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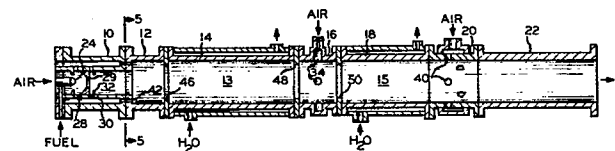
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⑳ **Method and apparatus for burning a fuel.**

⑴ A method for burning a fuel, containing chemically bound nitrogen, in a two-stage, rich-lean combustion process, including; introducing the fuel and at least one stream of primary air into a primary combustion region at a fuel-air ratio above the stoichiometric ratio and in a manner to intimately mix the fuel and air and establish a stabilized flame adjacent the upstream end of the primary combustion region, maintaining the flame in the primary combustion region for a period of time sufficient to produce a combustion product mixture containing less than a predetermined amount of NO<sub>x</sub> pollutants and abruptly terminating the primary combustion region while introducing at least one stream of secondary air into the secondary region in an amount sufficient to reduce the overall fuel-air ratio below the stoichiometric ratio and in a manner to prevent backflow of the secondary air into the primary combustion region. A method for initiating and maintaining the combustion process is described as well as apparatus for carrying out the combustion process.



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D e s c r i p t i o n

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Method and Apparatus for Burning a Fuel

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The present invention relates to a method for burning nitrogen-containing fuels and apparatus therefor.

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Air pollution has become a major problem in the United States and other highly industrialized countries of the world. Consequently, the control and/or reduction of said pollution has become the object of major research and development efforts by both governmental and nongovernmental agencies. It has

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been alleged, and there is supporting evidence, that automobiles employing

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conventional piston-type engines which burn hydrocarbon fuels are a major contributor to said pollution. Vehicle emission standards have been set by the United States Environmental Protection Agency (EPA) which are sufficiently restrictive to cause automobile manufacturers to consider employing alternate engines instead of the conventional piston engine.

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Another source of such pollution is the exhaust gases from large stationary installations such as boilers in power plants, and large stationary gas turbine engines employed as a driving force in power plants and other large installations. It is anticipated that this portion of the problem will almost certainly be aggravated in the relatively near future by the necessity to use lower quality fuels such as heavy petroleum oils, shale oils, coal liquids, etc. which contain relatively large amounts of fuel-nitrogen, e.g., up to about 2 weight percent or greater, as compared to presently available fuels which contain very little, if any, fuel-nitrogen. For example, #4 and #6 petroleum oils contain about 0.1 to 0.5 weight percent nitrogen, two typical crude shale oils, hereinafter referred to in the specific examples, contain 1.85 and 1.93 weight percent of chemically bound nitrogen, respectively, and a typical crude, solvent refined coal oil contains from about 1.0 to 1.5 weight percent of chemically bound nitrogen. By comparison, a typical petroleum-derived #2 fuel oil contains about 0.024 weight percent nitrogen. If all of the chemically bound nitrogen in the fuel is converted to nitrogen oxides, generally referred to as "fuel NO<sub>x</sub>", 1 percent by weight of nitrogen in a solvent refined coal oil, has the potential to produce about 1.928 pounds/million Btu or 1,300 ppmv (parts per million by volume at 3 percent excess oxygen, dry) of nitrogen oxides (NO<sub>x</sub>) while 1.85 and 1.93 percent by weight of nitrogen in crude shale oils, will potentially produce about 3.288 and 3.440 pounds/million Btu (2595 and 2642 ppmv), respectively. In addition, nitrogen oxides, produced by the hot-air reactions at flame temperatures, and referred to as "thermal NO<sub>x</sub>", also contribute to the total NO<sub>x</sub> pollutants in the flue gases from a combustion process.

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The federal limit for the discharge of  $\text{NO}_x$  pollutants into the atmosphere from steam generators burning liquid fossil fuel (1974 EPA New Source Performance Standards [NSPS]) is 0.30 pounds/million Btu (about 230 to 237 ppmv for typical shale oils). Some State limitations are even more stringent, for example the California standard is 225 ppmv. These limits include both fuel  $\text{NO}_x$  and thermal  $\text{NO}_x$ . While these standards can be met when burning low nitrogen (below about 0.1%), petroleum-derived fuel oils, serious complications are encountered when high nitrogen fuels such as heavy petroleum-derived fuel oils, crude shale oils and crude coal oils are burned under conventional utility boilers. For example, since thermal  $\text{NO}_x$  increases with temperature, modern utility boilers, which preheat the combustion-supporting air to 600-800°F for improved efficiency, produce thermal  $\text{NO}_x$  alone which can approach the specified emission standards. Consequently, in order to meet these standards, the conversion of fuel nitrogen to  $\text{NO}_x$  emissions, in a fuel having about 2.0 weight percent bound nitrogen, should not be more than about 5 percent. It has been reported in the literature that, when shale oils with about 2.0 weight percent nitrogen are burned in a stationary boiler of an electrical generating station,  $\text{NO}_x$  emissions on the order of 700 to 900 ppmv can be anticipated and, when solvent refined coal oils, with slightly more than 1.0 weight percent nitrogen, are burned, at least 20 to 50 percent of the fuel nitrogen is converted to  $\text{NO}_x$  emissions (260 to 650 ppmv).

While it has been suggested that high levels of fuel nitrogen can be reduced by severe hydro-treating, such techniques have not been commercially developed and, even if available, pilot plant tests indicate that such refining of crude shale oils and crude coal oils would increase costs by about \$3.00 to \$5.00 per barrel.

It has also been suggested that  $\text{NO}_x$  emissions from crude, high nitrogen fuels can be reduced by blending the high nitrogen fuel with low nitrogen petroleum-derived fuel oils or by burning crude, high nitrogen fuels in selected burners of a boiler while burning low nitrogen petroleum-derived fuel oils in

other burners. In addition to requiring substantial volumes of petroleum-derived fuel oils, these techniques require additional equipment for handling and blending and/or feeding two separate fuels.

5 Another suggestion for reducing NO<sub>x</sub> emissions from high nitrogen fuels is the addition of a fuel additive, such as an additive containing manganese. Obviously, this technique adds the cost of the additive to the operation and requires facilities for handling and blending the additive.

10 To date the most promising technique has been a two-stage, rich-lean combustion process, in which a primary combustion zone is operated fuel-rich and a secondary combustion region is operated fuel-lean. However, even with the best of these techniques only limited success has been attained and the conversion of fuel-nitrogen to NO<sub>x</sub> emissions is still well above the governmental limits. For example, it has been reported that the NO<sub>x</sub> level, when burning crude, solvent refined coal oil containing 1.12 weight percent nitrogen, cannot  
15 be reduced below about 0.4 lb./million Btu without resorting to the additional techniques of blending with low nitrogen, petroleum-derived fuels or using additives. Similarly, the best of the two-stage techniques, heretofore available, have failed to reduce NO<sub>x</sub> emissions from crude shale oils, containing about 2.0 weight percent nitrogen, to values less than about twice the governmental  
20 maximums.

The major problems in past operations of two-stage, rich-lean combustion processes has been the failure to recognize that control of factors other than the rich fuel-air ratio is necessary in order to attain minimum NO<sub>x</sub> production, and that the efficiency of fuel utilization and control of other  
25 pollutants must also be taken into consideration. The almost universal thinking of those skilled in the art has been that, so long as combustion is initiated in a fuel-air mixture having a fuel-air ratio above the stoichiometric ratio (usually referred to as a fuel-air equivalence ratio above 1.0) and the flame is thereafter diluted with air to reduce the overall fuel-air ratio to below the

stoichiometric ratio (a fuel-air equivalence ratio below 1.0), this is all that is necessary in order to accomplish the desired result. The ultimate efficiency of fuel utilization and the volume of unburned or partially burned fuel (generally indicated by the HC and CO content of the flue gases) has also been generally ignored by such prior art investigators.

Summary of the Invention

It is therefore an object of the present invention to overcome these shortcomings of the prior art.

Yet another object of the present invention is to provide an improved method and apparatus for burning high nitrogen fuels.

Another and further object of the present invention is to provide an improved method for the combustion of high nitrogen fuels in which the  $\text{NO}_x$  content of the flue gases is substantially reduced.

Yet another object of the present invention is to provide an improved method and apparatus for the combustion of high nitrogen fuels wherein the  $\text{NO}_x$  content is substantially reduced while concomitantly maintaining high combustion efficiency.

A further object of the present invention is to provide an improved two-stage, rich-lean combustion method and apparatus for burning high nitrogen fuels.

Still another object of the present invention is to provide an improved method for initiating and maintaining an effective and efficient, two-stage, rich-lean combustion process for burning high nitrogen fuels.

As used herein and in the claims, unless otherwise specified, the terms "fuel-nitrogen", "chemically bound nitrogen", "organic nitrogen", and similar terms are employed to refer to nitrogen which is chemically bound into the fuel molecule. Also, as used herein and in the claims, unless otherwise specified, the terms "fuel-air equivalence ratio", "equivalence ratio" and the symbol " $\phi$ " are employed to refer to the ratio of the fuel flow (fuel available)

to the fuel required for stoichiometric combustion with the air available. Stated another way, said equivalence ratio is the ratio of the actual fuel-air mixture to the stoichiometric fuel-air mixture. For example, an equivalence ratio of 1.5 means the fuel-air mixture in the zone is fuel-rich and contains 1.5 times as much fuel as a stoichiometric mixture.

In accordance with the present invention, fuels containing chemically bound nitrogen are burned, in a two-stage, rich-lean combustion process, by introducing the fuel and at least one stream of primary air into a primary combustion region at a fuel-air ratio above the stoichiometric ratio and in a manner to intimately mix the fuel and air and establish a stabilized flame adjacent the upstream end of the primary combustion region; maintaining the flame in the primary combustion region for a period of time sufficient to produce a combustion product mixture containing less than a predetermined amount of  $\text{NO}_x$  pollutants and abruptly terminating the primary combustion region while introducing at least one stream of secondary air into a secondary combustion region in an amount sufficient to reduce the overall fuel-air ratio below the stoichiometric ratio and in a manner to prevent backflow of the secondary air into the primary combustion region.

#### Brief Description of the Drawings

FIGURE 1 is a view, partially in cross section, of a combustor, e.g., Combustor A, in accordance with the invention.

FIGURE 2 is an enlarged view of the upstream end portion of the combustor of FIGURE 1.

FIGURE 3 is an enlarged view, in elevation, taken along the line 3-3 of FIGURE 2 and illustrating one set of tangential entry ports or slots.

FIGURE 4 is an enlarged view, in elevation, taken along the line 4-4 of FIGURE 2 and illustrating another set of tangential entry ports or slots.

FIGURE 5 is a view, in elevation, taken along the line 5-5 of FIGURE 1.

FIGURE 6 is a view looking at the upstream side of a variable dome member, as shown in FIGURE 2, and which can be employed in the combustors of the invention.

FIGURE 7 is an enlarged view in elevation of an element of the dome member shown in FIGURE 6.

FIGURE 8 is an enlarged view in elevation of another element of the dome member shown in FIGURE 6.

5           FIGURE 9 is a schematic representation of the combustor of FIGURE 1 and sets forth certain design characteristics of said combustor.

FIGURE 10 is an enlarged view, partially in cross section, of a modification of the upstream end portion of the combustor of FIGURES 1 and 2.

10           FIGURE 14 is a perspective view, partially in cross section, of the variable dome of FIGURES 11, 12 and 13, mounted in the combustor of FIGURE 10.

FIGURE 15 is an enlarged view, partially in cross section, of the upstream end portion of another combustor, in accordance with the invention.

FIGURE 16 is an enlarged view, partially in cross section, of the upstream end portion of yet another combustor, in accordance with the invention.

15           FIGURE 17 is a cross-sectional view of the upstream end portion of the combustor of FIGURE 15 together with a partial view of a modified central portion of a combustor which can be utilized with any of the previously illustrated upstream portions of combustors in accordance with the invention.

20           FIGURES 18 and 19 are partial cross-sectional views of two modified central portions of combustors useful in accordance with the invention.

FIGURE 20 is a partial cross-sectional view of a nozzle useful in the combustors of the invention.

FIGURES 21 and 22 are cross-sectional views, taken along the lines 21-21 and 22-22, respectively, of FIGURE 20.

25           FIGURE 23 is a diagrammatic illustration of a combustor test installation and includes Combustor X, another combustor in accordance with the invention.

FIGURE 24 is a diagrammatic illustration in cross section of said Combustor X, shown above in FIGURE 23.

30           FIGURES 25-29 are cross-sectional, schematic illustrations of a series of burners of the present invention adapted for use in or as process heaters, industrial furnaces, boilers.



FIGURE 30 is a diagrammatic view in elevation illustrating a combustor(s) of the invention installed in the wall of a power plant boiler and employed in combination with said boiler.

FIGURES 31 and 32 are elevational and top cross-sectional schematic views of a process heater utilizing burners of the character shown in FIGURES 25-29.

FIGURE 33 is a cross-sectional schematic view of a test furnace incorporating a burner as illustrated in FIGURES 27 and 28.

FIGURES 34 and 35 are schematic plan and elevational views, respectively, partially in section, of the air supply system utilized for the test furnace of FIGURE 33.

FIGURE 36 is a cross-sectional, schematic view of a test furnace incorporating a burner as illustrated in FIGURES 27 and 28.

FIGURE 37 is a schematic plan view, partially in section, of the air supply system for the test furnace of FIGURE 36.

FIGURES 38-47, inclusive, are data curves setting forth the results obtained in the test runs of the working examples given hereinafter.

FIGURES 48-61, inclusive, are data curves setting forth the results obtained in other test runs of the working examples given hereinafter.

FIGURES 62-69, inclusive, are data curves setting forth the results obtained in additional working examples of the invention.

FIGURES 70-77 are data curves setting forth the results of prototype tests conducted in the test furnaces of FIGURES 33 and 36.

#### Description of the Preferred Embodiments

In its more specific aspects the present invention relates to an improved method of operating a two-stage, rich-lean process for the burning of fuels, containing a high concentration of chemically bound nitrogen, particularly heavy fuels, such as shale oils and coal-derived oils, improved methods of initiating and maintaining such combustion and improved apparatus therefor.

In a still more specific aspect, the improved method includes the controlled mixing of a predetermined fuel-rich ratio of fuel and primary air in the upstream end of a primary combustion region to intimately mix the fuel and air and establish a stabilized flame adjacent the upstream end of the primary combustion region and maintenance of the flame in the primary combustion region for a predetermined minimum residence time sufficient to produce a primary region combustion product mixture containing a preselected maximum amount of NO<sub>x</sub> pollutants; and abrupt termination of the primary combustion region along with the introduction of secondary air into a secondary combustion region in an amount sufficient to produce an overall fuel-lean ratio of fuel and air and in a manner to prevent backflow of the secondary air from the secondary combustion region into the primary combustion region.

Various means may be utilized to initially mix the fuel and the air and establish a stabilized flame in the upstream end of the primary combustion

region. Preferably, a generally axial flow component is established and a generally radial flow component is established at a selected critical ratio.

5 The axial flow component comprises the fuel or a mixture of the fuel and a portion of the primary air, introduced in a generally axial direction, or one of these in combination with at least one separate stream of primary air, introduced in a generally radial direction. The fuel or fuel-air mixture is preferably introduced as an open cone having a 90° apex angle. When an additional air stream is introduced axially the air is preferably introduced about the fuel or fuel-air mixture so as to contact the fuel adjacent its point of  
10 introduction. The latter is preferably accomplished by introducing the fuel or fuel-air mixture as an open cone. The separate air stream may be introduced in a strict axial direction or as a swirling stream in a generally axial direction. Finally, the separate stream of air may be introduced through fixed area openings or through variable area openings to thus introduce a fixed or a variable  
15 amount of air in the axial direction.

The radial flow component comprises the remainder of the primary air introduced from a peripheral location about the fuel and in a direction generally inwardly toward the axis of the combustion zone to contact the axial flow component and produce a combustible mixture of fuel and air. The radially  
20 introduced primary air may be introduced in a strict radial direction or in a generally radial direction as a swirling stream and can comprise a plurality of separate streams spaced along the upstream end of the primary combustion region and any combination of strict radial or swirling streams. For example, a single swirling stream, plural swirling streams, such as two swirling streams  
25 rotating in opposite directions, a single strict radial stream, plural strict radial streams, a single strict radial stream and one or more swirling streams, etc. In the event that no axial air stream is introduced or the axial air stream is not variable, it is preferred that one of the radial air streams be variable. Best results are obtained if the most upstream radial air stream is

variable. In any case, the most upstream radial airstream should contact the fuel substantially immediately following the point of introduction of the fuel.

5 As indicated, the manner of introducing the fuel and primary air can assume various forms so long as an axial flow component and a radial flow component is established. The ratio of the axial flow component and the radial flow component is also a significant factor in obtaining minimum  $\text{NO}_x$  pollutant production. While the ratio of the axial flow component and the radial flow component can be attained and maintained by externally adjusting the relative volumes of fuel and/or air introduced in the axial and radial directions, it is most effective and convenient to have either a variable axial or radial air stream and adjust the volume of the variable air stream.

10 Also, irrespective of the manner of introducing the fuel and primary air and irrespective of the adjustment of the variable air stream, it is preferred that the relative volumes of fuel and total primary air, introduced into the primary combustion region, be sufficient to establish and maintain a selected critical primary region fuel-air equivalence ratio between 1.0 and about 1.8, preferably between about 1.2 and about 1.7. The selected primary region fuel-air equivalence ratio should be correlated with the ratio of the axial flow component and the radial flow component. This is best accomplished by varying the ratio of the axial flow component and the radial flow component at a plurality of primary region fuel-air equivalence ratios within the stated range until a minimum value of  $\text{NO}_x$  pollutants is attained.

20 In one aspect of the invention the mixture of fuel and air is constricted to an axial stream having a diameter smaller than the diameter of the primary region and is then expanded, either gradually or abruptly, to the full diameter of the primary region. Preferably, such constriction occurs immediately after the introduction of the most upstream radial air stream. More downstream ones of the streams of air may be introduced either into the thus constricted stream or during the expansion thereof.

In connection with one embodiment of the invention, where primary air is introduced in a swirling or tangential manner, the percent swirl or tangential air (of the total primary air) introduced downstream from the variable air is desirably maintained between about 35 and 100 percent by volume of the total primary air. It is also desirable where two swirl air streams are utilized, that the more downstream one of the two be larger than the other.

