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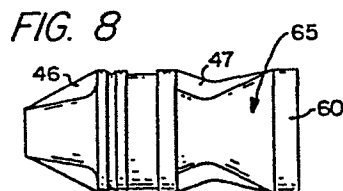
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54 **Method of manufacture of a metallic sabot.**

57 The method of manufacture of a sabot component which comprises the steps of severing an aluminum workpiece (50) from bar stock, upsetting the workpiece in a first die, by axial force, to constitute a first preform (51) having a desired uneven mass distribution along the axis thereof, hot forging the first preform in a second die, by transverse force, to constitute a second preform (52) having an unsymmetrical mass distribution transverse to the axis thereof, hot forging the second preform in a third die, by transverse force, to constitute a stamping (60) having opposed unsymmetrical formed surfaces, solution heat treating the stamping to a "W" condition, freezing the stamping to render its metallography dormant, cold forging the stamping in the third die, artificially aging the stamping to a "T8" condition, trimming the stamping in a fourth die, and anodizing the trimmed stamping.



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Method of Manufacture of a Metallic Sabot

The present invention relates to a method of manufacture of a metallic sabot according to the preamble of claim 1.

In the field of munitions it is often necessary  
5 to attack targets which are armored so that simple  
explosion of a munition against the armor is not  
sufficient to disable the target. For such use there  
have been developed projectiles known as "penetrators"  
which are not necessarily explosive on impact, but are  
10 of density, configuration, and propulsive force  
sufficient for maximum penetrating effects. One such  
device is known as a "long rod" penetrator: it is of  
dense metal such as a tungsten alloy, and is of  
relatively small diameter compared to the bore of the  
15 cannon generally available. It extends along a

longitudinal axis from a forward, aerodynamic point to a rearward stabilizing fin structure. To obtain sufficient propulsion force, it is necessary to make these projectiles parts of munitions having cartridge cases several times the diameter of the body of the projectile, although of course the mouth of the cartridge case must be large enough to pass the rearward fin structure of the projectile. Means must also be provided for guiding the relatively small penetrator down the larger barrel of the firing piece and preventing the escape of propulsive gases down the barrel past the projectile.

These functions have been accomplished in the past by providing the penetrator with circumferential external buttress grooves along its midportion and engaging similar internal grooves in a bi-partite "sabot" secured in place by a loosely fitting "obturator ring" of nylon which rides in the rifling of the discharging barrel, thus preventing escape of propulsion gases. Rearwardly the sabot is grooved for crimped connection with the mouth of the cartridge case, within which the fin structure of the penetrator is received. Forwardly the sabot is configured to cooperate with a protective cap during travel through the barrel: cap, ring, and sabot separate from the penetrator upon leaving the barrel, to avoid increasing the drag of the penetrator during flight.

It is the object of the present invention to devise a method of manufacture of a metallic sabot which results in a cheap sabot.

5 This object is achieved according to the characterizing features of claim 1. Further advantageous steps of the inventive method and of a sabot produced by said method may be taken from the subclaims.

0 The present invention contemplates redesigning the sabot so that its parts can be made by metal forming, using a preformed rod and a compound die. In order to accomplish this, the amount of metal flow required in going from bar  
5 procedures must be followed if a precise product of stable dimensions is to result.

In the drawing, in which like reference numerals identify corresponding parts throughout the several views:

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FIGURE 1 is a general view of a munition of the type in question,

FIGURE 2 shows a sabot according to the invention, and

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FIGURES 3-8 show steps in the manufacture of a sabot half according to the invention.

As shown in FIGURE 1, a munition 10 according to the invention comprises a penetrator 11 arranged for firing from a cartridge case 12 containing a propulsion explosive 13 which is fired when a primer or percussion cap 14 is impacted by the firing pin of the gun, not shown. The munition as shown is inserted into the chamber 15 of the firing piece, which has a barrel 16.

The penetrator is shown to extend along a longitudinal axis 18 from a forward portion 19, pointed for aerodynamic reasons, to a rearward fin section 20, for flight stabilization. Approximately at its midportion, the penetrator is provided with a set of circumferential buttress grooves 21 for engagement by matching grooves 22 in a sabot 23 to which the mouth 24 of cartridge case 12 is crimped at a groove 25. A forwardly tapering protective cap 26 has forward apertures 27 and a rearward lip 28 which engages a forward groove 30 in sabot 23. An obturator ring or a band 31 of nylon surrounds the rearward portion of sabot 23, and is received in a channel 32.

