Margins Exceed what Proposed
- Instrument & System Capabilities Exceed All Science Reqs
- May 2014: Critical Design Review
- October 2014: System Integration Review
- January 2015: Deliver instruments to ATLO
- Participating Scientist Program
  - ~dozen new scientists before launch
- November 2015: Confirm landing site
- December 2015: Ship to Vandenberg
- March 2016: Launch
- September 2016: Landing

The InSight Mission to Mars

· Ostabar 2010. End of primary mission



## InSight Landing Region: Elysium Planitia





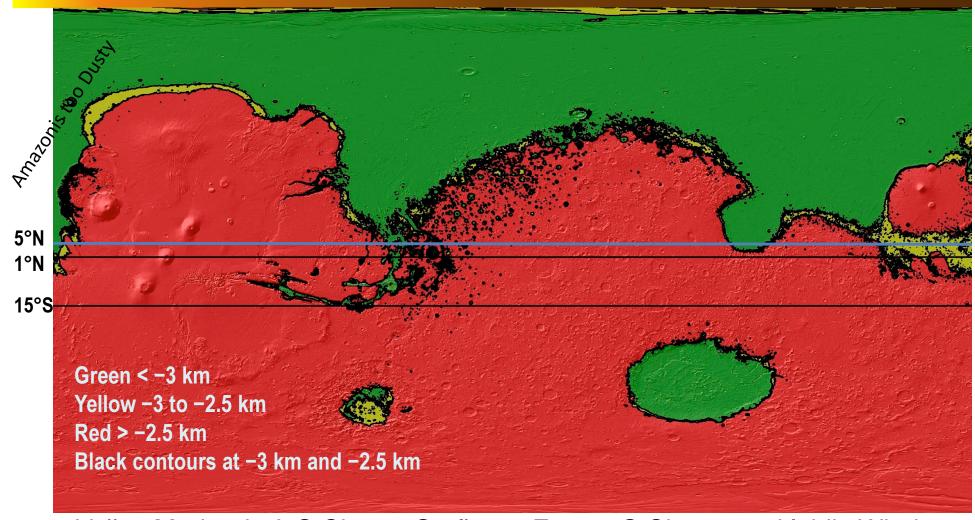
### **Landing Site Constraints**

- Latitude: 15°S to 5°N: Sufficient Solar Power Margins
  - -5% to 3%, 2% Elysium Planitia; Thermal
- Elevation: <-2.5 km MOLA: Sufficient Atmosphere for EDL
- **Ellipse Size:** 139 km × 27 km [99.5% ellipse];
  - $-130 \times 27 \text{ km} \rightarrow 110 \times 25 \text{ km}$
- Thermal Inertia:  $>100-140 \text{ J m}^{-2} \text{ K}^{-1} \text{ s}^{-1/2}$ 
  - Avoid surfaces with thick dust that is not loading bearing
  - Prefer ~200 J m<sup>-2</sup> K<sup>-1</sup> s<sup>-1/2</sup> for uncemented or poorly cemented soil
  - Radar reflective surface
- Rock Abundance: <10%
  - 99% Safe Landing and Opening Solar Panels
- Smooth Flat Surface: No large relief features
  - Slopes <15° for Safe Touchdown and Radar Tracking (1-5 m & 84 m)</li>
- Deploy Instruments: [<10% Rock Abundance, <15°Slope]</li>
- Broken up regolith >5 m thick: Hesperian Cratered Surface
  - Penetration of the Mole