The flame thus established and stabilized adjacent the upstream end of the primary combustion region is maintained in the primary combustion region for a period of time sufficient to prevent the conversion of fuel-nitrogen to  $\text{NO}_x$  pollutants from exceeding a predetermined maximum amount. The preselected amount of fuel  $\text{NO}_x$  pollutants produced by fuel-nitrogen conversion is less than about 10 percent. In terms of lbs.  $\text{NO}_x$ /million Btu, this amounts to less than about 0.350 lb. fuel  $\text{NO}_x$ /million Btu for most fuels and a total fuel and thermal  $\text{NO}_x$  of about 0.450 lb./MM Btu. The residence time necessary to accomplish this is usually at least about 35 milliseconds. Preferably, the conversion of fuel-nitrogen to fuel  $\text{NO}_x$  pollutants is less than about 7.5 percent and ideally below about 5 percent (about 0.270 and 0.180 lb. fuel  $\text{NO}_x$  or 0.365 and 0.275 lb. total  $\text{NO}_x$ /MM Btu, respectively). It has been found, in accordance with the present invention, that no significant decrease in the  $\text{NO}_x$  pollutant content of the flue gas is attained when the residence time in the primary combustion region is above about 100 milliseconds. Therefore, as a practical matter, in order to maintain the overall length of the combustor within reasonable limits and maintain heat losses at a minimum, the residence time in the primary combustion region should not exceed about 140 milliseconds and preferably about 120 milliseconds.

It has also been found, however, that once maximum  $\text{NO}_x$  pollutant reduction has been attained, the primary combustion region must be immediately and abruptly terminated. This also may be accomplished in several different ways. In general, at least one stream of secondary air is introduced into the secondary combustion region in an amount sufficient to reduce the overall fuel-air equivalence ratio of the combustion zone below the stoichiometric ratio

(preferably about 0.87) and in a manner such that backflow of secondary air into the primary combustion zone is prevented. The latter may be accomplished by reducing the diameter of the flame path by mechanical means, such as by locating at least one annular baffle or a nozzle-type constriction immediately adjacent and, preferably, downstream from the point of introduction of the secondary air. The flame, thus diluted with secondary air, is then rapidly expanded into the secondary combustion region to attain intimate mixing of the combustion products of the primary combustion region and the secondary air.

It has been found that, if the primary region is not abruptly terminated, the secondary air will tend to backflow into the primary region, thus counteracting the advantage of the primary region and upsetting the critical ratio of the axial flow component and the radial flow component, the critical primary region fuel-air equivalence ratio and the critical primary region residence time, and/or a gradual transition from the primary region to the secondary region results. Such backflow of secondary air, gradual transition from the primary to the secondary region and/or extension of the primary region residence time beyond the limits previously discussed will result in significant increases in the  $\text{NO}_x$  pollutant level, inordinately large volumes of CO in the flue gas and/or incomplete and inefficient burning of the fuel.

It has also been found, in accordance with the present invention, that the residence time in the secondary combustion region should be greater than about 15 milliseconds. This minimum secondary region residence time and the intimate mixing of the combustion products of the primary combustion region and secondary air are necessary in order to complete combustion of unburned fuel and partially burned fuel which result from the fuel-rich combustion in the primary combustion region. It has been found that a residence time of 15 milliseconds or greater will result in a carbon monoxide content in the flue gas of less than about 300 ppm by volume. Such a level of CO in the flue gas indicates that combustion has been essentially complete and efficient and effective use of the fuel has been attained.

In addition to controlling the previously-mentioned process variables, it has further been discovered that such variables are interdependent and should be correlated with one another in practicing the present invention. Specifically, the manner of introducing and contacting the fuel and primary air in the primary combustion region, the fuel-air equivalence ratio in the primary combustion region, the flame residence time in the primary combustion region and the manner of terminating the primary combustion region, including the volume and manner of introducing secondary air, must all be controlled and coordinated with one another. As will be shown by the examples hereinafter set forth, the nature of the fuel, particularly its volatility and nitrogen content, and the burner type and configuration cause the optimum values of these variables to change, both as to absolute values as well as in their relation to one another, thereby further emphasizing the necessity of proper control of such variables and their relationship to one another.

This interdependence of the operating variables and the proper correlation of each with the other will be apparent from the following description of the method of initiating and maintaining combustion of a fuel, containing chemically bound nitrogen, in a two-stage, rich-lean combustor, in accordance with the present invention.

The combustion process is initiated by determining the primary combustion region residence time at which minimum  $\text{NO}_x$  pollutants are produced, determining the primary combustion region fuel-air equivalence ratio at which minimum  $\text{NO}_x$  pollutants are produced and determining the ratio of the axial flow component and the radial flow component at which minimum  $\text{NO}_x$  pollutants are produced. In certain instances, the secondary combustion region fuel-air equivalence ratio and residence time at which predetermined maximum amounts of CO and/or unburned and partially burned fuel are produced should also be determined. The order of making the specified determinations may be varied and any two or more of said determinations may be made simultaneously without departing from

the invention. Once the combustion process is initiated, such combustion may be maintained by making one or more of the specified determinations and adjusting the corresponding process variable to a value at which minimum  $\text{NO}_x$  pollutants are produced and/or the predetermined maximum amounts of CO and/or unburned and partially burned fuel are produced.

The primary combustion region residence time is dependent upon two major factors, namely; the length of the primary combustion region and the primary combustion region flow velocity. While the primary combustion region length, that is the distance between the tip of the fuel injection nozzle and the midpoint of the means for introducing secondary air, will generally be fixed, there are certain minimal requirements in accordance with the present invention. Specifically, the primary combustion region length should be selected such that a minimum primary combustion region residence time of at least about 35 milliseconds can be attained at practical primary combustion region flow velocities. As a practical matter, the maximum length of the primary combustion region will be limited to some extent by the particular environment in which the burner is to be utilized and the energy (heat) loss resulting from the utilization of the primary combustion region. Accordingly, the length of the primary combustion region should be as short as possible. The maximum length should therefore be such that a maximum residence time of about 140 milliseconds, and preferably a maximum residence time of 120 milliseconds will be attained. For example, in a combustor having an internal diameter of 6 inches, a length of about 58 inches is considered adequate. For combustors of larger or smaller diameter the length would be determined by utilizing approximately the same ratio of length to diameter. Having thus set the length of the primary combustion region, the primary combustion region residence time at which minimum  $\text{NO}_x$  pollutants are produced is determined by varying the flow velocity at each of a plurality of primary combustion region fuel-air equivalence ratios between 1.0 and about 1.8. The velocity should be varied within practical limits.



Obviously, if the velocity is too low, mixing of the fuel and air will be poor, thus producing a nonhomogeneous or heterogeneous mixture, and, if the velocity is too high, the flame will go out or will not stabilize adjacent the upstream end of the primary zone. While such velocity may be varied between about 1 and 500 feet/second, practical flow velocities are between about 5 and 100 feet/second, and preferably between about 5 and 50 feet/second.

It is apparent that the primary combustion region fuel-air equivalence ratio at which minimum  $\text{NO}_x$  pollutants will be produced will be determined to some extent simultaneously with the determination of the primary combustion region residence time. However, this equivalence ratio is generally a rough approximation. Therefore, the primary region equivalence ratio at which minimum  $\text{NO}_x$  pollutants are produced is preferably determined by varying the equivalence ratio between 1.0 and about 1.8 at the determined primary combustion region residence time.

The ratio of the axial flow component and the radial flow component can then be determined by varying said ratio at the determined primary region residence time and the determined primary region fuel-air equivalence ratio. However, it is preferred that the ratio of the axial flow component and the radial flow component be varied at each of a plurality of different primary region fuel-air equivalence ratios between 1.0 and about 1.8, to thereby simultaneously determine the ratio of the axial flow component and the radial flow component and the primary region fuel-air equivalence ratio at which minimum  $\text{NO}_x$  pollutants are produced. As a practical matter, the preferred combustor, in accordance with the present invention, includes a variable dome surrounding the fuel injection nozzle and having axially or radially directed openings which may be adjusted. Accordingly, by simply varying the size of the dome openings, the ratio of the axial flow component and the radial flow component can be varied. In those instances where primary air is introduced at one or more points downstream from the variable primary air introduction point, variation of the dome

openings will also vary the relative volumes of primary air introduced at the two points, since all of the primary air is introduced from a single source, the primary combustion region fuel-air equivalence ratio is set and the sizes of the primary air introduction ports downstream of the variable primary air introduction point are fixed. This is an important factor, since it has also been found that, when the primary air introduced downstream from the variable primary air introduction point is introduced in a swirling or tangential manner, the volume of air introduced through the swirl ports should be maintained between about 35 and 100 percent of the total air introduced in the primary combustion region.

The secondary combustion region residence time can be determined in the same manner as the primary region residence time. Specifically, the length of the secondary combustion region should be selected to produce a residence time of at least about 15 milliseconds at practical secondary combustion region flow velocities. At this equivalence ratio substantially complete combustion of unburned and partially burned fuel from the primary combustion region will be attained. This criterion will generally be met if the flue gas contains less than about 300 ppmv of CO emissions. For a 6-inch internal diameter combustor, a secondary combustion region length of about 33 inches will be adequate at the determined primary combustion region flow velocity. Since no significant reduction in CO emissions occurs at residence times beyond about 30 milliseconds, and preferably beyond about 25 milliseconds, the length of the secondary combustion region need not be greater than a value necessary to attain this residence time. Where the combustor is utilized in a boiler of an electrical generating plant or a like installation, the flame will be expanded directly into the heated region of the boiler after constriction and dilution with secondary air, as specified herein. In this case, the minimum secondary combustion region residence time will be readily met by the volume of the heated region and the total combustion zone can be shortened to a length just sufficient to include



the means for constricting the flame and introducing secondary air. Preferably, the secondary combustion region, in this case, should also include a short section of the flame tube to assure intimate mixing of the secondary air with the combustion products from the primary combustion region.

5           The secondary combustion region fuel-air equivalence ratio is determined by varying the equivalence ratio between about 1.0 and 0.5 while maintaining the previously determined variables constant, and selecting the equivalence ratio at which the CO emissions are below about 300 ppmv. As a practical matter, a secondary combustion region equivalence ratio of about 0.87 will be used, 10 since this represents a value of 3 percent excess oxygen (dry basis) in the flue gas,  $\text{NO}_x$  emission limitations are generally measured at this level and this ratio has been found to produce CO emissions below the specified maximum.

To the extent that more definitive values of the various variables are desired, a curve of the variable versus minimum  $\text{NO}_x$ , HC or CO emissions can be 15 plotted and the critical value of the variable may be selected as the minimum  $\text{NO}_x$ , HC or CO point on the curve.

Once the combustion process has been initiated, as previously discussed, there should be no need to make further changes, so long as the same fuel is utilized. However, should the measured  $\text{NO}_x$  emissions and/or CO emissions 20 increase significantly from those initially determined, appropriate adjustments of the previously discussed variables should be made. While any one or more of the primary combustion region variables may be adjusted, it will usually be adequate to adjust the variable primary air, since the volatility of the fuel is most likely to vary and such adjustment of the variable primary air will alter 25 the fuel-air mixing and compensate for variations in fuel volatility. However, should the bound nitrogen content of the fuel change, the primary combustion region fuel-air equivalence ratio should also be adjusted and possibly the primary zone residence time.

Adjustments of the primary region fuel-air equivalence ratio and the 30 ratio of the axial flow component and the radial flow component can be simultaneously made by adjusting the dome openings at each of a plurality of different

primary combustion region fuel-air equivalence ratios on either side of the original ratio. If it is known, by test, that the nitrogen content is higher than the initial value, it will generally be sufficient to adjust the dome openings at each of a series of equivalence ratios above the original ratio. On the other hand, if it is known, by test, that the nitrogen content is lower than the initial value, it will usually be sufficient to adjust the dome openings at a plurality of different equivalence ratios below the original.

Referring now to the drawings, wherein like or similar reference numerals are employed to denote like or similar elements, the invention will be more fully explained.

FIGURES 1-5, inclusive, illustrate a combustor in accordance with the invention. Preferably, said combustor comprises a flame tube having an upstream fuel-air mixing section 11, an intermediate primary combustion section or region 13 located downstream from and in communication with said mixing section 11, and a secondary combustion section or region 15 located downstream from and in communication with said primary combustion section 13. Said flame tube is provided at its upstream end with a dome member 17. A fuel inlet means 24 is provided for introducing a stream of fuel into the upstream end portion of said mixing section 11. As illustrated in FIGURES 1 and 2, said fuel inlet means comprises a fuel conduit 19 leading from a source of fuel and extending through fuel flange 21 into communication with the central cavity formed in the downstream side of dome member 17 and which is adapted to receive fuel nozzle 24 mounted therein. An annular orifice means is disposed on the downstream side of said dome member 17. Said orifice means can be formed integrally with said dome member or, as here illustrated, can preferably comprise an annular adapter 26 disposed between the downstream end of said dome member 17 and the upstream end of said flame tube. An orifice formed in said orifice means or adapter 26 can be considered to define the outlet from said dome member 17 and the inlet into said mixing section 11.

A first air inlet means is disposed in the wall of said flame tube for admitting a first stream of air into said mixing section 11 in a circular-like direction adjacent the inner wall thereof. Said first air inlet means preferably comprises a plurality of tangential slots 28 extending through the wall of the upstream end portion of said mixing section 11 at a first station in the flame tube adjacent said outlet from said dome member 17 and upstream from a first orifice 29. In a preferred embodiment a second air inlet means is disposed in the wall of said flame tube downstream from said first air inlet means for admitting a second stream of air into said mixing section 11 in a circular-like direction adjacent the inner wall thereof. Said second air inlet means preferably comprises a plurality of tangential slots 30 extending through the wall of the downstream end portion of said mixing section 11 at a second station in the flame tube adjacent and downstream from said first orifice 29. A second orifice 32 is disposed in said flame tube adjacent and downstream from said tangential slots 30. A third air inlet means, comprising at least one opening 34, is provided in the wall of said flame tube at a third station downstream from said second air inlet means 30 and said second orifice 32 for admitting a stream of air comprising secondary air into said secondary combustion section 15. If desired or necessary diluent air can be admitted through ports 40.

A fourth air inlet means, preferably variable, is provided in said dome member for admitting a variable volume of a third stream of air through said dome member, around said fuel inlet nozzle 24, and into said mixing section 11 of said flame tube. As described further hereinafter, said variable air inlet means comprises at least one air passage means of variable cross-sectional area provided in and extending through said dome member 17 into communication with said mixing section 11, and means for varying the cross-sectional area of said air passage means and thus controlling the volume of said stream of air admitted through said dome member and into said mixing section.

In some embodiments of the invention wherein a said second air inlet means, e.g., tangential slots 30, is not provided, said first orifice 29 can be

considered to define the outlet from said mixing section 11 and the inlet to said primary combustion section or region 13. In those preferred embodiments wherein a said second air inlet means, e.g., tangential slots 30, is provided, said second orifice 32 can be considered to define the outlet from said mixing section 11 and the inlet to said primary combustion section or region 13.

Said flame tube can be fabricated integrally if desired. However, for convenience in fabrication, said flame tube can preferably be formed with the wall portion thereof which comprises said mixing section 11 divided into separate sections similarly as here illustrated. Thus, in one preferred embodiment said tangential slots 28 can be formed in an upstream first wall section 36 of said flame tube, preferably in the upstream end portion of said first wall section with the downstream wall of said adapter 26 forming the upstream walls of said slots 28. In this preferred embodiment said first orifice 29 is formed in the downstream end portion of said first wall section 36. In said preferred embodiment said tangential slots 30 can be formed in an intermediate second wall section 38 located adjacent and downstream from said first wall section 36. Preferably, said second wall section 38 is disposed with its upstream edge contiguous to the downstream edge of said first wall section 36, and said tangential slots 30 are formed in the upstream end portion of said second wall section 38 with the downstream edge of said first wall section 36 forming the upstream walls of said slots 30. In this preferred embodiment said second orifice 32 is formed in said second wall section 38 and adjoins said slots 30 formed therein. Preferably, the inner wall surface of said first wall section 36 tapers inwardly from the downstream edge of said tangential slots 28 to the upstream edge of said first orifice 29 to form an inwardly tapered passageway from said slots to said orifice. Preferably, the inner wall surface of said second wall section 38 extends radially outwardly from the downstream edge of said second orifice 32. The remainder of said flame tube can conveniently comprise the flanged spools or sections 10, 12, 14, 16, 18, 20, and 22 as illustrated in FIGURE 1. For convenience, said spools 14 and 18 were provided

with jackets for circulation of water or other coolant therethrough. Such cooling permitted operation of the combustor through a wide range of operating conditions including stoichiometric fuel-air mixtures without heat damage. A small amount of wall cooling air, admitted through ports 42, was used to cool the unjacketed spool 12.

In one presently preferred embodiment a third orifice means 46 is disposed at an intermediate location in said primary combustion section and downstream of said second air inlet means. Preferably, a fourth orifice means 48 is disposed in said primary combustion section upstream from and adjacent said third air inlet means. A fifth orifice means 50 is disposed in the upstream end portion of said secondary combustion section downstream from and adjacent said third air inlet means. Said third, fourth, and fifth orifice means aid in the mixing of the gases flowing through the flame tube and promote the homogeneity thereof.

It will be understood that the combustors described herein can be provided with any suitable type of ignition means and, if desired, means for introducing a pilot fuel to initiate combustion. For example, a sparkplug (not shown) can be mounted to extend into first combustion region 13.

Referring to FIGURE 1, for example, for the purpose of calculating residence time in connection with the method of the invention, the primary combustion region can be considered to be the region from the downstream tip of fuel nozzle 24 to the midpoint of the air inlet ports 34, and the secondary combustion region can be considered to be the region from the midpoint of said ports 34 to the midpoint of the openings 40.

Said first orifice 29 and said second orifice 32 have been illustrated as being circular in shape and this is usually preferred. However, it is within the scope of the invention for either or both of said orifices to have other shapes, e.g., triangular. Said flame tube has been illustrated as being cylindrical in shape and this is usually preferred. However, it is within the scope of the invention for said flame tube to have any other suitable shape.

Referring to FIGURES 6, 7, and 8, a variable dome member 17 comprises a fixed circular back plate 128 centrally mounted in an opening 138 provided in fuel flange 21 by means of a pair of mounting bars 132. A plurality of spaced apart openings 134, arranged in a circle, are provided in said plate 128. A stop  
5 pin 136 projects perpendicularly from one of said bars 132. Said opening 138 in fuel flange 21 is in communication with air supply conduit 44 (see FIGURE 2). A centrally disposed circular boss member 140 projects outwardly from the upstream face of said fixed plate 128 for receiving and mounting a front adjustable plate 142 thereon.

10 Said front plate 142 is circular-like, and of the same size as, said fixed plate 128. A plurality of spaced apart openings 144 are provided in said front plate 142 and correspond in size and circular arrangement to that of said openings 134 in backplate 128. A pair of spaced apart stop pins 146 project perpendicularly from the side of said front plate 142. An actuator tab 148  
15 projects perpendicularly from one side of said front plate at a location spaced from said stop pins 146. Push rod 150 is pivotally connected to said actuator tab 148 in any suitable manner as shown. Said push rod 150 can be actuated in a back and forth manner by means of roller mechanism 152 mounted on the outside of fuel flange 21 in any suitable manner. Flexible shaft 154 extends through a  
20 control panel (not shown) and is connected to a rotatable knot (not shown) for movement of said shaft 154, said roller mechanism 152, and said rod 150 for rotating said front plate 142 within the limits imposed by stop pins 146 acting against stop pin 136.