In the prior art, sabot 23 was bi-partite, consisting of a pair of machined metallic members having a plane mating face containing the axis 18 of the munition, and held together, and into cap 26, by nylon band 31 which was expanded over the sabot and allowed to contract into peripheral channel 32.

FIGURE 2 shows a sabot according to the invention to a slightly larger scale. It is made up of two identical halves 40 and 41 having a mating plane containing the common axis 42. Jointly the halves define a crimping groove 43, a channel 44 for band 31, and a groove 45 for cap 26. The halves are manufactured by a stamping procedure, which forms both the convex outer portion and the concave inner portion containing the necessary buttress grooves. By reason of the forming procedure it is also possible to produce sabot halves with webs 46 and 47 which give a longitudinal stiffening to the structure not readily obtainable in prior, all machined units.

The steps followed in manufacturing half sabots according to the present invention will now be described, referring to FIGURES 3-8. The units can be made of aluminum or magnesium, and FIGURE 3 shows a workpiece 50 severed from bar stock for further treatment. The workpiece is first upset in a first die, by axial force, to produce a first preform 51 having a

desired uneven mass distribution along its axis to facilitate further forming. Preform 51 is next hot-forged in a second die, by transverse force, to produce a second preform 52, opposite views of which are shown in FIGURES 5A and 5B. It will be aparent that preform 52 has an unsymmetrical mass distribution transverse to the axis, with opposite unsymmetrically formed faces 53 and 54, the latter being generally flat but having a longitudinal groove or cavity 55 bordered by raised ridges 56 and 57.

Preform 52 is next hot-forged in a third die, by transverse force, to constitute a first stamping 60 shown in FIGURE 6 to have opposite unsymmetrically formed surfaces 61 and 62, the latter again being basically flat but including a longitudinal cavity 63 having a buttress thread portion 64. The provision of ridges 56 and 57 in second preform 52 has been found to materially improve the sharpness of the buttress threads at the flat surface of first stamping 60.

Because of the relatively large change in dimensions and the metal flow necessary to accomplish this, the first stamping 60 is next solution heat treated to bring it to a "W" metallurgical condition, and then is frozen to render its metalography dormant. The stamping is returned to the same third die for cold-forging, to a form shown in FIGURE 7, then

artificially aged to a "T8" metallurgical condition, and trimmed to produce a second stamping 65--see FIGURE 8.

With respect to the "W" and "T8" metallurgical conditions it is pointed to "Metals Handbook", 8th Edition, Volume 1, Copyright by the American Society for Metals.

If webs 46 and 47 are not desired, these can be removed at the trimming operation.

The sabot half is now ready for anodizing or other final treatment.

It has been found that sabot halves made as described above are of precise dimensions and maintain their dimensional stability. A considerable cost reduction is also accomplished by following the stamping procedures instead of the previous machining operations.

The operation of the projectile with a stamped sabot is the same as with previous machine sabots. Upon firing, the propulsion gases from explosive 13 force the sabot out of the cartridge at the crimped joint and propel the projectile down the cannon barrel 16, obturator 13 riding in the rifling of the barrel to prevent escape of propulsion gases and decouple rifling spin action to the projectile. As the projectile moves out of the barrel, centrifugal force causes the halves



40 and 41 of the sabot to separate, releasing cap 26,  
and the resulting aerodynamic forces overcome the  
strength of ring 31 and permit the sabot to separate  
completely from buttress grooves 21 of the projectile  
5 along the mating plane, so that an aerodynamically clean  
projectile proceeds towards its target.

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Claims:

1. Method of manufacture of a metallic sabot, c h a r a c -  
t e r i z e d b y the steps of:  
upsetting a workpiece, having a longitudinal axis, by  
axial force in a first die, to constitute a first preform  
5 having a desired uneven mass distribution along said axis,  
and  
hot forging the preform, by transverse force in a second  
die, to constitute a second preform having a desired un-  
symmetrical mass distribution transverse to said axis.  
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2. Method according to claim 1, c h a r a c t e r i z e d  
b y the further step of hot forging the second preform  
in a third die, by transverse force, to constitute a  
stamping having opposite, unsymmetrical formed surfaces.  
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3. Method of manufacture of a metallic sabot, c h a r a c -  
t e r i z e d b y  
die forging a workpiece to constitute a preform having a  
desired uneven mass distribution along an axis and an  
20 unsymmetrical mass distribution transverse to the axis,  
hot forging the preform in a further die, by transverse  
force, to constitute a stamping having opposite un-  
symmetrical surfaces,  
solution heat treating the stamping to a "W" condition,  
25 freezing the stamping to render it metallographically  
dormant, and  
cold forging the stamping in said further die.
4. Method according to claim 3, c h a r a c t e r i z e d  
30 b y the further steps of artificially aging the stamping  
to a "T8" condition, and trimming the stamping in another  
die.
5. Method according to claim 4, c h a r a c t e r i z e d  
35 b y anodizing the trimmed stamping.