No Other Science Requirements: Just Land Safely



#### **Global Latitude and Elevation**



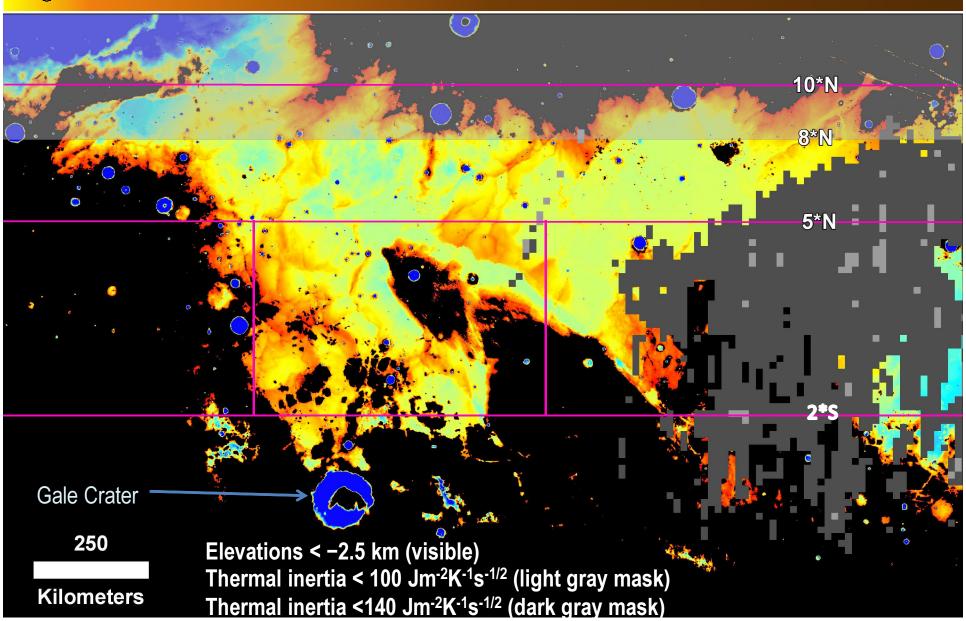
Valles Marineris & S Chryse Outflow S Isidis Planitia Both Rocky

<sub>7/29/13</sub>Elysium

Expect S Chryse and Isidis Windy GCMs Storm Tracks High N Lat. Valles Marineris Canyons Windy S Elysium Low Winds