In assembly, said fuel flange 21 is mounted between adjacent flanges  
25 as shown in FIGURE 2. The upstream end of the flame tube fits to adapter 26 which in turn is secured to the downstream face of dome member 17. Fuel conduit 19 extends through said flange 21 and communicates with a central cavity formed in the downstream side of dome member 17 which is adapted to receive fuel nozzle 24 mounted therein. The central opening 156 in front plate 142 fits into  
30 boss member 140 on backplate 128 and said front plate is held in sliding engagement with backplate 128 by means of cap screw 158 and washer 160. Said push



rod 150, by virtue of the back and forth movement described above, rotates said front plate 142 to bring openings 144 therein into and out of register with openings 134 in said backplate 128 to thus vary the effective size of opening provided in variable dome 17 and vary the amount of air passed through said dome into mixing section 11 and then into primary combustion region 13. As shown in FIGURE 2, said openings 144 and 134 are in full register and the dome member is completely open. As shown in FIGURE 6, said openings are out of register and the dome member is completely closed.

In the practice of the invention, it is sometimes desirable to control the effective size of the openings in the variable dome 17 of the combustors of the invention in accordance with fuel flow to the combustor. This can be accomplished manually by means of the push rod 150 and associated elements. However, in continuously operating combustors which operate over a varied range of operating conditions, it is sometimes desirable that the effective size of the dome openings be controlled automatically. Any suitable control means can be provided for this purpose, for example, the control means diagrammatically illustrated in FIGURE 6. Said control means can be adapted to the combustor of FIGURE 1 by providing an orifice in fuel conduit 19, operatively connecting said orifice to a controller unit 109, and operatively connecting said controller unit by a suitable linkage 110, to shaft 154 of rack and roller mechanism 152 which moves push rod 150 back and forth. Thus, controller 109 responds to the flow of fuel through the orifice in conduit 19, actuates linkage 110, which is operatively connected to shaft 154, and programs the back and forth movement of rod 150. The specific control means comprising the orifice in fuel conduit 19, controller 109, and linkage 110 forms no part, per se, of the present invention. Said control means can be modified or substituted for by any means known in the art. An automatic control means such as described above can be employed on the combustors of the invention when said combustors are employed, for example, in a gas turbine engine, either stationary or mobile. In a mobile installation such as a vehicle, such a control means can be employed to vary the amount of air admitted through the dome member in accordance with fuel flow as said fuel flow changes with

charged a speed of the vehicle. In a stationary installation such as a large gas turbine engine driving a standby generator in a power plant, such a control means can be employed to vary the amount of air admitted through the dome member in accordance with fuel flow when the generator must be put on the line rapidly.

5           FIGURE 10 shows a modified dome member 17, mixing zone 11 and primary combustion region 13 for the combustor of FIGURES 1 through 5. In this modification, various elements have been simplified to make the combustor particularly suitable for the combustion of heavy oils, such as shale oil.

          Specifically, the nozzle 24 (referred to in detail hereinafter) is  
10 adapted to thoroughly mix the fuel in air for improved atomization of the fuel. The remainder of the primary air to the primary region 13 of the combustor is introduced through conduit 162. Conduit 162 thus supplies primary air through tangential slot means 28 and 30 and through variable air inlet means 164. FIGURE 40 also illustrates a suitable ignition means 166. Ignition means 166  
15 is mounted to communicate with mixing region 11 through a passage through dome member 17. Ignition means 166 includes a sparkplug 168, a fuel inlet 170 (for a fuel such as propane) and an air inlet 172.

          Referring to FIGURES 11, 12 and 13, the dome member can comprise a fixed generally cylindrical member 80 (see FIGURE 12) closed at one end and open  
20 at the other end. A plurality of openings 82 are provided at spaced apart locations around the circumference of said cylindrical member 80 adjacent the closed end thereof. An opening 84 is provided in said closed end for receiving a fuel nozzle. The outlet of said fuel nozzle would be positioned similarly as shown for nozzle 24 in FIGURE 2. Another opening 88 is provided in said closed  
25 end for receiving an igniter means (not shown) which would also extend to a position adjacent the outlet of the fuel nozzle. Openings 92 are provided for receiving mounting bolts (not shown) for mounting the dome member onto the central portion of a fuel flange such as fuel flange 21. Said central portion of the fuel flange would be adapted to accommodate the fuel nozzle similarly as

shown in FIGURE 2, and also the igniter means. A mounting flange 94 is connected to and provided around the open end of said cylindrical member 80 for mounting said member 80 on the upstream end of a combustor flame tube, similarly as shown in FIGURE 2. A groove 96 is provided in said flange 94 around the open base of said cylindrical member 80. A pair of spaced apart stop pins 98 project from said flange 94 perpendicular thereto and adjacent said cylindrical member 80. An orifice 95, preferably tapered inwardly, is provided in said flange 94 adjacent and in communication with the open end of said cylindrical member 80.

The adjustable throttle ring 100 of FIGURE 11 is mounted around said cylindrical member 80 and is provided with a plurality of spaced apart openings 102 therein of a size, number, and shape and at spaced apart locations, corresponding to said openings 82 in cylindrical member 80. Said throttle ring fits into groove 96 in flange 94. An actuator pin 104 projects outwardly from the outer surface of said throttle ring 100 and coacts with said stop pins 98 to limit the movement of said ring 100. Friction lugs 106 are provided on the top and the bottom of said ring 100 for movably bearing against the surface on which cylindrical member 80 is mounted, and the bottom of groove 96, respectively. FIGURE 13 is a cross section of ring 100 mounted on member 80.

Any suitable means can be provided for actuating actuator pin 104. Such actuating means can comprise a Y-shaped yoke which fits around actuator pin 104, with the bottom leg of the Y connected to a rotatable control rod which extends through the outer housing or casing of the combustor. Rotation of said control rod will pivot the Y-shaped yoke and coact with said pin 104 to cause rotation of throttle ring 100 within the limit of the space between stop pins 98 and thus adjust the register and effective size of the opening provided by openings 82 and 102. As shown in FIGURE 13, said openings 82 and 102 are in direct register with each other to provide the maximum opening into the dome. When flange 94 is mounted on the upstream end of a flame tube, such as the flame tube in FIGURE 1, air introduced through openings 82 and 102 will be introduced

radially, e.g., around and generally perpendicular to the direction of introduction of fuel. Said openings 82 and 102 have been illustrated as being circular, and this is usually preferred. However, said openings can be rectangular, e.g., square, if desired.

5           FIGURE 14 shows the variable air inlet means of FIGURES 11, 12 and 13 mounted in the dome 17 of the combustor shown in FIGURES 1 through 5 or that of FIGURE 10. FIGURE 14 also shows a suitable operating mechanism including pivot arm 174, control shaft 176 and dome rotation indicator 178. This particular variable dome member is preferred when burning heavy fuels.

10           FIGURE 15 of the drawings shows an alternative mixing-primary combustion region 11-13 which can be utilized in the practice of the present invention. The structure of FIGURE 15, in an overall sense, is similar to that of FIGURE 10 and, therefore, is also well suited for the combustion of heavy oils, such as shale oil.

15           The apparatus of FIGURE 15 differs from that of FIGURE 10 in the manner of introducing primary air and mixing the fuel and air. Specifically, a neck section 180 is formed downstream of nozzle 24 to form an annular primary air chamber 182. Neck means 180 diverges outwardly, in the downstream direction, to substantially the full diameter of the flame tube, thus forming flared or  
20           diverging section 184. All of the primary air, beyond that utilized in nozzle 24, is introduced through variable dome 164 and slots 186. This means of introduction of primary air causes the air to radially impinge against the stream of air-fuel from nozzle 24, thus bringing about intimate mixing and stabilization of the flame adjacent the upstream end of the primary combustion zone.

25           FIGURE 16 shows still another mixing-primary combustion region 11-13 which is adapted to attain intimate mixing of a heavy fuel, such as shale oil, and air and stabilize the flame adjacent the upstream end of the primary region.

          The combustor of FIGURE 16 is similar to the previous embodiment of  
30           FIGURE 15, except that a single primary air introduction means 188 (in addition to the air introduced through nozzle 24) is provided which combines the features

of a variable air inlet and tangential air introduction. More specifically, primary air introduction means 188 includes fins 190, disposed at 45° angles such that the air entering through the variable inlet enters the mixing region 11 tangentially or in a circular manner, as through the air inlet means 28 and 30 of FIGURES 1 through 5, 9 and 10. The structure of FIGURE 16 also includes a flared or diverging section 192, similar to section 184 of FIGURE 15, but shortens the neck section 180 of FIGURE 15.

In all of the embodiments of the mixing-primary combustion regions 11-13, previously discussed, intimate mixing is attained in the mixing region 11 and the flame is stabilized adjacent the upstream end of primary combustion region 13. However, it should be recognized that there is no sharp line of demarcation separating the mixing region 11 and the primary region 13, since, as previously stated, the upstream end of the primary combustion region 13 is considered to begin at the tip of the fuel nozzle 24 and the primary combustion region flame is established and stabilized at slightly different points in the upstream end of the primary combustion region in the several embodiments described. Hence, the mixing region 11 and the primary combustion region actually merge into a single multi-function mixing-primary combustion region, but, in all cases, the flame is stabilized "adjacent the upstream end of the primary combustion region". Viewed in another way, the upstream end of the primary combustion region can be said to differ in location for differing embodiments and, in essence, be located approximately at the point along the mixing region-primary combustion region 11-13 where the primary combustion region flame is stabilized. In this case, it is then most accurate to say that the flame is stabilized "adjacent the point where the mixing region 11 and the primary combustion region 13 merge."

By contrast to the mixing-primary combustion region 11-13, there is, in fact, a definite, sharp line of demarcation between the primary combustion region 13 and the secondary combustion region 15. This line of demarcation is "adjacent the point of introduction of secondary air." Previously discussed

FIGURES 1 and 10 and FIGURES 17, 18 and 19, illustrate several embodiments of combustors adapted to establish an abrupt change from the fuel-air ratio of the primary combustion region 13 to that of the secondary combustion region 15, while at the same time preventing back-flow of secondary air into the primary combustion region 13 and creating intimate mixing in the secondary combustion region 15.

In the embodiment of FIGURES 1 and 10, the above is accomplished by converging and then expanding the flame front by virtue of orifices 48 and 50 and the radial introduction of secondary air through radial air inlet ports 34 located between the two orifices.

FIGURE 17 shows a combustor with the specific mixing-primary combustion region 11-13 of FIGURE 15 but a modified means for terminating the primary combustion region. In this embodiment, the flame front is converged and then expanded by locating a nozzle in the combustor while simultaneously introducing secondary air in a radial fashion into the vena-contracta of the nozzle. Specifically, an annular, angular ring 194 converges, in a downstream direction, from the full diameter of the primary combustion region 13 to a reduced diameter. Secondary air is then introduced radially downstream of ring 194 through conduit 196 and air inlet ports 198. Still further downstream, an annular ring or flange 200, having an inside diameter equal to the inside diameter of ring 194, is located. The embodiment of FIGURE 17 accomplishes the same results as that of FIGURES 1 and 10; namely, an abrupt change from the fuel-air ratio of the primary combustion region 13 to that of the secondary combustion region 15, prevention of back-flow of secondary air into the primary combustion region 13 and intimate mixing in the secondary combustion region 15, but has the advantage that sharp corners, where carbon can collect, are eliminated.

FIGURE 18 shows still another means of introducing secondary air to provide an abrupt change from the fuel-air ratio of the primary combustion region 13 to that of the secondary combustion region 15, prevent back-flow of secondary air into the primary combustion region 15 and provide intimate mixing

29  
in the secondary combustion region 15. This is accomplished by providing secondary air inlet ports 202 which introduce secondary air radially and at an angle toward the downstream end of the combustor. An orifice means 50 similar to that of FIGURE 1 is provided.

FIGURE 19 illustrates yet another means of introducing secondary air, in which radially-disposed secondary air inlet tubes 204 extend into the flame tube to form a reduced diameter central dilution region. Orifice means 50 is located as in previously discussed FIGURE 1.

FIGURES 20, 21 and 22 show in greater detail the fuel nozzle 24.

In accordance with FIGURES 20, 21 and 22, fuel is introduced through passage 206 and thence into mixing chamber 208. Air is introduced through passage 210 and thence tangentially into mixing chamber 208 through air inlet means 212. The fuel and air are intimately mixed in chamber 208 and then impinge or blast against impingement or blast plate 214. This particular type of fuel nozzle is particularly advantageous for use with heavy fuels, such as

shale oil, since impingement of the fuel-air mixture against plate 214 serves to break up droplets of the fuel. The particular nozzle illustrated is available from Delavan Manufacturing Co., West Des Moines, Iowa as an "Air blast nozzle" and is available for different flow capacities and fuel-air ratios. The plate 214 is also adjustable so as to alter the exit angle from the nozzle. As used in the present invention, the exit angle of the fuel-air mixture was selected to be about 90°.

In one method of operating the combustor of FIGURES 1-8, a stream of fuel is introduced into the upstream end portion of mixing section 11 via fuel nozzle 24. Said fuel nozzle can be any suitable type of nozzle, e.g., a spray nozzle, an air assist nozzle, etc. The type of nozzle employed will depend to some extent at least, upon the type and properties of the fuel being used. A first stream of air is introduced into said mixing region 11 in a swirling or circular-like direction around said stream of fuel. Preferably, said first stream of air is introduced in a circumferential direction, around said fuel, and tangential the inner wall of said mixing section, as by means of tangential slots 28.

Preferably, a second stream of air is introduced into said mixing region 11 in a swirling or circular-like direction around said stream of fuel. Said second stream of air will also preferably be introduced in a circumferential direction, around said fuel, and tangential the inner wall of said mixing section, as by means of tangential slots 30. Said first stream of air, and said second stream of air when used, can have the same direction of swirl or a different direction of swirl, e.g, both clockwise, both counterclockwise, or one clockwise and one counterclockwise. Preferably, the directions of swirl will be opposite as illustrated in FIGURES 3 and 4. When employing the slots 28 illustrated in FIGURE 3 the direction of swirl will be clockwise, looking downstream. When employing the slots 30 illustrated in FIGURE 4 the direction of swirl will be



counterclockwise, looking downstream. It is also preferred that the volume of said second stream of air be greater than the volume of said first stream. This is preferred so as to more effectively counteract or neutralize the swirl of the first stream of air, as well as the axial component of the stream flowing axially in the mixing section, and thus provide a more homogenous mixture of fuel and air flowing from said mixing region into the primary combustion region. Said first and second streams of air thus comprise primary combustion air.

Said first stream of air, and/or said second stream of air when used, and said stream of fuel form a fuel-rich combustible mixture which is passed from said mixing region 11 into the primary combustion region 13. In said primary combustion region only partial combustion of said fuel-rich combustible mixture is caused to occur and a mixture comprising hot combustion products and partially combusted fuel is formed. Said combustible mixture is maintained in said primary combustion region for a period of time which is sufficient to provide a total residence time in said mixing region and said primary combustion region which is sufficient to reduce the conversion of fuel-nitrogen to  $\text{NO}_x$  emissions.

Said combustion products and partially combusted fuel mixture is then passed into the secondary combustion region which is located downstream from and in communication with said primary combustion region. A stream of secondary air is introduced into said secondary combustion region via ports 34, located at the upstream end thereof. In said secondary combustion region the combustion is completed under fuel-lean conditions with the resultant burnout of CO to  $\text{CO}_2$ .

Thus, the combustors discussed can be operated in a manner and the methods of the invention comprise an operation, wherein no axial air is introduced into the upstream end portion of mixing region 11. As shown by the examples given hereinafter such a method of operation gives good results and is one preferred method of operation, depending upon circumstances.

However, generally speaking, there will be more situations where it will be preferred to introduce a third stream of air into the upstream end of,

and generally axially with respect to, said mixing region, e.g., through dome member 17. Thus, generally speaking, such a method of operation can be said to be a more preferred method, again depending upon the circumstances or situation. When said third stream of air is used, either one or both of said first and second streams of air can be used in combination therewith. When only one of said first and second streams of air is used, the above-referred to third stream of air will become the second stream of air introduced into said mixing region.

When introducing said third stream of air through a variable dome member such as dome member 17, or through a nonvariable dome member, one presently preferred method of introducing said stream of fuel is to introduce same in the form of a hollow cone which diverges from its point of origin. Said third stream of air is then introduced around the stream of fuel, intercepts said cone, and mixes with said fuel.

Another preferred method of operation, particularly when using a heavy fuel, comprises introducing said third stream of air through a variable dome member such as that illustrated in FIGURES 11, 12 and 13. In such instances said third stream of air is introduced around said fuel in a generally radial direction which is generally perpendicular to the direction of introduction of said fuel.

Referring now to FIGURES 23 and 24 there is illustrated another combustor (designated herein as Combustor X) in accordance with the invention. Said Combustor X comprises an upstream fuel-air mixing section 11', an intermediate primary combustion section 13' located downstream from and in communication with said mixing section, and a secondary combustion section 15' located downstream from and in communication with said primary combustion section. Said mixing section 11' is defined at its downstream end by an orifice plate or means having an orifice 29' formed therein. A first air inlet means comprising tangential entry ports 28' is provided for introducing a first stream of air into said mixing section 11', similarly as in the combustor illustrated in

FIGURES 1 and 2. Although only one tangential air entry means is illustrated in FIGURE 24, it is within the scope of the invention to provide said Combustor X with a second tangential air entry means, similarly as in the combustor illustrated in FIGURES 1 and 2.

5 As illustrated in FIGURE 24, said Combustor X is provided with means for introducing a second stream of air into mixing section 11' in admixture with the stream of fuel. Said second stream of air is introduced into said stream of fuel via radial port(s) 52 supplied from the same plenum chamber supplying tangential entry ports 28'. As illustrated in FIGURE 24, said Combustor X is  
10 not provided with a variable dome member. However, it is within the scope of the invention to do so. It is also within the scope of the invention for fuel nozzle 24' in FIGURE 24 to be any suitable type of fuel nozzle, including an air-assist fuel nozzle. Preferably, said secondary air inlet means can comprise a plurality of fingers extending into the flame tube.

15 The operation of said Combustor X is substantially like that described above and elsewhere herein for the operation of the combustors represented by FIGURES 1 and 2. Said operation will be clear to those skilled in the art in view of the disclosure herein.