6. Sabot manufactured according to the method of claims 1 to 5, characterized in that it consists of aluminum alloy.

5 7. Sabot according to claim 6, characterized in that it comprises longitudinal stiffening webs (46, 47).

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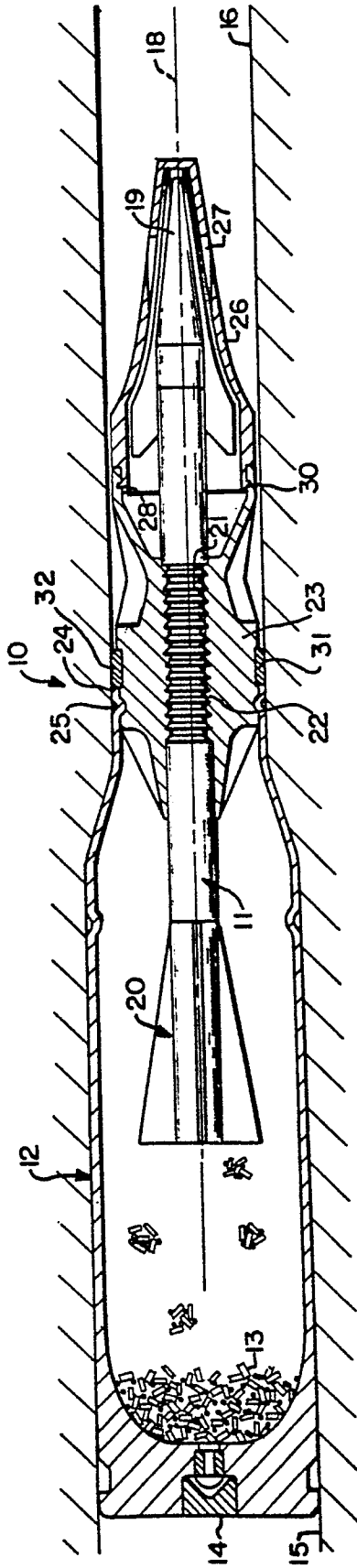