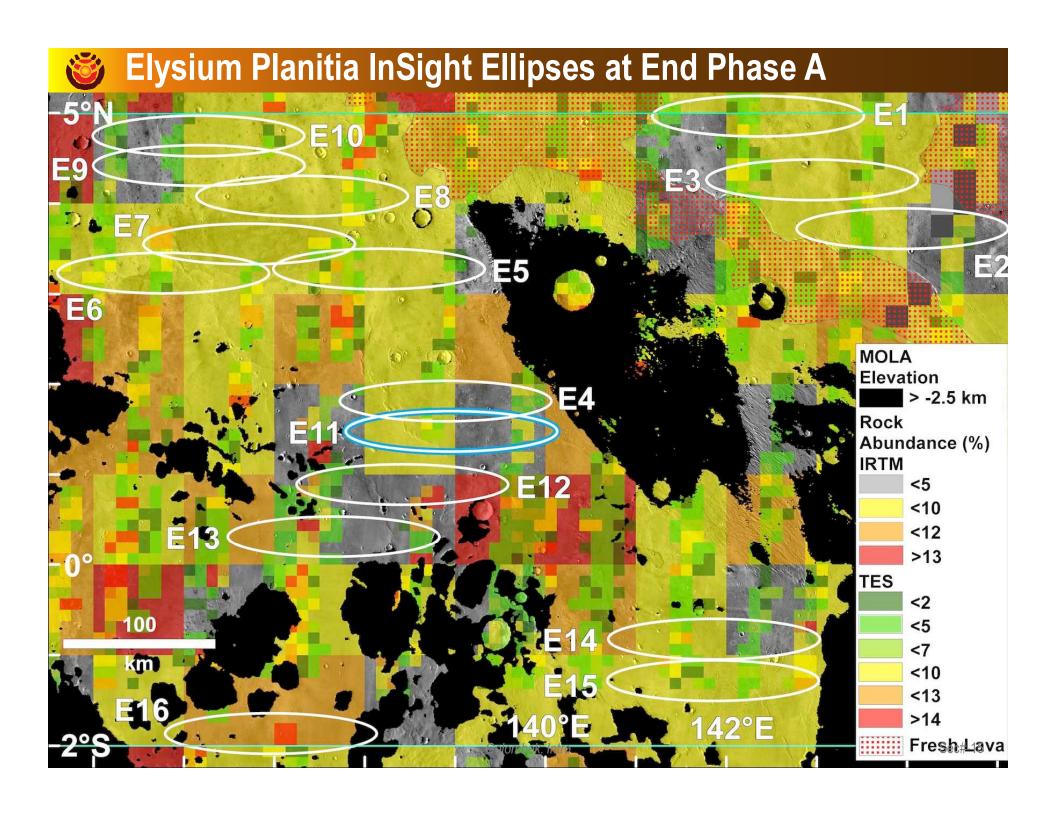


#### Elysium Planitia Elevation, Latitude & Thermal Inertia



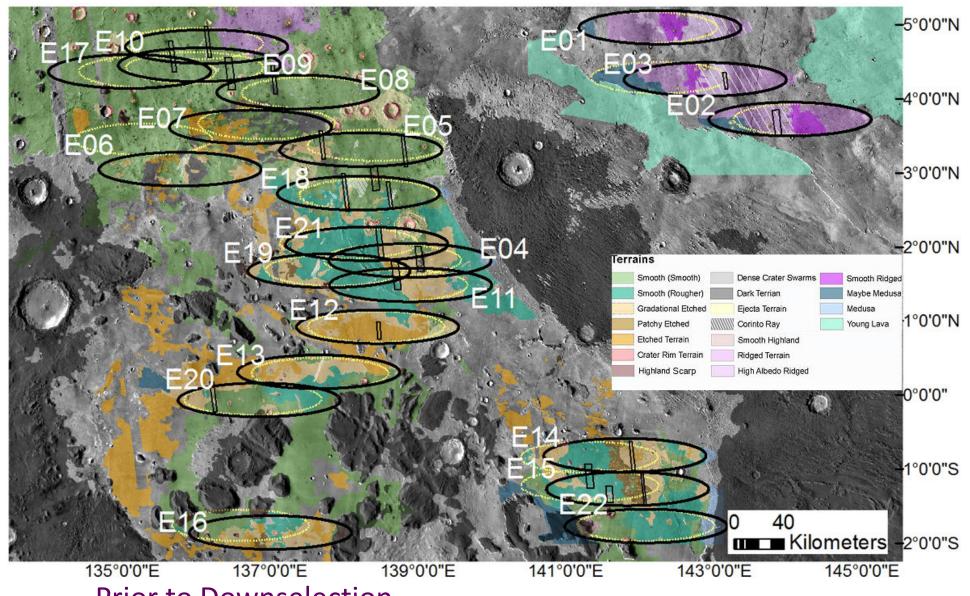
7/29/13

Golombek: Intro