In the above-described methods of operating the combustors, or combustion zones, of the invention, which introduce air tangentially, in accordance  
20 with one embodiment of the invention, the amount of swirl air introduced into the mixing region 11 or 11' will be within the range of from about 35 to 100 percent of the total air introduced into said mixing region and then into the primary combustion region. Stated another way, the amount of air introduced  
25 through the dome member and/or with the fuel and into said mixing region can be an amount of up to 65 percent of said total air.

As shown by the examples given hereinafter, which utilize a light fuel with high nitrogen content, it has been found that the total amount of air introduced into said mixing region and then into the primary combustion region

should be an amount (relative to the amount of fuel) which is only sufficient to form a fuel-rich combustible mixture of fuel and air having a fuel-air equivalence ratio within the range of from 1.05 to about 1.7, if one is to obtain a significant reduction in the amount of fuel-nitrogen which is converted to NO<sub>x</sub> emissions, e.g., to 10 percent or less. When it is desired to obtain a further reduction in said conversion, e.g., to 7.5 percent or less, said fuel-air equivalence ratio should be maintained in the range of from about 1.14 to about 1.56.

As shown by said examples given hereinafter, it has been surprisingly discovered that there is a definite relationship between said fuel-air equivalence ratio ( $\phi$ ) and the total residence time in said mixing region and said primary combustion region. It has been found that, for relatively light fuels, said residence time should be maintained within the range of from about 30 to about 120 milliseconds, preferably 45 to 75 milliseconds in many instances, and should be sufficient, when correlated with said fuel-air equivalence ratio, to significantly reduce the conversion of fuel-nitrogen to NO<sub>x</sub> emissions, e.g., to less than 10 percent. The correlation or relationship between said fuel-air equivalence ratio and said residence time is discussed further hereinafter in connection with the examples.

It has also been surprisingly discovered that for the best results, e.g., the minimum conversion of fuel-nitrogen to NO<sub>x</sub> emissions, when tangentially introduced air is used, the amount of swirl air used should also be correlated with said fuel-air equivalence ratio and said residence time. This correlation is discussed further hereinafter in connection with the examples.

FIGURES 25, 26, 27 and 28 schematically illustrate burners, in accordance with the present invention, which are specifically designed for use in furnaces, boilers, process heaters and the like. These burners all have in common the fact that the heated zone of the furnace, heater or boiler becomes the downstream portion of the secondary combustion zone and, accordingly, the portion of the secondary combustion zone forming an integral part with the

primary combustion zone is extremely short, generally just of sufficient length to permit the introduction of secondary air in accordance with the present invention.

The combustor of FIGURE 25, referred to herein as combustor J, accomplishes intimate mixing of the fuel and a first volume of air or primary air by means of a configuration of the type shown in FIGURE 10 hereinabove described and the primary combustion zone is abruptly terminated and the secondary combustion zone initiated by a step-type, abrupt expansion as set forth in U.S. Patent 4,205,524. Specifically, the main body section 214 of the burner is comprised of ceramic sections and is attached to the main heating section 216 of a furnace generally constructed of fire brick. The burner 214 is surrounded by air plenum 218 which supplies most of the primary air to the burner as well as all of the secondary air. Fuel is supplied at a central location and in an axial direction by an air blast atomizer, utilizing a part of the first volume of air or primary air. The fuel nozzle extends into the air plenum and is surrounded by a wall 222 defining an air introduction or mixing zone. A second portion of the primary air is introduced through air slots 224, the size of which are adjustable by means of damper 226. While the second portion of primary air entering through slots 224 enters in a radial direction, this direction is changed to axial as the air exits the zone 222 through reduced diameter throat 228. Thus, the fuel and first portion of the primary air and the second portion of the primary air form the axial flow component referred to herein. A third portion of the primary air is introduced through tangential, counterclockwise primary air slots 230. This air thus introduced in a generally radial direction then mixes with the fuel and air from the air blast nozzle and slots 224 in zone 232. Finally a fourth portion of the primary air is introduced through tangential, clockwise air slots 234. Accordingly, the air entering through slots 230 and 234 form the generally radial flow component referred to herein. While section 232 is referred to as a mixing zone, in actual practice when the burner is operating at peak performance the flame will be seated adjacent

throat 228 of the burner. Fuel rich combustion thus takes place in sections 232 and 214 of the burner, resulting in the production of an effluent containing unburned fuel CO and carbon dioxide and most of the nitrogen in the form of elemental nitrogen. The flame front leaving section 214 of the burner is then abruptly expanded into section 238. Section 238 is provided with air slots 240 through which the secondary air or the remainder of the air is introduced so as to produce a fuel lean mixture. A damper 242 is also provided to adjust the volume of this air. The air through slots 240 enters radially and is of sufficient velocity to essentially penetrate the flame front to its axis and mix with the moving front and effluent from the primary combustion region. The addition of air through the secondary air slots 240 thus produces an overall fuel lean fuel air mixture. This mixture is then again expanded into the heating zone 216 of the furnace. This expansion completes the mixing of the primary zone effluent or flame front and the secondary air. By way of example, section 232 has a length to diameter ratio of about 1. A fuel-air mixing zone forming a part of zone 236 also has a length diameter ratio of 1 and, what may be termed the NO<sub>x</sub> reduction portion of zone 236, is about 4 feet long at a flow velocity of about 50 feet per second. In the operation of this burner the air introduced would be approximately 2.8 percent to the air blast nozzle, 26.6 percent through slots 224, 14.1 percent through slots 230 and 21 percent through slots 234, thus introducing a first volume of air or primary air equal to 64.5 percent of the total air to combustion system. The remaining 35.5 percent of the total air is introduced through secondary air slots 240 to produce an overall fuel/air ratio having 3 percent excess oxygen or a fuel/air equivalence ratio of 0.87. Thus, under these designed conditions, the fuel/air equivalence ratio in the primary combustion region or fuel-rich region would be 1.35. As in previous configurations the flame is seated adjacent the constriction 228 (the end of nozzle 220 and the primary combustion as fuel-rich combustion extends from this point to the secondary air introduction slots 240.

FIGURE 26 shows yet another embodiment of the present invention specifically designed for use with a process heater, boiler or the like. The burner

of FIGURE 26 utilizes a fuel-primary air mixing configuration similar to that of FIGURE 15 heretofore described and an abrupt termination means for the introduction of secondary air and mixing of the secondary air with the flame front of the character set forth in U.S. Patent 4,205,524.

5                   Specifically, the burner 242 is connected to heated section 244 of a furnace, boiler or the like. As in the previous embodiment, the heated region 242 comprises the downstream portion of the fuel lean or secondary region of the combustion system. Burner 242 is surrounded by air plenum 246 which supplies most of the primary air and all of the secondary air to the burner system. Fuel  
10 is supplied by air blast atomizer 248 which utilizes a first part of the first volume of air or primary air. Atomizer 248 extends into mixing section 250 which is supplied with air through air introduction slots 252. A second portion of the first volume of air or the primary air is introduced through slots 252 and this volume is controlled by a damper 254. As in the previous embodiment, the air  
15 enters through slots 252 in a generally radial direction. However, in mixing section 250 and passage through restricted neck portion 256 the second portion of primary air becomes an axial flow component along with the fuel and air from air blast atomizer 248. A third volume of primary air is introduced through air slots 258 which are controlled by damper 260. The air introduced through  
20 slots 258 is introduced in a true radial direction and becomes the sole radial flow component. Simplified mixing occurs in this particular arrangement, both due to the sudden expansions of the air-fuel mixture from orifice 256 and by the air penetrating toward the core of the mixture in the axial direction from slots 258. The air-fuel mixture is ignited and the flame seated adjacent the  
25 orifice 256 and the flame front passes through the primary combustion region 262 as a fuel-rich mixture. Effluent from section 262 comprises flue gases containing unburned and partially burned fuels. This effluent is then expanded abruptly into section 264 where it is diluted to a fuel lean overall mixture by the introduction of sufficient secondary air through slots 266. The size of slots 266 is  
30 controlled by damper 268 and the mixture is then further expanded into the

heating section 244 of the furnace or the like. As in the previous embodiment, abrupt termination of the primary combustion zone occurs as a result of the expansion of the flame front in section 264 the radial or axial introduction of the secondary air through slots 266 and the further expansion into the furnace proper 244. In this particular burner a typical arrangement would have a fuel-air mixing section downstream from slots 258 and having a length/diameter ratio of about 1 in section 262 of the burner followed by a  $\text{NO}_x$  reduction zone about 4 feet long at a flow velocity of 50 feet per second. To achieve a fuel/air equivalence ratio in the primary combustion section or zone of about 1.55, air distribution would be approximately 2.4 percent to air blast atomizer 248, 23.3 percent through slots 252, 30.6 percent through slots 258 and the remainder of 43.7 percent through slots 266, thus providing an overall fuel air equivalence ratio of 0.87 at 3 percent excess oxygen.

Yet another embodiment of the present invention, specifically designed for use as a burner with a combustion system in which the heated zone of a furnace, boiler or other heater comprises a secondary combustion zone, is shown in FIGURE 27. The embodiment of FIGURE 27 includes a fuel and primary air mixing configuration of the character shown in detail in FIGURE 16 and a means for abruptly terminating the primary combustion zone of the character previously described in connection with FIGURES 1, 9, 17, 18 and 19. As previously shown in FIGURES 25 and 26, a rather long primary combustion zone was illustrated. However, in many instances where the burner is to be utilized in a combustion system for a heater, boiler or the like, space limitations are a determining factor and thus a long burner is not acceptable. Consequently, the burner of FIGURE 27 is designed to fit into a shorter space than the two previous burners and for this purpose has a substantially larger diameter than the burners of FIGURES 25 and 26 so as to provide essentially the same residence time in the primary combustion zone. Since the primary combustion region, and thus the flame front passing through the primary combustion zone, is larger in cross section in this particular configuration it is helpful to have the improved abrupt



39  
termination means for the termination of the primary zone and the initiation  
of the secondary zone, wherein improved mixing is attained by reducing the  
peripheral dimensions of the flame front and introducing the secondary air  
either immediately prior to the reduction of the dimensions of the flame front  
5 or into the reduced dimension section of the flame front and then abruptly  
expanding, in this case into the heated zone of the furnace or the like. The  
burner 270 of FIGURE 27 was designed as an actual prototype burner for burning  
shale oil with a heat output of 10 million BTU. The burner 270 is provided  
with fuel nozzle 272 having a gas inlet 274, for introducing gas to ignite the  
10 burner, and a fuel inlet 278, in this case a shale oil inlet, and a steam inlet  
280. It is well known that the sensible heat of the steam makes steam superior  
atomizing medium for extremely heavy material such as shale oil, as opposed to  
air as an atomizing means. However, an air blast atomizer or the like as shown  
in the previous embodiments could just as readily be utilized. All of the pri-  
15 mary air is introduced to the burner 270 through a primary air plenum 282. All  
of this primary air passes through spin vanes 284 which provide a swirling  
motion to the air introduced a generally radial direction. Consequently, in the  
configuration illustrated herein, fuel is the sole axial flow component and the  
air through spin vanes 284 is the sole radial flow component. These two flow  
20 components mix in the central region of the air plenum 282 and are passed through  
necked down orifice portion 286 and abruptly expanded into the primary combustion  
zone 288. The volume of air passing through the spin vanes 284 could be adjust-  
able by means of a damper, as specifically illustrated in FIGURE 16, but none is  
used in the present application, control of the volume of air being made upstream  
25 of its introduction into the plenum 282 by appropriate damper means. The flame  
front passing through primary combustion region 288 is then gradually reduced  
in diameter to the diameter of orifice 290. Thus, the abrupt primary zone  
termination means shown specifically is that illustrated in FIGURE 17 as opposed  
to that of FIGURES 1 and 9. The reason for this is that this particular con-  
30 figuration is better for a heavy oil or the like which has a tendency to produce

carbon deposits. By gradually reducing the diameter of the flame front the pockets which would be formed by abrupt reduction to the orifice size are eliminated, thus reducing the probabilities of carbon deposit and the like. In this particular instance, secondary air is introduced through a secondary air plenum 292, separate from the primary air plenum. However, as is obvious to anyone skilled in the art and as was done in the previous embodiments, a single air plenum may supply both primary or secondary air or they may be separate as in the present case and controlled by external valves or one or both may be controlled by dampers such as the damper 294 on the secondary air slots 296 of the present embodiment. Secondary air is introduced radially into the reduced diameter portion of the flame front, thus increasing penetration of the secondary air to the central axis of the flame front and enhancing mixing and rapid conversion from a fuel rich configuration to a fuel lean configuration. The mixture is then further mixed by abrupt expansion into the heated region of the furnace proper 298. This particular configuration had a 6-inch diameter throat 286, a primary zone diameter of 28 inches, and a diameter of 20.5 inches for orifice 290. The length of the primary combustion region from the upstream edge of the secondary air inlet 296 to the throat 286 was 44 inches and the portion of the secondary combustion region from the upstream edge of secondary air slots 296 to the heated zone 298 of the furnace proper was 7 inches. As previously indicated, damper 294 adjusts the size of secondary air openings 296 which, in the specific case shown, comprise twelve 2-inch diameter holes.

Yet another embodiment, designed as a prototype of a process burner having a 10 million BTU output, is illustrated in FIGURE 28. The configuration of the primary air and fuel mixing mechanism is of the general character shown in detail in FIGURE 15 of the drawings, while the means for abruptly terminating the primary combustion region and initiating the secondary combustion region or going from the fuel rich combustion region to the fuel lean combustion region is specifically the type shown in the previous example and described with respect to FIGURE 17 of the drawings. Specifically, the burner 300 is provided with a

fuel nozzle 302 having an ignition gas inlet 304, a fuel oil inlet 306 and a steam inlet 308. This nozzle extends into a primary air plenum 310. In this particular embodiment a first portion of primary air is introduced through air slots 312, which are controlled in size by damper 314, and a second portion of the primary air is introduced through air slots 316, controlled by damper 318. As is obvious from the illustration of FIGURE 28 the air introduced through air slots 312 is introduced in an initial radial direction but becomes a portion of the axial flow component by passing through reduced diameter throat 320 surrounding the downstream end of nozzle 302. The air introduced through air slots 316 then intercepts and mixes with the axial flow component by passage radially through slots 316. The air-fuel mixture then expands into the primary combustion region 322, initially gradually and then abruptly. The initial gradual configuration is simply designed to produce a smooth transition from the mixing region to the burner proper and is generally dictated by the thickness of the walls of the burner, the design being, of course, adapted to prevent formation of carbon, etc. in this transition zone. However, as in the previous embodiments, the expansion into the large diameter primary combustion zone also aids in the mixing of the fuel and air. The flame front which is generally seated adjacent the entry to the primary combustion zone then passes through the primary combustion zone, is gradually reduced in diameter and passes through reduced diameter orifice 324. All of the secondary air is introduced into secondary air plenum 326 and thence through radial air inlets 328, which are controlled by damper 330. Finally, the flame front, mixed with sufficient secondary air to provide a lean overall fuel-air mixture, is abruptly expanded into the furnace proper 332. The embodiments of FIGURES 27 and 28, referred to as combustors L and M, respectively, were successfully tested for the burning of shale oil, as will be hereinafter illustrated by the examples.

FIGURE 29 illustrates a schematic arrangement of the burner of FIGURE 28 adapted for commercial use with relative dimensions referred to thereon. Specifically, burner 334 is provided with a fuel nozzle 336, supplied with fuel

through line 338 and steam through line 340. All of the primary and secondary air is supplied to common air plenum 342. In contrast to the embodiment of FIGURE 28, burner N of FIGURE 29 does not have dampers controlling the volume of primary and/or secondary air introduced into the burner. This would be the usual, most simplified version of the burner for use in industrial operations. Specifically, the relative volumes of primary and secondary air would be controlled by properly selecting the size of openings through which the various streams of air are introduced into the furnace, thus one would control only the volume of air to the air plenum so as to produce an overall fuel/air ratio of a predetermined amount, in most cases 3 percent excess air or a fuel/air equivalence ratio of 0.87. Primary air is introduced through air inlets 344 and 346, respectively. The air through inlets 344 becomes a part of the axial flow component by flow through orifice 348 around fuel nozzle 336. The primary air introduced through inlets 346 is the sole radial flow component in this particular configuration. The mixture of air and fuel is then expanded into the primary combustion region 350, as in the previous embodiment, and the flame front is then gradually reduced in diameter to the diameter of orifice 352. Secondary air is then introduced through air inlets 354 into the reduced diameter flame front and, finally, the mixture is expanded abruptly into the furnace proper or heated section of an industrial facility 356.

The following table illustrates the relative dimensions of burners of the character illustrated in FIGURE 29, designed for heat outputs from 10 million to 100 million BTU, respectively.

TABLE

Burner Size MM BTU/Hr.	A	B	C	D	E
	Fire Box O.D., in.	Wind Box Dia., in.	Fire Box Length, in.	Wind Box Length, in.	Fuel Gun Extension, in.
10	37	44	56	66	19
15	42	50	62	74	19
18	44	53	66	78	19
25	48	58	73	85	19
30	40	64	83	99	19
	60	72	95	113	22
	80	78	105	125	24
35	100	82	113	135	24

The burner schematically shown in FIGURE 29 is shown in a horizontal position as opposed to the vertical position shown in FIGURES 25 through 28. Thus, it is obvious that the burners may be mounted on a furnace or the like in either direction without, in any way, affecting the operation thereof. In addition, the burner of FIGURE 29 is a forced draft configuration with an integral windbox and is operable on either a gaseous or a liquid fuel. Thus, an extremely simple yet highly effective burner, for use on a wide variety of fuels, is provided herein which has the capability of reducing  $\text{NO}_x$  pollutants below EPA standards irrespective of the type of fuel or the amounts of fuel bound nitrogen.

Referring to FIGURE 30, there is illustrated a combustor of the invention employed in combination with a furnace such as the furnace of a boiler. Any of the combustors of the invention can be so employed. Said combustor can be employed with any type furnace comprising a shell 54 which encloses a heated region 56. In the practice of this combination of the invention, the flame tube of at least one said combustor would be mounted on said shell 54 so as to discharge into said heated region 56. Said combustors can be mounted on said shell in any suitable manner. Preferably, said combustor will be mounted with a shortened secondary combustion section on the outside of said shell in windbox or plenum chamber 58, as indicated in FIGURE 30, for supply of the primary air inlets 28 and/or 30, and secondary air inlet 34. If desired, said combustor can be provided with a variable dome as described above for the admission of a variable, peripheral stream of primary air into the mixing region. If desired said variable dome can also be supplied with air from windbox 58, or can be supplied from a separate source of air.

By so mounting the combustor with a shortened secondary combustion section outside shell 54, and with the combustor discharging combustion products 60 into the heated region 56, said heated region becomes the downstream portion of the secondary combustion region. There is thus provided a unitary combustion zone in which the above-described methods of the invention can be carried out under the operating conditions described herein.