FIG. 1

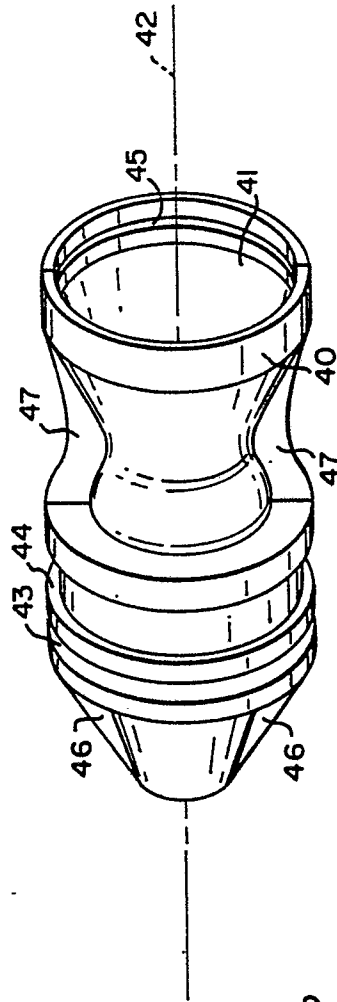


FIG. 2

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FIG. 3

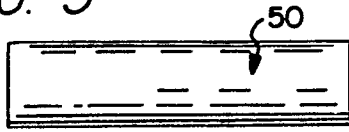


FIG. 4

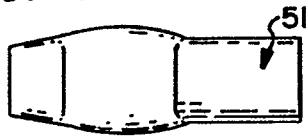


FIG. 5A

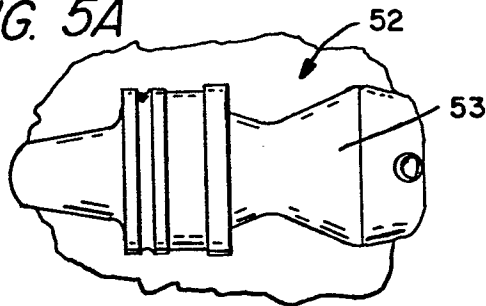


FIG. 5B

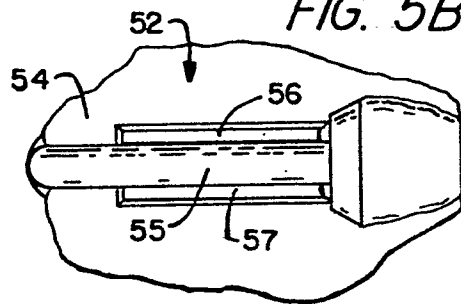


FIG. 6A

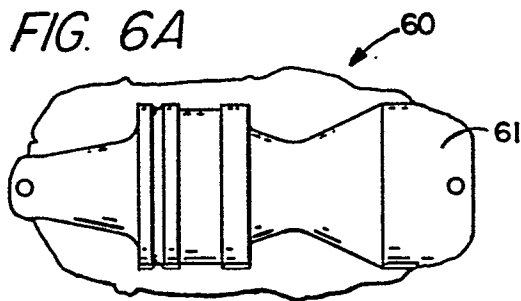


FIG. 6B

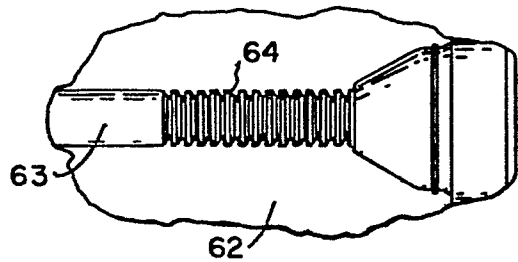


FIG. 7

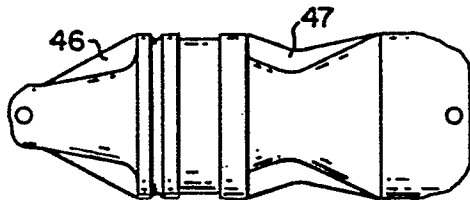
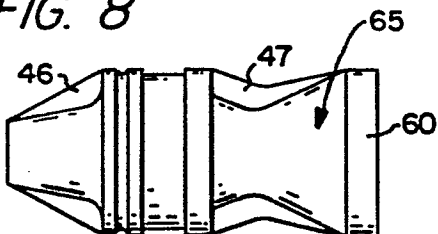


FIG. 8





European Patent  
Office

**EUROPEAN SEARCH REPORT**

**0158828**

Application number

EP 85 10 2846

DOCUMENTS CONSIDERED TO BE RELEVANT			
Category	Citation of document with indication, where appropriate, of relevant passages	Relevant to claim	CLASSIFICATION OF THE APPLICATION (Int. Cl.4)
Y	US-A-4 360 954 (BURNS et al.) * Column 1, lines 26-34; column 2, lines 34-67 *	1,6	F 42 B 13/16 B 21 K 23/00
Y	US-A-2 754 576 (FOSTER) * Figures 1-4; column 4, lines 14-75; column 5, lines 1-59; column 9, lines 45-58 *	1,6	
A	EP-A-0 068 397 (RHEINMETALL) * Whole document *	1,3	
A	US-A-1 601 016 (CARLSON) * Page 2, lines 55-129 *	1-4	
A	GB-A-1 239 473 (ROCKWELL) * Page 1, lines 46-74 *	3	TECHNICAL FIELDS SEARCHED (Int. Cl.4)
A	US-A-4 326 464 (PRICE) * Figure 4; column 1, lines 59-68; column 3, lines 12-30 *	7	F 42 B B 21 K B 21 J
The present search report has been drawn up for all claims			
Place of search THE HAGUE	Date of completion of the search 26-06-1985	Examiner HAMMOND A.D.	

**CATEGORY OF CITED DOCUMENTS**

X : particularly relevant if taken alone  
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T : theory or principle underlying the invention  
E : earlier patent document, but published on, or after the filing date  
D : document cited in the application  
L : document cited for other reasons  
& : member of the same patent family, corresponding document