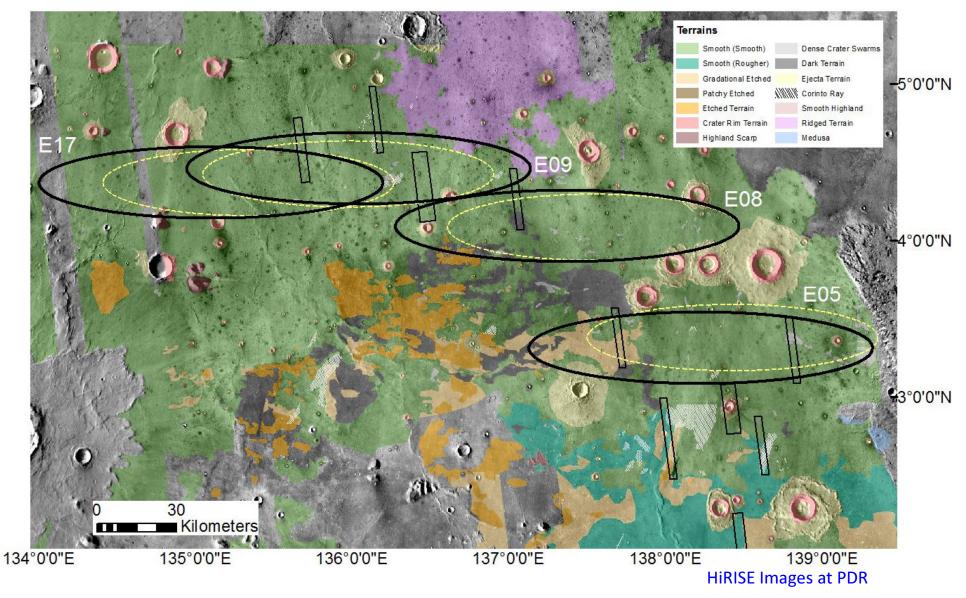
## **Terrain Map & Ellipses**



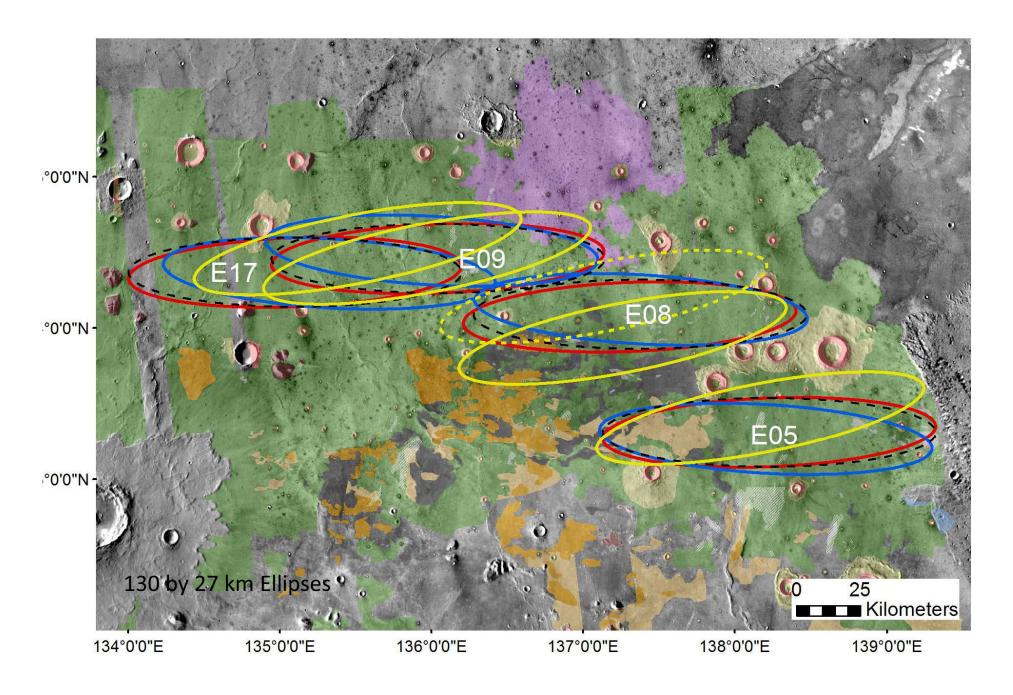
**Prior to Downselection** 

20 HiRISE Footprints at PDR