FIGURES 31 and 32 of the drawings illustrate schematically an arrangement for utilizing a plurality of burners in accordance with the present invention in an industrial process heater. Specifically, the illustrated arrangement could be utilized as a crude oil heater for a crude oil cracking unit in a refinery. In accordance with FIGURES 31 and 32, the process heater or furnace is designated generally as 358. The furnace 358 comprises a primary heated section 360 which discharges flue gases through stack 362. Discharging into the heated section of the furnace 360 are a plurality of burners of the character illustrated in FIGURES 25 through 29, respectively, in this case comprising 12 such burners 364. As in the previous embodiments described, the burners are each provided with primary and secondary air and fuel and discharge into the main heating section of the furnace 360 which becomes the downstream portion of the secondary combustion region of the burners. Both primary and secondary air is provided to the burners through common air plenums 366 and 368, respectively. Air is supplied to plenums 366 and 368 by appropriate blowers 370 and 372, respectively. Air to blowers 370 and 372 is drawn in from the atmosphere by appropriate means controlled by dampers 374 and 376, respectively. The dampers are in turn controlled by continuously or intermittently sampling the exhaust gas through stack 362 by means of sampling means 378, passing the sample through oxygen analyzer 380 and utilizing the signal from analyzer 380 for the control of dampers 374 and 376 through recorder-controllers 382 and 384. In the particular instance shown, the crude oil to the crude oil cracking unit passes through heat exchange tubes 386 mounted in heated section 360 of the furnace or heater. In a typical operation of such a unit, the size of the primary and secondary air inlets can be predesigned to obtain the desired split between the volume of primary and secondary air as in FIGURE 29, or the primary and secondary air inlets can be provided with appropriate dampers. In the latter case the dampers in the primary and secondary air inlets would be adjusted to obtain the desired split of the total air flow to the burners. The fuel flow would be established for the level of heat release desired from the burner. Then the

total air flow would be adjusted to reach a level  $\lambda$ . For example, 3 percent excess air in the exhaust gas. While 3 percent is typical, more or less can be tolerated with the lower limit established by exhaust smoke or excessive carbon monoxide emissions and the higher levels carrying with them a heat loss to the excess air flowing through the furnace.

Another type of installation would be to place an individual air plenum around all of the burners, doing away with the separate air lines to the primary and secondary air inlets or the two individual plenums. With more than one burner, the individual plenums would be manifolded to the central air blower, or blowers, with dampers to control the air split between burners and assure a balance in the quantity of air flow to each burner. While air flow to the individual burners could be judged by utilizing the pressure drop across the dampers and pressure in the plenum chambers, primary control of the furnace would still rest upon analysis of the exhaust gas to maintain a desired level of excess oxygen, usually 3 percent as indicated. The design illustrated in FIGURES 30 and 31 shows two major banks of burners firing across the heat exchange tubes. Separate blowers would supply air to each plenum as indicated to balance air flows. However, this could be done using one larger blower with the air flow split by dampers. While not shown, heat exchangers may be utilized to preheat the inlet air using the exhaust gas from the heater stack. Primary and secondary air flows will be split by selective sizing of the openings to the burners and fuel flow would be adjusted using fuel pressure to balance flows in the different burners. As before, the overall control of the stoichiometry in the furnace will be ultimately by maintaining the oxygen level in the exhaust gas at a fixed level, for example, about 3 percent.

FIGURE 33 of the drawings shows, schematically, a test furnace arrangement utilized to test the 10 million BTU burners illustrated in FIGURES 27 and 28. In accordance with FIGURE 33, the burner 386 fires vertically into the furnace proper 388. Thus, the furnace 388 becomes the downstream portion of the secondary combustion region. The furnace has a 6-inch internal

liner of fire brick with an outer steel shell. The burner was provided with a window 390 designed to observe the character of the flame in the furnace. The furnace had an internal diameter of about 6 feet and a total length of about 16 feet. The furnace discharged into an exhaust stack 392, provided with an appropriate damper 394. Probes for gas samples, temperature and pressure measurements extended into the stack at point 396.

FIGURES 34 and 35 show, schematically, air supply systems for the furnace of FIGURE 33. All air was supplied from an air blower and preheater 398 through conduit 400. Conduit 400 was provided with a 16-inch Venturi 402 and terminated in an air header 404. From air header 404 the air was split between a primary air conduit 406 and a secondary air conduit 408. Primary and secondary air conduits 406 and 408 were provided with appropriate control dampers 410 and 412, respectively. In addition, primary air supply line 406 was provided with an 8-inch diameter Venturi 414.

The schematic of FIGURE 36 shows the installation of the test burners of FIGURES 27 and 28 to fire horizontally into another test furnace arrangement. In this particular instance, the burner 416 was designed to fire horizontally into the furnace so as to provide better access to the controls of the burner, etc. The furnace also differed from that of FIGURE 33 to the extent that the furnace of FIGURE 33 was what could be termed a "hot wall" furnace in that the furnace shell tended to maintain the temperatures within the furnace and to a great extent have exhaust gases exiting the furnace at higher than desired temperatures, thus reconverting some of the thermal nitrogen into thermal  $\text{NO}_x$ . By contrast,, furnace 418 is referred to as a "cold wall" furnace, to the extent that it is provided a steel shell with a water wall for cooling. Furnace 418 discharges exhaust gases through stack 420. The furnace was provided with an observation window 322. The wall thickness of the furnace was approximately 9 inches and the diameter of the furnace about 12 feet. Stack 420 was 4 feet in diameter and was provided with an appropriate damper 424 and samples of the gas and the temperature and pressure of the exhaust gas were taken at point 426 in the stack.



FIGURE 37 illustrates schematically the air supply system for the furnace of FIGURE 36. In FIGURE 37, air was supplied for both primary and secondary air to the burner by air blower and preheater 428. This air was discharged through a main air conduit 430, provided with a 16-inch Venturi 432 and terminated in an air header 434. The air from air header 434 was then split into a primary air conduit 436 and a secondary air conduit 438. The primary air conduit is provided with a primary air damper 440 and secondary air conduit was provided with an air damper 442. The primary air conduit 436 is also provided with an 8-inch Venturi 444.

In any of the above-described methods of operation, the relative volumes of the various streams of air can be controlled by varying the sizes of the inlet openings therefor, relative to each other. The above-described variable domes can be employed for controlling the axially or radially introduced air. Flow meters or calibrated orifices can be employed in the conduits supplying the other streams, if desired.

As to general operating conditions, it is within the scope of the invention to operate the combustors or combustion zones employed in the practice of the invention under any conditions which will give the improved results of the invention. For example, it is within the scope of the invention to operate said combustors or combustion zones at suitable inlet air temperatures of up to about 1500°F, or higher; at pressures within the range of from about 1 to about 40 atmospheres, or higher; at flow velocities within the range of from about 1

to about 500 feet per second, or higher; and at heat input rates within the range of from about 30 to about 1200 Btu per pound of air. Generally speaking, the upper limit of the temperature of the air streams will be determined by the means employed to heat same, e.g., the capacity of the regenerator or other heating means, and materials of construction in the combustor or combustion zone, and/or the turbine utilizing the hot gases from the combustor. For a furnace type installation, preheating should be limited to 600° to 800°F. Generally speaking, operating conditions in the combustors of the invention will depend upon where the combustor is employed. For example, when the combustor is employed with a high pressure turbine, higher pressures and higher inlet air temperatures will be employed in the combustor. Thus, the invention is not limited to any particular operating conditions. As a further guide to those skilled in the art, but not to be considered as limiting on the invention, presently preferred operating ranges for other variables or parameters are: heat input, from 30 to 500 Btu/lb. of total air to the combustor; combustor pressure, from 3 to 10 atmospheres; and reference air velocity, from 50 to 250 feet per second.

The following examples will serve to further illustrate the invention.

EXAMPLE I

A test program was carried out to investigate the effect of the equivalence ratio  $\phi$  in the primary combustion region on the conversion of fuel-nitrogen to  $\text{NO}_x$  emissions. Said test program was carried out employing Combustor A, a combustor in accordance with the invention. Said Combustor A had a configuration essentially like that illustrated in FIGURE 1, and its design characteristics are set forth in the schematic representation thereof set forth in FIGURE 9. Said test program comprised operating the combustor over a program comprising the 10 different sets of test or operating conditions set forth in Table I below.

As set forth above, said Combustor A was provided with a variable dome whereby the amount of air (axial air) introduced through said dome into

the mixing region and then into the primary combustion region could be varied. As shown in a subsequent example herein, the amount of the swirl air (introduced via the tangential openings), relative to the amount of said axial air, introduced into the mixing region and then into the primary combustion region, has an effect on the conversion of fuel-nitrogen to  $NO_x$ . Thus, in each of said 10 sets of operating conditions set forth in Table I, the amount of swirl air is the optimum amount for obtaining minimum conversion of fuel-nitrogen to  $NO_x$ . Said optimum amount was determined by carrying out a series of test runs, at each of said 10 test conditions, wherein the amount of axial air was varied by rotating the dome in five degree increments over its entire range (0 to 45 degrees), i.e., from fully closed to fully open.

In said runs a first stream of tangential air was introduced through tangential slots 28, a second stream of tangential air was introduced through tangential slots 30, a third stream of air (axial-when admitted) was introduced through the openings in the dome 17, and a stream of fuel was introduced via fuel nozzle 24, with all of said streams being introduced into the mixing region 11 located in the upstream end portion of the combustor. A stream of secondary air was introduced via ports 34, and a stream of dilution or quench air was introduced via ports 40. The total amounts of said streams of air and fuel are set forth in said Table I. Flow velocity in the primary combustion region was held constant at 50 feet per second so as to maintain the residence time therein constant at 60 milliseconds. The equivalence ratio  $\phi$  in the secondary combustion region was held constant at 0.87 so as to achieve three percent excess oxygen (dry basis) in the exhaust gases. Test conditions 9 and 10 were included in the program so as to extend the range of the primary combustion region equivalence ratio  $\phi$  to 0.87 in order to simulate the operation of a single-stage combustor.

The test program included two groups of runs I and II carried out under the same conditions as described above. In the Group I runs a base

fuel, Philjet A-50 (a kerosine), was used. Said base fuel contained zero percent fuel-nitrogen. In the Group II runs the fuel used was a synthetic fuel consisting of said base fuel to which there had been added sufficient pyridine to result in said base fuel containing 2.000 weight percent of fuel-nitrogen.

During each of said runs the exhaust gas from the combustor was analyzed under specifically controlled conditions to determine the concentration of  $\text{NO}_x$ , CO, and unburned hydrocarbon (HC). Since the base fuel contained no fuel-nitrogen, the  $\text{NO}_x$  values from the Group I runs were due to thermal  $\text{NO}_x$  and said Group I runs established a base for thermal  $\text{NO}_x$  produced at each test condition. Then, since the Group II runs were carried out at the same test conditions, the  $\text{NO}_x$  values therefrom established the total  $\text{NO}_x$  produced at each said test condition. The theoretical amount of  $\text{NO}_x$  which would have been produced from the fuel-nitrogen in the synthetic fuel used in said Group II runs, if all of said nitrogen had been converted to  $\text{NO}_x$ , was calculated to establish the theoretical  $\text{NO}_x$  (t  $\text{NO}_x$ ). Subtraction of the  $\text{NO}_x$  values in the Group I runs from the  $\text{NO}_x$  values in the corresponding Group II runs established the increment of  $\text{NO}_x$  (i  $\text{NO}_x$ ) produced from the fuel-nitrogen at each test condition. Then,  $i\text{NO}_x/\text{tNO}_x \times 100 =$  percent conversion of fuel-nitrogen to  $\text{NO}_x$  at each test condition.

FIGURES 38, 39, 40 and 41 of the drawings set forth the results of said runs. Referring to said FIGURE 38, the results there set forth show that when it is desired to maintain the conversion of fuel-nitrogen to  $\text{NO}_x$  emissions at 10 percent or less, the equivalence ratio in the primary combustion region should be maintained within the range of from about 1.04 to about 1.6; and within the range of from about 1.14 to about 1.51 when it is desired to maintain said conversion at less than 7.5 percent. It will be noted that the minimum conversion of about 6 percent was obtained over the range of 1.3 to 1.4 equivalence ratio, and that above 1.4 the percent conversion increased

rapidly. Referring to FIGURE 39, the curve for the Group II runs (2.0 weight percent nitrogen in the fuel) shows that over the equivalence ratio range of about 1.24 to about 1.44 the total  $\text{NO}_x$  produced was less than the California maximum of 225 parts per million by volume. FIGURE 40 shows the level of unburned hydrocarbons, or HC emissions, in the flue gas and FIGURE 41 the level of CO emissions in the flue gas.

TABLE I

## OPERATING CONDITIONS FOR COMBUSTION OF HIGH-NITROGEN FUELS

Test Condition	Comb. Oper. Pressure in. Hg. Abs.	Inlet Air Temp. °F.	PRIMARY REGION				SECONDARY REGION				DILUTION REGION				
			Fuel lb/hr	Fuel $\phi$	Total Air lb/sec	Temp. °F.	Flow Velocity ft/sec	$\phi$	Total Air lb/sec	Temp. °F.	Flow Velocity ft/sec	$\phi$	Total Air lb/sec	Temp. °F.	Flow Velocity ft/sec
1	32	650	52.4	1.80	0.120	3100	50	0.87	0.126	3690	120	0.30	0.472	1950	203
2	32	650	47.7	1.70	0.115	3230	50	0.87	0.109	3690	109	0.30	0.430	1950	185
3	32	650	43.5	1.60	0.112	3350	50	0.87	0.093	3690	100	0.30	0.392	1950	169
4	32	650	39.3	1.50	0.108	3500	50	0.87	0.077	3690	90	0.30	0.354	1950	152
5	32	650	36.3	1.40	0.107	3640	50	0.87	0.064	3690	83	0.30	0.327	1950	141
6	32	650	32.0	1.30	0.101	3750	50	0.87	0.049	3690	73	0.30	0.288	1950	124
7	32	650	28.9	1.20	0.099	3850	50	0.87	0.037	3690	66	0.30	0.260	1950	112
8	32	650	26.2	1.10	0.098	3900	50	0.87	0.025	3690	60	0.30	0.235	1950	101
9	32	650	23.8	1.00	0.098	3900	50	0.70	0.042	3300	59	0.30	0.186	1950	92
10	32	650	21.9	0.87	0.103	3690	50	0.70	0.026	3300	54	0.30	0.171	1950	85

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EXAMPLE II

to FI:

Another test program was carried out to further investigate the effect of the equivalence ratio  $\phi$  in the primary combustion region on the conversion of fuel-nitrogen to  $\text{NO}_x$  emissions. This test program was carried out in essentially the same manner as that described above in Example I except that Combustor B, another combustor in accordance with the invention was employed. Said Combustor B was like Combustor A except that the length of spool 14 (see FIGURE 9) was increased from 18 inches to 24 inches to provide a primary combustion region having a length of 42 inches instead of 36 inches; and the length of spool 18 (see FIGURE 9) was increased from 12 inches to 30 inches to provide a secondary combustion region having a length of 36 inches instead of 18 inches. Said increase in the length of the primary combustion region increased the residence time therein from 60 to 70 milliseconds. Said increase in the length of the secondary combustion section increased the residence time therein from 18 to 36 milliseconds. All other operating conditions were the same as in the runs of Example I. See Table I.

FIGURES 42, 43, 44, 45, 46 and 47 of the drawings set forth the results obtained. FIGURES 14 and 15 are representative of the results obtained when the combustor was operated at a given test condition and the amount of axial air was varied to determine the optimum amount of swirl air to use for obtaining minimum conversion of fuel-nitrogen to  $\text{NO}_x$  emissions at said test condition (Test Condition No. 5 in Table I). The results set forth in FIGURE 44 show that when it is desired to maintain the conversion of fuel-nitrogen to  $\text{NO}_x$  emissions at 10 percent or less, the equivalence ratio in the primary combustion region should be maintained within the range of from about 1.08 to about 1.65; and within the range of from about 1.18 to about 1.56 when it is desired to maintain said conversion at less than 7.5 percent. It will be noted that the minimum conversion of about 5.7 percent was obtained over the range of 1.34 to 1.42 equivalence ratio, and that above 1.42 the percent

conversion increased rapidly. Referring to FIGURE 45, the curve for the Group II runs (2.0 weight percent nitrogen in the fuel) shows that over the equivalence ratio range of about 1.28 to about 1.51 the total  $\text{NO}_x$  produced was less than the California maximum of 225 parts per million by volume. FIGURE 47 shows that the increased residence time in the secondary combustion region was very effective in reducing the CO emissions and, together with FIGURES 44 and 45, that this was done without any detrimental effect on the  $\text{NO}_x$  emissions.

#### EXAMPLE III

In another series of runs carried out in Combustor A under conditions similar to those used in Example I above (see Table I above), and at an equivalence ratio of 1.4 in the primary combustion region, it was found that the CO emissions could be reduced to about 160 ppmv by using sufficient excess air in the secondary combustion region to achieve an equivalence ratio therein of 0.87. This was at 3 percent excess oxygen, dry basis, in the exhaust gases. Further increases in excess air reduced the CO emissions to about 100 ppmv at an equivalence ratio of about 0.75. Again, this reduction in CO emissions was accomplished without detrimental effect on the  $\text{NO}_x$  emissions.

#### EXAMPLE IV

Another test program was carried out to further investigate the effect of the equivalence ratio  $\phi$  in the primary combustion region on the conversion of fuel-nitrogen to  $\text{NO}_x$  emissions, and to evaluate the performance of Combustor X when operated in a rich-lean combustion process in accordance with the invention. Said Combustor X had a configuration essentially like that illustrated in FIGURES 23 and 24. Said test program comprised operating the combustor over a program comprising the 13 different sets of test or operating conditions set forth in Table II below.

Said Combustor X, unlike Combustors A and B, was not provided with a variable dome. In carrying out said test runs a first stream of air was introduced via tangential openings 28' into the mixing region 11' located in



the upstream end portion of the combustor. Said first stream of air was introduced in a swirling direction around an admixture of fuel and a second stream of air which was introduced via nozzle 24' into the upstream end of said mixing region in an axial direction with respect thereto. A stream of secondary air was introduced into the flame tube via finger ports 34'. The total amounts of said streams of air and fuel are set forth in said Table II below. Flow velocity in the primary combustion region was held constant at 25 feet per second so as to maintain the residence time therein constant at 47 milliseconds. The equivalent ratio ( $\phi$ ) in the secondary combustion region was held constant at 0.5 so as to achieve three percent excess oxygen (dry basis) in the exhaust gases. In said Combustor X the amount of said first stream of air (swirl air) was fixed at 74 percent, and the amount of said second stream of air (axial air) was fixed at 26 percent.

TABLE II  
OPERATING CONDITIONS FOR COMBUSTION OF HIGH-NITROGEN FUELS

Test Condition	Comb. Oper. Pressure in. Hg. Abs.	Inlet Air Temp. °F.	PRIMARY REGION					SECONDARY REGION				
			Fuel lb/hr	φ	Total Air lb/sec	Temp. °F.	Flow Velocity ft/sec	φ	Total Air lb/sec	Temp. °F.	Flow Velocity ft/sec	
5	30	500	3.51	2.0	25.9	2750	25	0.5	77.8	2400	89	
	30	500	2.91	1.8	23.9	3020	25	0.5	62.2	2400	74	
10	30	500	2.42	1.6	22.4	3260	25	0.5	49.3	2400	61	
	30	500	1.97	1.4	20.8	3540	25	0.5	37.5	2400	50	
	30	500	1.78	1.3	20.2	3660	25	0.5	32.3	2400	45	
	30	500	1.60	1.2	19.7	3760	25	0.5	27.6	2400	41	
	30	500	1.44	1.1	19.4	3830	25	0.5	23.3	2400	37	
15	30	500	1.32	1.0	19.5	3810	25	0.5	19.5	2400	34	
	30	500	1.23	0.9	20.3	3650	25	0.5	16.2	2400	31	
	30	500	1.15	0.8	21.3	3450	25	0.5	12.8	2400	29	
	30	500	1.08	0.7	22.9	3180	25	0.5	9.2	2400	28	
	30	500	1.01	0.6	24.8	2900	25	0.5	5.0	2400	26	
20	30	500	0.94	0.5	27.7	2550	25	0.5	0.0	2400	24	

The test program included two groups of runs, Group I and Group II, carried out with the base fuel and synthetic fuel respectively, as described above in Example I, for determining the percent conversion of fuel-nitrogen to  $\text{NO}_x$  emissions.

5            FIGURES 48, 49, 50 and 51 of the drawings set forth the results of said runs. Said FIGURE 48 shows that when it is desired to maintain the conversion of fuel-nitrogen to  $\text{NO}_x$  emissions at 10 percent or less, the equivalence ratio  $\phi$  in the primary combustion region should be maintained in the range of from about 1.04 to about 1.12; and within the range of from about 1.1  
10 percent when it is desired to maintain said conversion at 7.5 percent or less.

            Based on the results obtained in the above Examples I, II, III and IV, the data shown in FIGURES 38, 43 and 48, and other results obtained in subsequent examples herein, it is concluded that in the combustion of fuels containing fuel-nitrogen, by two-stage combustion in a combustor having a  
15 primary combustion region and a secondary combustion region, the equivalence ratio in the primary combustion region should be maintained within the range of 1.05 to 1.7 when it is desired to maintain the conversion of fuel-nitrogen to  $\text{NO}_x$  emissions at 10 percent or less, and that said equivalence ratio should be maintained within the range of from 1.14 to 1.56 when it is desired to maintain  
20 said conversion at 7.5 percent or less.

#### EXAMPLE V

            Another test program was carried out to investigate the effect of residence time in the primary combustion region on the conversion of fuel-nitrogen to  $\text{NO}_x$  emissions. Said test program was carried out employing additional combustors in accordance with the invention, i.e., Combustors C, D, E,  
25 and F.

            Said Combustor C was like Combustor A except that the length of spool 14 (see FIGURE 9) was decreased from 18 inches to 12 inches to provide a

primary combustion region having a length of 30 inches instead of 36 inches the length of spool 18 (see FIGURE 9) was increased from 12 inches to 54 inches to provide a secondary combustion region having a length of 60 inches instead of 18 inches.

5 Said Combustor E was like Combustor A except that the length of spool 18 (see FIGURE 9) was increased from 12 inches to 36 inches to provide a secondary combustion region having a length of 42 inches instead of 18 inches.

Said Combustor F was like Combustor A except that the length of spool 14 (see FIGURE 9) was increased from 18 inches to 42 inches to provide a  
10 primary combustion region having a length of 60 inches instead of 36 inches.

A first portion of said investigation of the effect of residence time comprised changing the length of the primary combustion region as indicated above. A second portion of said investigation comprised decreasing the flow velocity in the primary combustion region from 50 feet per second to 25 feet  
15 per second. In both of said portions of said investigation the test runs carried out included two groups of runs (Group I and Group II), carried out with the base fuel and synthetic fuel respectively, as described above in Example I, for determining the percent conversion of fuel-nitrogen to  $\text{NO}_x$  emissions. The test runs wherein the flow velocity in the primary combustion  
20 region was maintained at 25 feet per second were carried out at the test conditions set forth in Table III below. In analyzing and compiling the data obtained, pertinent data from previous test runs carried out in Combustors A, B, and X were included. This compilation of data is set forth in Table IV-Part 1 and Table IV-Part 2, given below, which summarizes the operating con-  
25 ditions and results obtained.

TABLE III  
OPERATING CONDITIONS FOR COMBUSTION OF HIGH-NITROGEN FUELS

Test Condition	Comb. Oper. Pressure in. Hg Abs.	Inlet Air Temp. °F.	PRIMARY REGION					SECONDARY REGION					DILUTION REGION				
			Fuel lb/hr	φ	Total Air lb/sec	Temp. °F.	Flow Velocity ft/sec	φ	Total Air lb/sec	Temp. °F.	Flow Velocity ft/sec	φ	Total Air lb/sec	Temp. °F.	Flow Velocity ft/sec	φ	
24	32	650	26.2	1.80	0.060	3100	25	0.87	0.063	3690	60	0.30	0.236	1950	102		
25	32	650	23.9	1.70	0.058	3230	25	0.87	0.054	3690	55	0.30	0.215	1950	93		
26	32	650	21.8	1.60	0.056	3350	25	0.87	0.046	3690	50	0.30	0.196	1950	84		
27	32	650	19.6	1.50	0.054	3500	25	0.87	0.039	3690	45	0.30	0.177	1950	76		
28	32	650	18.1	1.40	0.053	3640	25	0.87	0.032	3690	42	0.30	0.163	1950	70		
29	32	650	16.0	1.30	0.051	3750	25	0.87	0.025	3690	37	0.30	0.144	1950	62		
30	32	650	14.4	1.20	0.049	3850	25	0.87	0.018	3690	33	0.30	0.130	1950	56		
31	32	650	13.1	1.10	0.049	3900	25	0.87	0.013	3690	30	0.30	0.118	1950	51		
32	32	650	11.9	1.00	0.049	3900	25	0.70	0.021	3300	34	0.30	0.093	1950	46		
33	32	650	10.9	0.87	0.051	3690	25	0.70	0.013	3300	31	0.30	0.086	1950	42		

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TABLE IV - PART I  
OPERATING CONDITIONS FOR MINIMUM NO<sub>x</sub> EMISSIONS WITH HIGH-NITROGEN FUELS

Compressor Number	Primary-Combustion Region				Swirl Air at Minimum Conversion of Fuel-N to NO <sub>x</sub> <sup>*</sup> %	Test-Fuel Nitrogen, wt %	Emissions, (3% Excess O <sub>2</sub> , Dry) NO <sub>x</sub> from Base-Fuel, * Test-Fuel		Fuel-N Conversion to NO <sub>x</sub> <sup>*</sup> %
	Length in	Flow Velocity, ft/sec	Residence Time, milli-sec	Fuel-Air φ at Minimum Conversion of Fuel-N to NO <sub>x</sub> <sup>*</sup> %			ppmv	ppmv	
5	E	0	50	0	100	0.100	113	153	30.2
						0.500		292	27.0
10						1.000		414	22.8
						1.500		552	22.2
						2.000		644	20.2
	A						135	674	20.5
15	B						147	697	20.9
	C						169	736	21.5
	D						150	674	19.9
	F						138	659	19.8
	D		25				233	766	20.3
	E						221	736	19.6
20	X				74 Set		57	6	21.9

\*Base-Fuel contained 0.0% nitrogen

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TABLE IV - PART II  
OPERATING CONDITIONS FOR MINIMUM NO<sub>x</sub> EMISSIONS WITH HIGH-NITROGEN FUELS

Combustor Number	Primary-Combustion Region				Swirl Air at Minimum Conversion of Fuel-N to NO <sub>x</sub> , %	Test-Fuel Nitrogen, wt %	Emissions, (3% Excess O <sub>2</sub> Dry)		Fuel-N Conversion to NO <sub>x</sub> , %
	Length in	Flow Velocity, ft/sec	Residence Time, millisecc	Fuel-Air $\phi$ at Minimum Conversion of Fuel-N to NO <sub>x</sub>			NO from NO <sub>x</sub>	Base-Fuel, * Test-Fuel, ppmv	
5									
10	36	50	60	1.4	37	0.100	67	101	25.6
						0.500		172	15.8
						1.000		202	10.3
						1.500		224	7.9
						2.000		224	5.9
15	10 Probe		17				42	374	12.6
	16 Probe		27				44	289	9.3
	19 Probe		32				41	254	8.1
	22 Probe		37				43	250	7.9
	28 Probe		47				40	234	7.4
	36 Probe		60				39	194	5.9
20	18		30				89	314	8.6
	30		50	1.1	100		55	242	7.1
	36		60	1.2	77		53	205	5.8
	42		70	1.4	37		54	202	5.7
	60		100	1.4	37		46	169	4.7
25	18	25	60	1.5	100		77	215	5.2
	36		120	1.7	100		67	221	5.8
	14		47	1.1	74 Set		48	228	6.3

\*Base-fuel contained 0.0% nitrogen

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The runs set forth in Part 1 of Table IV were carried out with an equivalence ratio  $\phi$  of 0.87 in the "primary combustion region". With the fuels used, this gives 3 percent excess oxygen (dry basis) in the exhaust gases. Such an operation simulates a one-stage, fuel-lean, combustion process which is common to stationary power plants and thus serves as a base-line performance representative of the prior art. The fact that the conversion of fuel-nitrogen to  $\text{NO}_x$  emissions was only 20 to 30 percent, depending upon the amount of organic or fuel-nitrogen in the fuel, reflects the superior fuel-air mixing characteristics of the combustors of the invention. Such superior mixing is essential for a "low- $\text{NO}_x$  combustor". The residence time in the "primary combustion region" has been designated as zero for data purposes because the combustors were operated as one-stage combustors in these runs and there is thus, in effect, no fuel-rich primary combustion region as such, as there is in a two-stage, rich-lean process.

The runs set forth in Part 2 of Table IV were carried out in accordance with the two-stage, rich-lean combustion method of the invention.

FIGURES 52, 53, 54 and 55 of the drawings set forth the results obtained. FIGURE 52 sets forth the relationship between the Total Minimum  $\text{NO}_x$  Emissions (at the optimum  $\phi$  in the primary combustion region) and Primary Region Residence Time, and includes all pertinent such data set forth in Table IV - Parts 1 and 2, including the runs at both 25 and 50 feet per second flow velocities. FIGURE 53 sets forth the relationship between Minimum Fuel-Nitrogen Conversion To  $\text{NO}_x$  Emissions (at the optimum  $\phi$  in the primary combustion region) and Primary Region Residence Time, and includes all pertinent such data set forth in Table IV - Parts 1 and 2, including the runs at both 25 and 50 feet per second flow velocities. FIGURE 54 sets forth the relationship between the optimum Primary Region Equivalence Ratio For Minimum Fuel-Nitrogen Conversion to  $\text{NO}_x$  emissions and Primary Region Residence Time, and includes all pertinent such data from Table IV - Parts 1 and 2.



FIGURE 55 is a replot of the data in FIGURE 53 to which there has been added the results of a careful sampling of  $\text{NO}_x$  concentration by quartz probes inserted through the wall of Combustor E at the locations indicated in FIGURE 56. During the run in which said probe tests were made said Combustor E was operated at test condition No. 5 in Table I above. It is considered noteworthy that the results obtained by probing the combustor are in complete agreement with the results obtained by varying the length of the primary combustion region and the results obtained by changing flow velocity in said primary region. Said probe results clearly illustrate the formation of  $\text{NO}_x$  from fuel-nitrogen and its subsequent destruction in the fuel-rich primary combustion region.

#### EXAMPLE VI

This example sets forth a continuation of the test programs outlined above in Examples I, II, IV, and V, and completes and establishes the correlation between: (a) the equivalence ratio ( $\phi$ ) in the primary combustion region; (b) the residence time in said primary combustion region; and (c) the percent swirl air (of the total air) introduced into the mixing region and then into said primary combustion region of a combustor, having downstream tangential primary air introduction means, when said combustor is operated in accordance with the two-stage, rich-lean combustion method of this invention.

The results of the test runs of this example are illustrated in FIGURES 57 and 58 of the drawings. Said runs were carried out in the manner described in Example V above, and employing the combustors there described. Said FIGURES 57 and 58 were plotted from data set forth in the above Table IV - Part 2. Said FIGURE 57 sets forth the relationship between residence time in the primary combustion region and the percent swirl air (of the total air) introduced into the mixing region and then into said primary combustion region, for minimum conversion of fuel-nitrogen to  $\text{NO}_x$  emissions. FIGURE 58 sets forth the relationship between the equivalence ratio in the primary combustion region and the percent swirl air (of the total air) introduced into

the mixing region and then into said primary combustion region, for minimum conversion of fuel-nitrogen to  $\text{NO}_x$  emissions. Referring to said FIGURES 57 and 58, it will be noted that in all instances the percent swirl air is always at least about 37 percent.

5

EXAMPLE VII

A test program was carried out to investigate the effect of the concentration of fuel-nitrogen in the previously described base fuel. Said test program was carried out employing the above-described Combustor E. Said test program comprised operating said combustor over a program comprising the  
10 different sets of test or operating conditions set forth in Table I above. Two groups of runs, I and II, were carried out using a base fuel and a synthetic test fuel, respectively, as described in Example I above. FIGURES 59, 60 and 61 of the drawings set forth the results obtained. The data are summarized in Table IV-Parts 1 and 2, above.

15

Referring to FIGURE 59, it will be noted that with all concentrations of fuel-nitrogen the minimum total  $\text{NO}_x$  emissions was obtained at an equivalence ratio of about 1.4 in the primary combustion region.

20

FIGURE 60 shows that, with the exception of the fuel containing only 0.1 percent fuel-nitrogen, the minimum conversion of fuel-nitrogen to  $\text{NO}_x$  emissions was obtained at an equivalence ratio of about 1.4 in the primary combustion region.

25

FIGURE 61 shows that the advantage of the two-stage, rich-lean, combustion process becomes greater as the concentration of fuel-nitrogen in the fuel increases. This is advantageous in that it is anticipated that many fuels in the future will typically contain in the order of 2.0 weight percent fuel-nitrogen, e.g., fuels from shale oil, liquefaction of coal, etc.

Referring to the above Table IV, Parts 1 and 2, it will be noted that the  $\text{NO}_x$  emissions concentration in the exhaust gases when using the fuels of varying fuel-nitrogen content levels off at about 224 ppmv when using the

two-stage, rich-lean combustion process. It is concluded that this indicates an equilibrium exists between the  $\text{NO}_x$  formation and the  $\text{NO}_x$  destruction reactions in the fuel-rich primary combustion region. No such level-off was evident in the one-stage, fuel-lean operation wherein the  $\text{NO}_x$  emissions reached 644 ppmv when the test fuel contained 2.000 weight percent fuel nitrogen.

#### EXAMPLE VIII

A series of runs was carried out to investigate the effect of the orifice 46 located downstream from and adjacent the second air inlet means 30 and the effect of the orifice 48 located in the downstream end portion of the primary combustion section and adjacent the downstream end thereof, and also the effect of the orifice 50 located downstream from the secondary air inlet 34 and in the upstream end portion of the secondary combustion section. See FIGURE 1.

The runs of this example were carried out in Combustor A (see FIGURE 1) and substantially duplicate combustors in which orifices 46 and 48 were removed and in which orifices 46, 48 and 50 were removed. In said runs the combustors were compared at test condition 5 of Table I above, for the purposes of this example.

Based on a comparison at a  $\phi$  of 1.4 in the primary combustion region and a  $\phi$  of 0.87 in the secondary combustion region, the data indicate that removal of the orifices 46 and 48 from the primary combustion section of Combustor A increased the thermal  $\text{NO}_x$  from about 50 to about 85 ppmv. It also decreased the fuel- $\text{NO}_x$  from about 160 to about 150 ppmv. The net result was that the total  $\text{NO}_x$  emissions increased from about 210 to about 235 ppmv. Thus, based on said data it is presently preferred that the combustors utilized be provided with orifices such as said orifices 46 and 48. However, in view of the small effects involved, it is within the scope of the invention to omit said orifices.

Said data also indicated that when said orifice 50 in the secondary combustion section was also removed, there was a marked increase in the

amount of CO emissions, e.g., from about 220 to about 860 ppmv at said 0.87  $\phi$  in the secondary combustion section. Thus, based on said data it is concluded that the combustors of the invention should be provided with a said orifice 50 when it is desired to abruptly terminate the primary combustion region and  
5 achieve adequate mixing in the secondary combustion section for control of CO emissions without resorting to excessive air dilution.

Based on the data given in all the above examples it is concluded that in practicing the method of the invention, using light fuels and the swirl air combustors, there is a definite relationship and/or correlation between:

10 (a) the fuel-air equivalence ratio  $\phi$  in the primary combustion region; (b) the residence time in said primary combustion region; and (c) the percent swirl air (of the total air) introduced into the mixing region and then into said primary combustion region of a combustion zone when said combustion zone is operated in a two-stage, rich-lean manner, e.g., with a fuel-rich primary combustion region  
15 and a fuel-lean secondary combustion region.

FIGURES 38 and 44 show that there is an optimum fuel-air equivalence ratio ( $\phi$ ) for the primary combustion region which must be maintained in order to obtain minimum conversion of fuel-nitrogen to  $\text{NO}_x$ . FIGURES 38 and 45 show that the same is true with respect to total  $\text{NO}_x$  emissions. Thus, if it is  
20 desired to hold the conversion of fuel-nitrogen below a given level, e.g., 10 percent, one must operate the fuel-rich primary combustion region within a relatively narrow range of  $\phi$ . One cannot just maintain the primary combustion region fuel-rich, as has been suggested by some of the prior art. Also, one cannot merely maintain the value of  $\phi$  above a certain value, e.g., above about  
25 1.43 as has been suggested by some of the prior art. This is true because, as shown in FIGURES 38 and 44, as  $\phi$  increases above the optimum minimum value, the conversion of fuel-nitrogen to  $\text{NO}_x$  emissions also increases. Thus, based on the data set forth in said FIGURES 38 and 44, it is concluded that if the fuel-rich primary combustion region is operated with too high a value for  $\phi$ , i.e.,

too rich, the partial combustion in said primary region is too incomplete and too much partially combusted fuel is carried over into the secondary combustion region where thermal  $\text{NO}_x$  can form. Also, fuel-nitrogen will be carried over into the secondary combustion region where it will be converted to  $\text{NO}_x$  emissions. Under these circumstances one is losing the advantage of a two-stage, fuel-rich, fuel-lean, process.