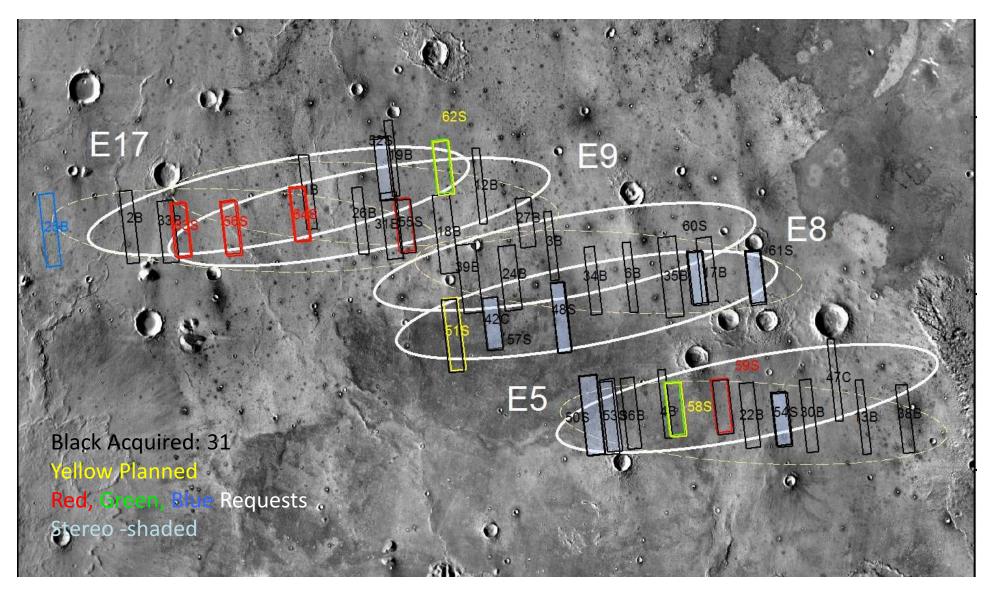
## Final Four Ellipses



#### Azimuthal Change at Open, Middle & Close

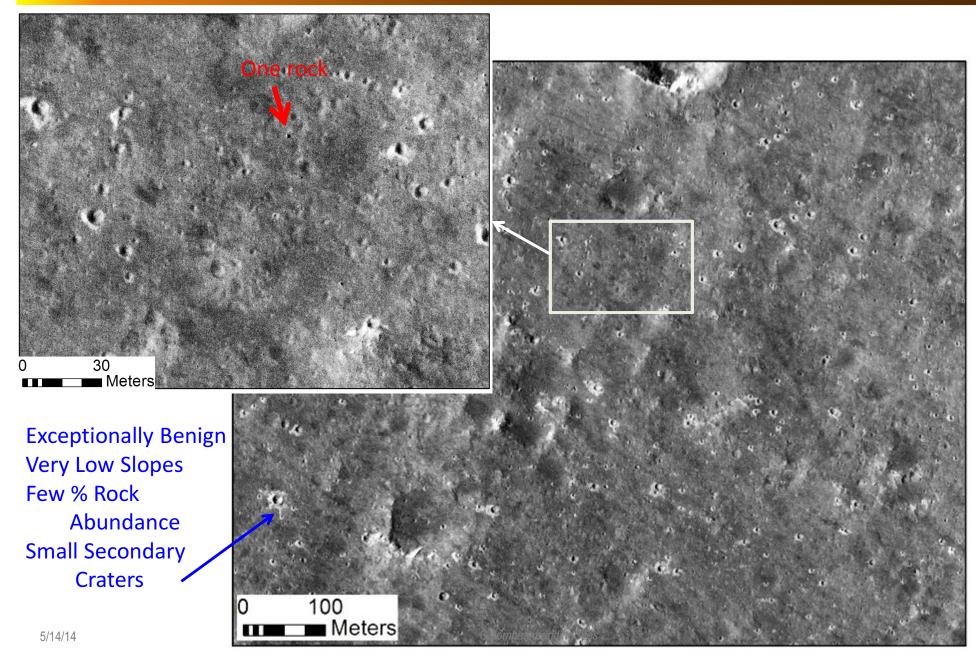


### HiRISE Images as of 5/6/14