However, the data of the above examples show that one cannot merely consider only the value of  $\phi$  in the primary combustion region. It was surprising to discover that one must also correlate the residence time in the primary combustion region with the  $\phi$  in said primary region. The data herein show that the optimum  $\phi$  in the primary combustion region for minimum conversion of fuel-nitrogen to  $\text{NO}_x$  increases with increasing residence time in said primary combustion region. FIGURE 54 herein shows that said optimum  $\phi$  is only slightly fuel-rich ( $\phi = 1.05$  to  $1.1$ ) at a residence time of 30 milliseconds and increases to very fuel-rich ( $\phi = 1.7$ ) at a residence time of 120 milliseconds. Thus, based on the data herein, e.g., said FIGURES 38, 44, and 54, it is concluded that one must consider, and correlate, both the value of  $\phi$  and the residence time in the primary combustion region in order to operate a two-stage, rich-lean combustion process so as to obtain reduced conversion of fuel-nitrogen to  $\text{NO}_x$  emissions.

It is presently believed that said variables of  $\phi$  and residence time in the primary combustion region are two of the most important variables in the operation of a two-stage, rich-lean combustion process, because when said variables are properly correlated one will obtain markedly reduced conversion of fuel-nitrogen to  $\text{NO}_x$  emissions. However, when introducing swirl air at a downstream location, the data in the above examples also show that there is a third important variable which should be considered if one is to obtain the best results, e.g., to insure that one obtains the minimum conversion of fuel-nitrogen to  $\text{NO}_x$  emissions. Said third variable is the percent swirl air (of

the total air) introduced into the mixing region and then into the primary combustion region of the combustion zone.

FIGURE 57 shows that the percent swirl air to obtain minimum conversion of fuel-nitrogen to  $\text{NO}_x$  emissions decreases with increasing residence time in the primary combustion region. FIGURE 58 shows that the percent swirl air to obtain minimum conversion of fuel-nitrogen to  $\text{NO}_x$  emissions decreases with increasing equivalence ratio in the primary combustion region. These were surprising and unexpected results when one also considers: (a) the showing in FIGURES 38 and 44 that there is an optimum fuel-air equivalence ratio in the primary combustion region for minimum conversion of fuel-nitrogen to  $\text{NO}_x$  emissions; and (b) the showing in FIGURE 28 that the optimum fuel-air equivalence ratio in the primary combustion region increases with increasing residence time in said primary region. Thus, it was surprising and unexpected to discover that one should correlate the percent swirl air introduced into the mixing region with both the equivalence ratio and the residence time in the primary combustion region, if one is to obtain the best results, i.e., the minimum conversion of fuel-nitrogen to  $\text{NO}_x$  emissions.

It is concluded from the results illustrated in the above examples that the percent swirl air should be used in the practice of the invention to complement both of said other operating variables (a) the fuel-air equivalence ratio in the primary combustion region and (b) the residence time in said primary combustion region, so as to obtain the minimum conversion of fuel-nitrogen to  $\text{NO}_x$  emissions.

In this manner all three of said interacting operating variables can be correlated to provide a flexible, efficient method which can be used with a wide variety of high nitrogen content fuels over a wide range of operating conditions to obtain the minimum conversion of fuel-nitrogen to  $\text{NO}_x$  emissions.

EXAMPLE IX

In an additional series of tests a crude shale oil, designated herein as shale oil A, was used as a fuel. The characteristics of this particular oil are set forth in Table V below:

5

TABLE VShale Oil A

	Flash Point, °F	215
	Pour Point, °F	85
	Water and Sediment, vol %	1.60
10	Carbon Residue on 10% Bottoms, %	2.40
	Ash, wt %	0.09
	Kinematic Viscosity, cSt	
	@ 100 F	40.0
	@ 122 F	21.6
15	Specific Gravity, 60/60 °F	0.9279
	Sulfur, wt %	0.69
	Nitrogen, wt %	1.93
	Carbon, wt %	84.47
	Hydrogen, wt %	11.66
20	Oxygen, wt %	1.18
	Heat of Combustion, Btu/lb	18,427
	Metal Content, ppm	
	Fe	75
	V	0.38
25	As	29.5
	Ni	2.17
	Pb	5.0
	Hg	0.13
	Na	18.0

30

In this series of tests, in addition to testing an actual crude shale oil, as opposed to a light fuel oil doped with pyridine to provide a high nitrogen content, three different burner configurations were also tested and compared. The combustor referred to herein as Combustor G is similar to that illustrated in FIGURES 1 and 2 of the drawings but was modified in order to

adapt the same to the heavy shale oil. The specific structure of the fuel-air mixing means is shown at FIGURE 10 of the drawings. The fuel oil nozzle 24 was that shown in FIGURES 20 through 22. This type nozzle was selected in order to attain as great an atomization of the heavy oil as possible prior to the introduction of the fuel into the flame tube. Combustor G also included a variable dome primary air inlet 164 which is illustrated in greater detail in FIGURES 11, 12, 13 and 14 of the drawings. The means for introducing the air in a swirling direction through air inlets 28 and 30 was the same as that previously illustrated and described in connection with FIGURES 3, 4 and 5. The combustor also included the nozzle-type means for introducing secondary air and terminating the primary combustion region shown in FIGURE 17 of the drawings.

The combustor referred to herein as combustor H is shown in FIGURE 15 of the drawings. In this case, the same fuel nozzle and variable primary air inlet of Combustor G were utilized as well as the nozzle-type means for terminating the primary zone.

The combustor referred to herein as Combustor I (FIGURE 16) included the same fuel nozzle of Combustor G and the same general configuration of variable primary air inlet except that the variable air inlet was the sole primary air inlet (in addition to the fuel nozzle) and included angular fins 190 designed to introduce the air in a tangential or swirling manner.

In all instances, the nominal length of the portion of primary combustion region measured from the midpoint of the primary air conduit 162 to the midpoint of the secondary air orifices was 58 inches and the nominal length of the secondary combustion region measured from the midpoint of the secondary air inlet ports to the midpoint of the dilution air inlet ports as shown in FIGURE 1 of the drawings was 33 inches. The maximum internal diameter of the flame tube including the major portion of the primary combustion region and the major portion of the secondary combustion region was 6 inches. Obviously, in an actual commercial burner the flame tube would be larger but it would be scaled



up in accordance with essentially the same ratio of internal diameter to length in order to attain the desired configuration necessary for attaining proper residence times, proper mixing and optimum reduction of  $\text{NO}_x$ . In this series of tests, the fuel-air equivalence ratio in the primary combustion zone was varied from 0.87 to 1.7. Air pressure to the fuel atomizer was 100 psig, the air temperature to the fuel atomizer was 650°F and the fuel temperature to the fuel atomizer was 250°F. The average heat input rate for the single stage combustion (fuel-air equivalence ratio of 0.87 throughout) was 757,000 BTU per hour while that for the two-stage combustion was 744,000 BTU per hour. The average heat release rate was 498,000 BTU per hour per cubic foot for the single stage combustion and 489,000 BTU per hour per cubic foot for the two-stage combustion. The primary zone residence time was estimated to be 110 milliseconds. The remaining test conditions are set forth in the following Table VI, in which the primary zone temperature, the primary zone flow velocity and the secondary zone temperature are estimated values. The base data for thermal  $\text{NO}_x$  was obtained on #1 fuel oil with 0% nitrogen.

The results of this series of tests are set forth in FIGURE 62 of the drawings in which the curve for Combustor G represents an average of eight runs, the curve for Combustor H represents an average of ten runs and that for Combustor I is an average of two runs. Obviously, runs made at a fuel-air equivalence ratio of 0.87 were single stage runs, since the entire burner was operated in a conventional fuel-lean manner.

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TABLE VI

## OPERATING CONDITIONS FOR COMBUSTION OF HIGH-NITROGEN FUELS

Test Condition	Comb. Oper. Pressure in. Hg Abs.	Inlet Air Temp. °F.	PRIMARY REGION				SECONDARY REGION				DILUTION REGION				
			Fuel lb/hr	Fuel $\phi$	Total Air lb/sec	Temp. °F.	Flow Velocity ft/sec	$\phi$	Total Air lb/sec	Temp. °F.	Flow Velocity ft/sec	$\phi$	Total Air lb/sec	Temp. °F.	Flow Velocity ft/sec
34	32	800	49.9	1.70	0.112	3360	50	0.87	0.106	3750	108	0.30	0.414	2150	194
35	32	800	47.7	1.65	0.110	3420	50	0.87	0.098	3750	103	0.30	0.396	2150	185
36	32	800	45.5	1.60	0.108	3480	50	0.87	0.091	3750	98	0.30	0.378	2150	177
37	32	800	43.4	1.55	0.107	3540	50	0.87	0.083	3750	94	0.30	0.361	2150	169
38	32	800	41.4	1.50	0.105	3600	50	0.87	0.076	3750	89	0.30	0.344	2150	161
39	32	800	39.5	1.45	0.104	3650	50	0.87	0.069	3750	85	0.30	0.328	2150	153
40	32	800	37.7	1.40	0.102	3700	50	0.87	0.062	3750	81	0.30	0.313	2150	146
41	32	800	36.1	1.35	0.102	3750	50	0.87	0.056	3750	78	0.30	0.300	2150	140
42	32	800	34.2	1.30	0.100	3800	50	0.87	0.049	3750	74	0.30	0.284	2150	133
43	32	800	32.5	1.25	0.099	3850	50	0.87	0.043	3750	70	0.30	0.270	2150	126
44	32	800	30.9	1.20	0.098	3900	50	0.87	0.037	3750	67	0.30	0.256	2150	120
45	32	800	27.8	1.10	0.096	3980	50	0.87	0.025	3750	60	0.30	0.230	2150	108
46	32	800	25.6	1.00	0.097	3920	50	0.70	0.042	3400	63	0.30	0.185	2150	99
47	32	800	23.2	0.87	0.101	3750	50	0.70	0.025	3400	57	0.30	0.163	2150	90

Several significant observations can be made by reference to FIGURE 62 of the drawings. First, it can be seen that, for the particular high nitrogen, heavy shale oil tested, thermal  $\text{NO}_x$  alone is substantially higher than the Federal limits when a combustor is operated as a single stage combustor. Even when operating the combustors in a two-stage, rich-lean manner, and under indicated operating conditions, the thermal  $\text{NO}_x$  represented nearly one-third of the total  $\text{NO}_x$  level permitted by Federal regulations. However, with each of the three burners it was possible to reduce the total  $\text{NO}_x$  content of the flue gases to values which essentially meet the Federal limitations.

By comparing FIGURE 62 with FIGURES 39, 45, 49 and 59 of the drawings, it is obvious that the minimum  $\text{NO}_x$  emissions were obtained at higher fuel-air equivalence ratios in the primary combustion region while operating with the heavy shale oil, as opposed to the previously tested light oil containing essentially the same amount of nitrogen. Thus, this shift in the fuel-air equivalence ratio for the primary combustion region can be attributed to the heavier character of the shale oil.

It is also interesting to note that minimum  $\text{NO}_x$  emissions were obtained at slightly different equivalence ratios for the three different combustors. Thus, since the three combustors differed primarily in the means for introducing primary combustion air and mixing the same with the fuel, the necessary equivalence ratio for minimum  $\text{NO}_x$  emissions is obviously dependent upon the nature of the fuel-air mixing and the intimacy of contact attained.

FIGURE 63 of the drawings is a plot of residence time in the primary combustion region versus minimum  $\text{NO}_x$  emissions from the data obtained at a primary combustion region equivalence ratio of 1.6 (test condition 36). This was considered to be the optimum equivalence ratio for Combustor H in the presently discussed series of runs. It is interesting to note from FIGURE 63 that minimum  $\text{NO}_x$  emissions are obtained when utilizing a primary combustion region residence time above about 35 milliseconds. However, still better

results are obtained at a residence time above about 100 milliseconds. From 100 milliseconds to 140 milliseconds residence time no noticeable reduction in  $\text{NO}_x$  emissions occurs. Accordingly, for optimum results, in this particular combustor operating on this particular shale oil, a preferred residence time is at least about 100 milliseconds and there is no reason to extend the residence time beyond about 140 milliseconds. Thus, in a commercial burner, there is no advantage in extending the length of the primary combustion region beyond that necessary to attain residence times between 140 and 100 milliseconds at acceptable primary combustion region flow velocities.

Comparing FIGURE 63 with FIGURES 52 and 53 of the drawings, it is to be seen that the minimum primary region residence times are somewhat higher for the heavy shale oil, as compared with the lighter oil doped with pyridine.

#### EXAMPLE X

The relationship of the primary zone equivalence ratio and the primary zone residence time is illustrated by FIGURE 64 of the drawings. The data plotted in FIGURE 64 are for shale oil A and combustor H. Table VII below sets forth the conditions employed in conducting the series of tests summarized in FIGURE 64. The tests were run by selecting different lengths for the primary combustion region from zero through seven feet (fuel-air equivalence ratios from 0 to 140 milliseconds), maintaining a constant primary region flow velocity of 50 feet per second and varying the primary zone equivalence ratio to obtain minimum  $\text{NO}_x$  emissions at the particular primary zone residence time.

TABLE VII

Primary Combustion Region

	<u>Length - Feet</u>	<u>Residence Time - Milliseconds</u>
5	0	0
	1.0	20
	2.0	40
	3.5	70
	5.0	100
10	7.0	140
	Inlet Air Pressure	32 in Hg abs
	Inlet Air Temperature	800°F
	Air Pressure to Fuel Nozzle	100 psig
	Air Temperature to Fuel Nozzle	650°F
15	Fuel Temperature to Fuel Nozzle	250°F
	Heat Input Rate	838,000 Btu/hr
	Primary Region Flow Velocity (Estimated)	50 ft/sec
	Secondary Region Equivalence Ratio	0.87
20	Secondary Region Temperature (Estimated)	3750°F

It is to be observed that the curve of FIGURE 64 is quite comparable in character to that of FIGURE 54. At a residence time of about 35 milliseconds, there is a break in the curve which clearly indicates that the minimum residence time set forth herein of 35 milliseconds represents a change in character. From 35 milliseconds to about 100 milliseconds, the curve follows a gradually increasing path and above 100 milliseconds the curve tends to flatten out. Thus, the heavier shale oil tends to shift the curve toward higher residence times when both the equivalence ratio and residence time are considered. By the same token, FIGURE 64 shows that, for the heavier shale oil, there is also a shift in the equivalence ratio to higher equivalence ratios.

EXAMPLE XI

FIGURE 65 of the drawings shows the importance of adjusting the ratio of the axial flow component and the radial flow component by utilizing a variable dome air inlet when operating with heavy oils. The data plotted in FIGURE 65 was obtained utilizing combustor H and shale oil A. In this particular instance, the dome opening was varied from 15 percent open to 97 percent

open at each of the designated primary combustion region equivalence ratios. The data plotted are averages of minimum  $\text{NO}_x$  emissions obtained in a plurality of tests under test conditions 34, 36, 38, 40, 42 and 44, as previously set forth in Table VI.

5           It is to be observed by a study of FIGURE 65 that at low equivalence ratios, which are too low to obtain optimum  $\text{NO}_x$  emissions, the minimum  $\text{NO}_x$  emission is obtained with the dome essentially 100 percent open. However, as the optimum equivalence ratio is approached in this particular case, the curve begins to flatten out and reverse as shown at an equivalence ratio of 1.5. At  
10           an equivalence ratio of 1.5, minimum  $\text{NO}_x$  emissions are obtained at a dome opening of about 15 percent open. However, the dome opening becomes quite significant at the optimum equivalence ratio of 1.6 (lowest  $\text{NO}_x$  emissions). This obviously illustrates the necessity of not only correlating the operating variables of primary zone residence, equivalence ratio and dome opening, but of  
15           adjusting these variables for each particular fuel oil. Thus, at the optimum equivalence ratio, there is also a dome opening which attains minimum  $\text{NO}_x$  emissions. In this particular instance, this opening occurs at about 47 percent open. Therefore, it is quite obvious that when operating on this particular heavy shale oil, the adjustment of the variable primary air inlet  
20           is most significant and, in fact, is necessary to the proper operation of the combustor. Obviously, when a fuel which has not been previously burned in a given two-stage combustor is to be utilized, in addition to determining the optimum residence time and the optimum fuel air equivalence ratio, it is necessary to determine the optimum dome opening for the variable air inlet in  
25           order to attain optimum mixing of the fuel and air in the primary combustion region and stabilize the flame adjacent the upstream end of the primary combustion region. It is also important that the variable opening dome be employed in commercial operations so that adjustments can be made for changes in the character of fuel to be burned as well as changes in the properties of a

particular fuel. Obviously, any given fuel supplied over a long period of time will vary to some extent, particularly with respect to volatility. Accordingly, should the  $\text{NO}_x$  emission level increase at any time, it can normally be brought back to the desired minimum level by simply adjusting the adjustable dome.

5           While, as previously indicated, the combustors of the present invention will operate to produce minimum  $\text{NO}_x$  emissions without a variable dome, it is necessary in such cases to design a burner for a particular fuel and no adjustments can be made should the fuel be changed or should the properties of a particular fuel vary during use of the combustor. While adjustments can be  
10 made in the primary region flow velocity and equivalence ratio, it is much simpler to be able to adjust the variable air introduction means to attain the desired degree of mixing when only the volatility of the fuel differs.

FIGURE 66 of the drawings is a replot of the data of FIGURE 65 on an expanded scale at the optimum equivalence ratio of 1.6, so as to emphasize the  
15 importance of adjusting the dome opening, once the optimum residence time and optimum equivalence ratio have been set.

The adjustment of the variable air can be carried out manually or automatically. In the latter instance an appropriate controller unit, such as controller unit 109 of FIGURE 6, would be actuated by a signal from an appropriate  
20 means for measuring the  $\text{NO}_x$  content of the flue gas.

#### EXAMPLE XII

Another series of tests was conducted in which a different shale oil, having a lower concentration of bound nitrogen, was burned in Combustor H. The conditions of this series of tests are set forth in the Table VIII below and  
25 the characteristics of this shale oil, referred to herein as shale oil B, are set forth in Table IX. In order to compare the results obtained with those obtained in the same combustor (H) and utilizing shale oil A, the plot of equivalence ratio versus  $\text{NO}_x$  emissions for combustor H and shale oil A of  
FIGURE 62 was replotted in FIGURE 67 along with the data obtained when operating  
30 on shale oil B.

It is to be observed that while the two shale oils have essentially the same characteristics, including volatilities, the difference in the level of bound nitrogen has a tendency to shift the optimum equivalence ratio. For the shale oil containing the lesser amount of bound nitrogen, the optimum  
5 equivalence ratio occurs at about 1.5, while, as previously indicated, the higher nitrogen fuel exhibits optimum results at an equivalence ratio of about 1.6.