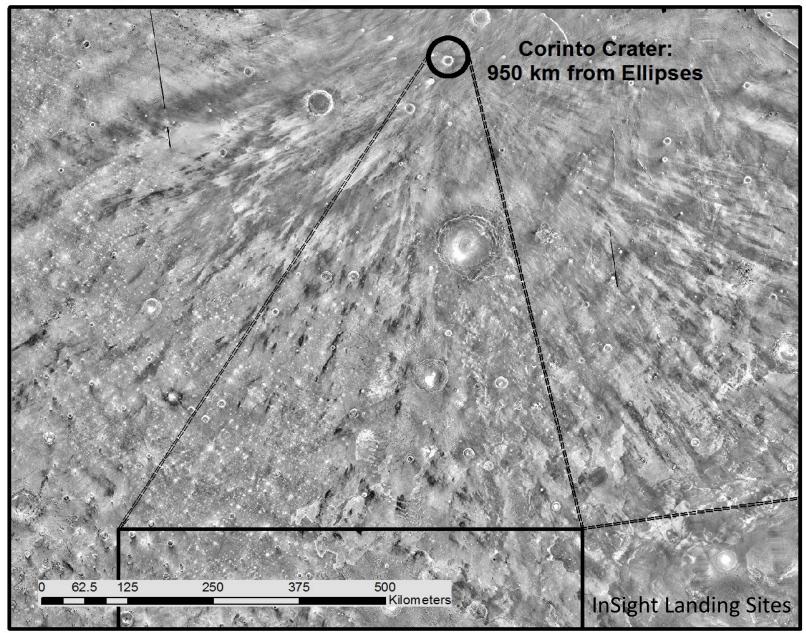




#### **Smooth Terrain: HiRISE**



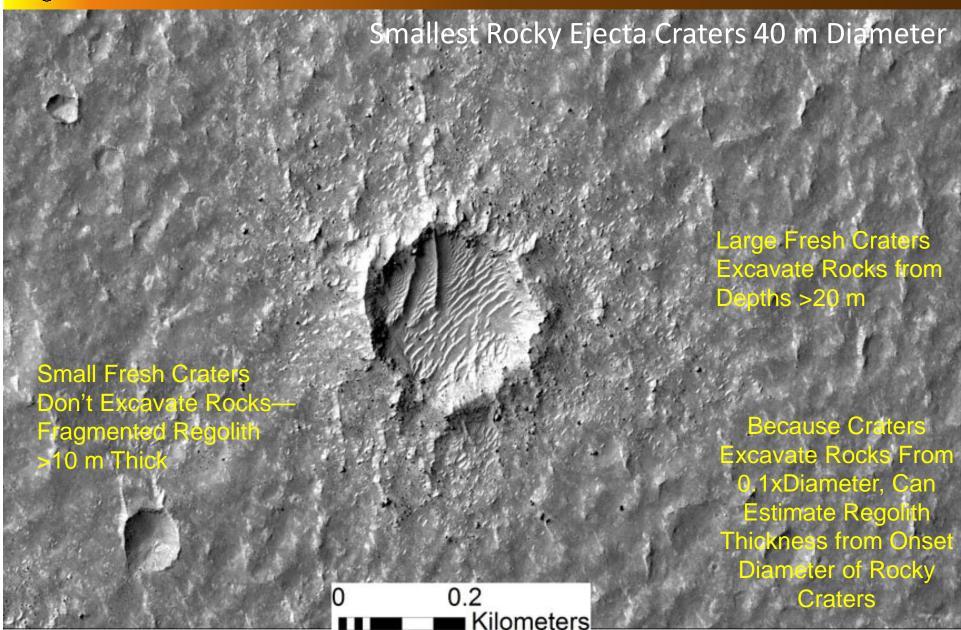
#### Secondary Crater Rays from Corinto: THEMIS IR Night



5/4/14 Golombek: Landing Sites 21



#### Fresh Craters = Boulders and Bedforms



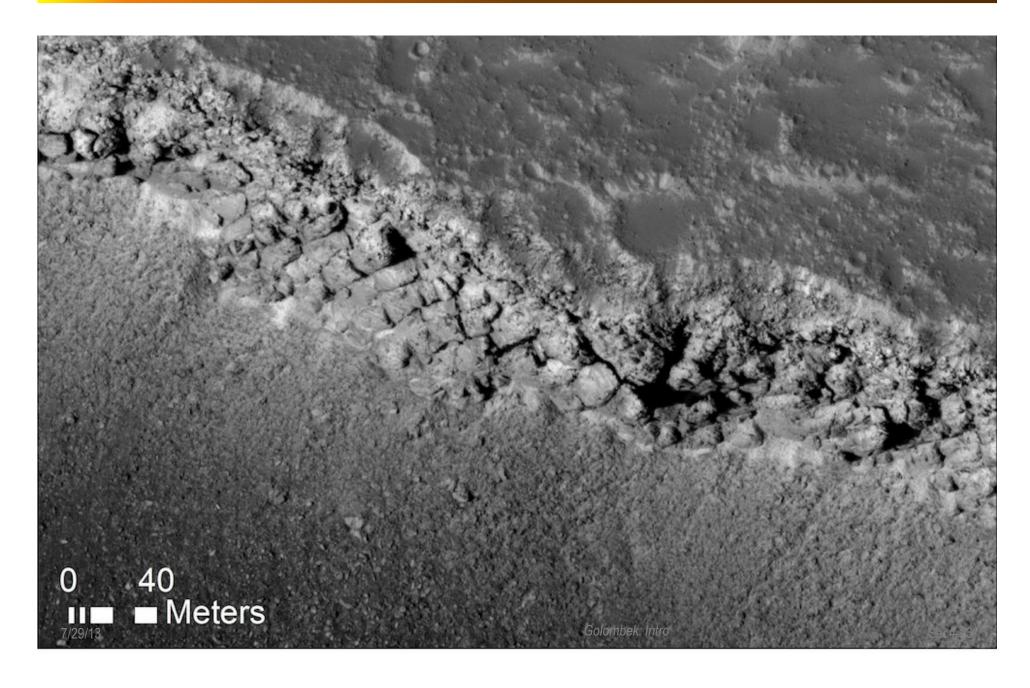
7/29/13

Golombek: Intro

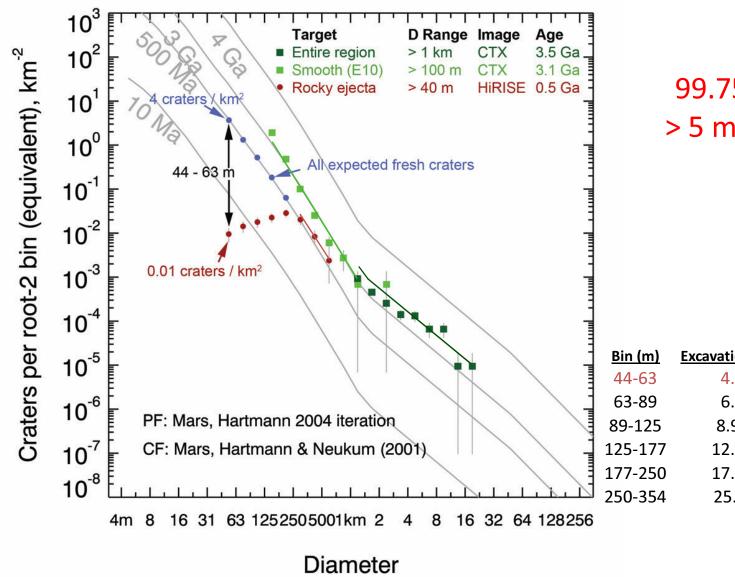
22



## **Cross Section of Regolith**



## Regolith Thickness

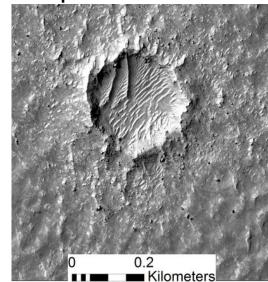


99.75% of area > 5 m of regolith

<u>Bin (m)</u>	Excavation Depth (m)	% of Expected
44-63	4.4 - 6.3	0.25
63-89	6.3 - 8.9	1.3
89-125	8.9 - 12.5	3.9
L25-177	12.5 - 17.7	10
177-250	17.7 - 25.0	48
250-354	25.0 -35.4	100

## Landing Site Hazards

- Some Non-Smooth Terrain in Some of Ellipses
  - Higher Slopes and more Rocks
  - Estimated Increased Risk due to More Rocks and Higher Slopes
- Secondary Craters Have Steep Slopes (>15)
  - Area covered is Small and Non Uniform
  - Risk is Estimated at ~0.4%-1.2% for Different Ellipses
- Rocky Ejecta Craters
  - Sprinkled Uniformly Across Region
  - <1% of Area has Higher Rock Abundance</p>
  - ~0.1-0.2% Added Risk
- Sum All Risks
  - -~1.4-2.2% Risk



Will Find Acceptable Ellipse(s) ~97-99% Safe



#### **Schedule**

- CSR End Phase A, 16 Ellipses in Elysium Planitia, May 24, 2012
- First Landing Site Workshop, 6/27/13
- First Landing Site Downselection, 7/29/13
  - Downselected from 22 to 4 Ellipses
- Presently Half Way through Process
- Second Landing Site Workshop, Summer/Fall 2014
  - Full discussion of everything known about sites
- Second Landing Site Downselection, Fall 2014
  - to 2-3 ellipses
- Complete HiRISE Coverage, Stereo Samples, Certification
- Landing Site Selection
  - Project Recommendation/Selection, 8/15
  - Peer Review, PPO Review, NASA Hq Concurrence, ~10/15

5/14/14 Golombek: Landing Sites 26