FIGURE 68 of the drawings illustrates the conversion of fuel-nitrogen to  $\text{NO}_x$  emissions at various equivalence ratios from the data obtained on shale  
10 oil B. It can be seen that the conversion of fuel-nitrogen to  $\text{NO}_x$  can be reduced to the desired 5 percent when operating in accordance with the present invention.



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TABLE VIII  
OPERATING CONDITIONS FOR COMBUSTION OF SHALE OIL D

	<u>Burner Pressure in Hg abs</u>	<u>Air Inlet Temperature, F</u>	<u>Primary Air Flow, lb/hr</u>	<u>Fuel Flow, lb/hr</u>	<u>Equivalence Ratio</u>	<u>Estimated Primary Zone Temperature, F</u>	<u>Estimated Primary Zone Velocity, ft/sec</u>
5	32	650	432	58	1.8	3100	50
	32	650	414	53	1.7	3230	50
	32	650	403	48	1.6	3350	50
	32	650	389	43	1.5	3500	50
10	32	650	385	40	1.4	3640	50
	32	650	364	35	1.3	3750	50
	32	650	356	32	1.2	3850	50
	32	650	353	29	1.1	3900	50
	32	650	353	26	1.0	3900	50
15	32	650	371	24	0.87	3690	50

Equivalence Ratio in Secondary Zone = 0.87 (37 Excess Oxygen, Dry)

Equivalence Ratio in Dilution Zone = 0.30 (Estimated 1950°F)

0073265

TABLE IX

PHYSICAL AND CHEMICAL PROPERTIES OF SHALE OIL B

	Flash Point, °F	160
	Pour Point, °F	65
5	Water and Sediment, vol %	0.3
	Carbon Residue on 10% Bottoms, %	2.11
	Ash, wt %	0.01
	Kinematic Viscosity, cSt	
	@ 100 F	30.9
10	@ 122 F	17.14
	Specific Gravity, 60/60 °F	0.9254
	Sulfur, wt %	0.46
	Nitrogen, wt %	1.85
	Carbon, wt %	82.80
15	Hydrogen, wt %	11.09
	Oxygen, wt %	1.39
	Heat of Combustion, Btu/lb	18,461
	Metal Content, ppm	
	Fe	37
20	V	0.79
	As	27.1
	Ni	1.85
	Pb	< 10
	Hg	0.22
25	Na	0.16

EXAMPLE XIII

In order to illustrate the necessity of also maintaining a minimum residence time in the secondary combustion region to attain substantially complete and efficient combustion of the fuel, data has been plotted as FIGURE 69 for a #4 fuel oil, doped with pyridine to obtain a 2.0% nitrogen content. The test conditions are set forth in Table X.

As previously indicated, by maintaining carbon monoxide emissions below about 300 parts per million, efficient utilization of the fuel can be

TABLE X  
OPERATING CONDITIONS FOR COMBUSTION OF HIGH-NITROGEN FUELS

5	Comb. Oper. Pressure in. Hg Abs.	Inlet Air Temp. °F.	Fuel lb/hr	PRIMARY REGION			SECONDARY REGION			DILUTION REGION					
				Total Air lb/sec	Temp. °F.	Flow Velocity ft/sec	Residence Time Ms	Total Air lb/sec	Temp. °F.	Flow Velocity ft/sec	Total Air lb/sec	Temp. °F.	Flow Velocity ft/sec		
	32	650	39.3	1.50	0.108	3500	50	0.87	0.077	3900	90	0.30	0.354	1950	152
	32	1000	76.1	1.50	0.208	3620	100	0.87	0.151	3970	187	0.30	0.682	2330	341
10	32	650	39.3	1.50	0.108	3500	50	0.87	0.077	3900	90	0.30	0.354	1950	152
	32	650	39.3	1.50	0.108	3500	50	0.87	0.077	3900	90	0.30	0.354	1950	152
	32	650	39.3	1.50	0.108	3500	50	0.87	0.077	3900	90	0.30	0.354	1950	152
	Shale														

\*Doped with pyridine to obtain 20% fuel nitrogen.

SAMPLE XIV

A series of tests was conducted utilizing prototype burner L of FIGURE 28. This burner had 4-inch thick ceramic walls and was designed to have a maximum pressure drop of 6 inches of water at its rated heat release of 10 MM BTU/Hr when operated with an air inlet temperature of 800°F. This burner was installed in the furnace shown IN FIGURE 33, utilizing the air supply shown in FIGURES 34 and 35. This air supply allowed preheating of the air, without vitiation, to 600°F. By metering the total air flow to the burner and the air flow to the primary combustion zone it was possible to establish the desired stoichiometry in both the primary and secondary combustion zones.

The performance of this prototype burner during the initial testing period was found to be acceptable. The burner lit easily and the air-jet mixing configuration displayed a stable flame without audible resonance. The pressure drop across the burner was less than the target of 6 inches of water, maximum, at the rated heat release (10 MM BTU/Hr). Combustion efficiency was 100 percent with excess oxygen levels in the exhaust gas down to 1 percent and with no visible exhaust smoke. The fuel-rich primary zone was free of deposits after extended operation on No. 6 fuel oil. The burner response to changing primary-zone stoichiometry is shown in FIGURE 70, i.e., NO<sub>x</sub> emissions were high when operated as a one-stage burner ( $\phi = 0.87$ ) and decreased markedly with two-stage operation, reaching a minimum with a primary zone equivalence ratio of about 1.5.

The performance of prototype burner L of FIGURE 27, in the same furnace and utilizing the same air system, is also shown in FIGURE 70. This approximates a high intensity vortex mixing system. While the level of NO<sub>x</sub> emissions dropped below that achieved with the burner of FIGURE 28, a high level of combustion instability was encountered which was evidenced by intolerable resonance. It is well known that resonance, or screeching combustion, can accelerate heat and mass transfer to improve fuel-air mixing.

The utilization of the test furnace of FIGURE 33 proved to be an unfortunate choice because of the relatively high temperatures achieved in the

furnace which, in turn, resulted in relatively high levels of "thermal NO<sub>x</sub>". As a result, testing of the prototype burners in this furnace was terminated at this point.

#### EXAMPLE XV

5           Prototype burners L and M of FIGURES 27 and 28, respectively, were also installed in a furnace configuration shown in FIGURE 36, utilizing the air supply system of FIGURE 37. As indicated, the test furnace of FIGURE 36 had a steel wall that was water cooled. As a result, the stack gas temperatures were reduced significantly, and this decreased the level of thermal NO<sub>x</sub> to a toler-  
10           able level, as illustrated by the data presented in FIGURE 71. The lower furnace temperature achieved in this environment is characteristic of that existing in process heaters which is the duty which was sought to be simulated in the test program. The performance of the prototype burners during this final  
15           period of testing continued to be acceptable, i.e., the burners were easy to light, the flame was compact, the exhaust gas was clean, there were no problems with distortion of the air registers or other hardware, and the burners proved to be durable.

          With the burner of FIGURE 28 (burner M) the data obtained burning No. 6  
fuel oil are presented in FIGURE 72. Similarly, the data obtained burning raw  
20           shale oil are presented in FIGURE 73. A statistical analysis of these data was made to determine the 95 percent confidence interval on the true mean value of NO<sub>x</sub> emissions at the minimum levels for each mixture-response curve, and the results were very acceptable. As an example, for No. 6 fuel oil at an inlet air  
temperature of 90°F, the minimum level of NO<sub>x</sub> emissions was observed to be  
25           145 ppmv at an equivalence ratio of 1.47, and the 95 percent confidence interval on the true mean value is 149± 8 ppmv NO<sub>x</sub>. Similar levels of the experimental air were found in other test conditions and with other fuels. It should be noted that no attempt was made during this investigation to optimize fuel-air  
mixing by varying the axial and radial components of the primary air flow, nor  
30           was any attempt made to optimize burner length (residence time) for minimum NO<sub>x</sub>

emissions. Based upon previous operating experience with smaller burners, it is anticipated that further reductions in NO<sub>x</sub> emissions could be made if the fuel-air mixing and the residence time were optimized in the prototype burner.

5 Burner L of FIGURE 27 was also tested and comparable levels of NO<sub>x</sub> emissions were obtained, as shown in FIGURE 74 for No. 6 fuel oil and FIGURE 75 for raw shale oil. Unfortunately, the operation of this burner was accompanied by resonant combustion, which varied in intensity from a moderate buzz to an intense screech. Therefore this burner would be considered unacceptable for commercial operation. However, modification to eliminate the problem of  
10 resonance is within the skill of one skilled in the art.

Finally, the performance of the prototype burners is summarized in FIGURE 76 for operation on No. 6 fuel oil and FIGURE 77 for operation on raw shale oil.

By way of summary, in accordance with the present invention, for most  
15 fuels and the various combustors hereof, the operating variables are within the limits set forth below.

The primary region fuel-air equivalence ratio should be between 1.0 and 1.8. Preferably, for a light fuel with high concentrations of nitrogen the equivalence ratio is between about 1.05 and about 1.7 and ideally about 1.14 to  
20 about 1.56. For a heavier oil, such as shale oil, the preferred range is about 1.3 to about 1.7 and ideally about 1.4 to about 1.65.

The residence time within the primary combustion region should be between about 30 milliseconds and about 140 milliseconds. For a light fuel the preferred range is about 30 to about 120 milliseconds and ideally between about  
25 45 and about 75 milliseconds. For heavy fuels the preferred range is about 35 to about 140 milliseconds and ideally about 100 to 140 milliseconds.

When primary air is also introduced in a swirling fashion at a point, downstream from the point of introduction of primary air immediately adjacent the fuel inlet, the volume of swirl air should be about 35 to about 100% of the  
30 total primary air utilized.

The fuel-air equivalence ratio in the secondary combustion region should be such as to produce an overall fuel-air equivalence ratio between about 0.50 and 1.0, preferably between 0.75 and 0.87 and most preferably 0.87.

5 The residence time in the secondary combustion region should be at least about 15 milliseconds and preferably at least about 30 milliseconds.

10 Stated in terms of the results to be obtained, the fuel nitrogen of the fuel should be converted to not more than 10%  $\text{NO}_x$ , preferably not more than 7.5% and ideally not more than 5.0% and the CO content of the flue gas should be less than about 300 ppmv. For a powdered fuel, such as a typical Western Kentucky coal, these fuel  $\text{NO}_x$  limits of 10%, 7.5% and 5.0% represent about 0.350, 0.262 and 0.175 lb. fuel  $\text{NO}_x$ /Million Btu, respectively. Corresponding values for a typical crude solvent refined coal oil are 0.216, 0.162 and 0.108, respectively; for shale oil A, exemplified herein, 0.344, 0.258 and 0.172, respectively; for shale oil B, exemplified herein, 0.329, 0.247, and 0.164, respectively; and for 15 the light fuel with 2.0% bound nitrogen, exemplified herein, 0.322, 0.242 and 0.161, respectively. Accordingly, the operating variables should be selected and correlated to reduce the fuel  $\text{NO}_x$  emissions to less than about 0.350 lb./Million Btu, preferably to less than about 0.290 and ideally to less than 0.180 and the total  $\text{NO}_x$  in the flue gas to less than 0.450, 0.365 and 0.275 lb. total 20  $\text{NO}_x$ /MM Btu, respectively.

The term "air" is employed generically herein and in the claims to include air and other combustion-supporting gases.

25 While the invention has been described above in terms of using a liquid fuel, the invention is not limited to the use of liquid fuels. It is within the scope of the invention to use vaporous or gaseous fuels, including prevaporized liquid fuels. It is also within the scope of the invention to use finely divided solid fuels, e.g., powdered coal.

30 While certain embodiments of the invention have been described for illustrative purposes, the invention is not limited thereto. Various other modifications or embodiments of the invention will be apparent to those skilled in the art in view of this disclosure. Such modifications or embodiments are within the spirit and scope of the disclosure.

## CLAIMS

1. A method for burning a fuel, including:

(1) intimately mixing said fuel and a first volume of air to obtain a homogeneous mixture of fuel and air having a fuel/air equivalence ratio above the stoichiometric ratio by introducing all of said fuel as at least a part of a central, axial flow component and introducing at least one stream of said first volume of air as a radial flow component;

(2) burning a substantial portion, but less than all, of said fuel under fuel-rich combustion conditions and without the addition of air other than said first volume of air to produce a fuel-rich flame front;

(3) terminating said fuel-rich combustion conditions, at least in part, by introducing a sufficient amount of a second volume of air to produce an overall fuel/air equivalence ratio below the stoichiometric ratio and a fuel-lean flame front; and

(4) continuing the burning of said fuel under fuel-lean combustion conditions.

2. A method in accordance with claim 1 wherein the fuel-rich combustion conditions are terminated and the fuel-lean flame front is produced abruptly .

3. A method in accordance with claim 1 wherein the fuel-rich combustion conditions are terminated and the fuel-lean flame front is produced abruptly by introducing the second volume of air into the



fuel-rich flame front from a peripheral location smaller than the periphery of the fuel-rich flame front.

4. A method in accordance with claim 1 wherein the fuel-rich combustion conditions are terminated and the fuel-lean flame front is produced abruptly by reducing the peripheral dimensions of the fuel-rich flame front, introducing the second volume of air at at least one of (a) a point immediately prior to said reduction of the peripheral dimensions of said fuel-rich flame front and (b) a point within the region of reduced peripheral dimensions of said fuel-rich flame front and then expanding the fuel-lean flame front thus produced.

5. A method in accordance with at least one of claims 1 to 4 wherein the fuel/first volume of air ratio is above a fuel/air equivalence ratio at which the  $\text{NO}_x$  pollutant content of the effluent from the burning method is below a predetermined concentration but below a fuel/air equivalence ratio at which the  $\text{NO}_x$  pollutant content of said effluent from said burning method is below said predetermined value.

6. A method in accordance with at least one of claims 1 to 5 wherein the peripheral dimensions of the mixture of fuel and the first volume of air are reduced at a point immediately after contact of the axial flow component with the radial flow component and is then expanded to the full peripheral dimensions of the fuel-rich flame front.

7. Apparatus for burning a fuel, including:

(1) mixing means for intimately mixing said fuel and a first volume of air to obtain a homogeneous mixture of fuel and air having a fuel/air equivalence ratio above the stoichiometric ratio by

introducing all of said fuel as at least a part of a central, axial flow component and introducing at least one stream of said first volume of air as a radial flow component;

(2) an elongated fuel-rich combustion means adapted to burn a substantial portion, but less than all, of said fuel under fuel-rich combustion conditions to produce a fuel-rich flame front and substantially impervious to the ingress of additional air other than said first volume of air;

(3) air introduction means adapted to introduce a second volume of air into said fuel-rich flame front in an amount sufficient to produce an overall fuel/air equivalence ratio below the stoichiometric ratio and a fuel-lean flame front; and

(4) fuel-lean combustion means adapted to continue the burning of said fuel under fuel-lean combustion conditions.

8. Apparatus in accordance with claim 7 wherein the means for introducing the at least one stream of the first volume of air as a radial flow component includes means for introducing said air from a plurality of peripheral openings inwardly toward the axial flow component.

9. Apparatus in accordance with claim 7 wherein the means for introducing the at least one stream of the first volume of air includes means for introducing said air from a peripheral location in a swirling manner and inwardly toward the axis of the axial flow component.

10. Apparatus in accordance with claim 9 wherein the means for introducing the at least one stream of the first volume of air in a

swirling manner includes two, spaced means adapted to introduce two streams of said first volume of air in swirling, opposite directions.

11. Apparatus in accordance with at least one of claims 7 to 10 wherein the means for intimately mixing the fuel and the first volume of air includes means for reducing the peripheral dimensions of said fuel-air mixture immediately after contact of the axial flow component with the radial flow component and then expanding said fuel-air mixture to the full peripheral dimensions of the fuel-rich combustion means.
12. Apparatus in accordance with at least one of claims 7 to 11 wherein the means for introducing the second volume of air includes means for abruptly terminating the fuel-rich combustion means and initiating the fuel-lean combustion means.
13. Apparatus in accordance with claim 12 wherein the means for abruptly terminating the fuel-rich combustion means and initiating the fuel-lean combustion means includes means for reducing the peripheral dimensions of the fuel-rich combustion means and thereafter expanding the peripheral dimensions to the peripheral dimensions of the fuel-lean combustion means and means for introducing the second volume of air at at least one of (a) a point immediately preceding said means for reducing the peripheral dimensions of the fuel-rich combustion means and (b) a point within the means for reducing the peripheral dimensions of the fuel-rich combustion means itself.
14. Apparatus in accordance with at least one of claims 7 to 13 wherein the means for intimately mixing the fuel and the first volume of air includes means for adjusting the ratio of the axial flow component and the radial flow component.



15. Apparatus for burning a fuel, particularly for fuel-rich combustion of a fuel and production of a fuel-lean flame front, including:

(1) mixing means for intimately mixing said fuel and a first volume of air to obtain a homogeneous mixture of fuel and air having a fuel/air equivalence ratio above the stoichiometric ratio;

(2) fuel-rich combustion means adapted to burn a portion, but less than all, of said fuel under fuel-rich combustion conditions to produce a fuel-rich flame front and substantially impervious to the ingress of additional air other than said first volume of air;

(3) termination means for terminating said fuel-rich combustion conditions, at least in part, by introducing a sufficient amount of a second volume of air to produce an overall fuel/air equivalence ratio below the stoichiometric ratio and a fuel-lean flame front; and

(4) discharge means for discharging said fuel-lean flame front into the heated section of a furnace.

16. Apparatus in accordance with claim 15 wherein the means for introducing and mixing the fuel and the first volume of air includes means for introducing said fuel as at least a part of a central, axial flow component and means for introducing at least a first stream of said first volume of air as a radial flow component.

17. Apparatus in accordance with claim 16 wherein the radial flow component includes a plurality of air inlets spaced about the periphery of said apparatus and adapted to introduce said air inwardly toward the axial flow component.

18. Apparatus in accordance with claim 16 wherein the means for introducing the radial flow component includes means for introducing the air from a peripheral location in a swirling manner inwardly toward the axial flow component.
19. Apparatus in accordance with claim 18 wherein the means for introducing the radial flow component includes two spaced means for introducing two streams of the at least one stream of the first volume of air from a peripheral location in a swirling manner and in opposite directions.
20. Apparatus in accordance with at least one of claims 15 to 19 wherein the means for introducing the second volume of air includes means for reducing the peripheral dimensions of the fuel-rich combustion means and introducing said second volume of air at at least one of (a) a point immediately preceeding said means for reducing the peripheral dimensions of said fuel rich combustion means; and (b) a point within the means for reducing the peripheral dimensions of the fuel-rich combustion means itself.



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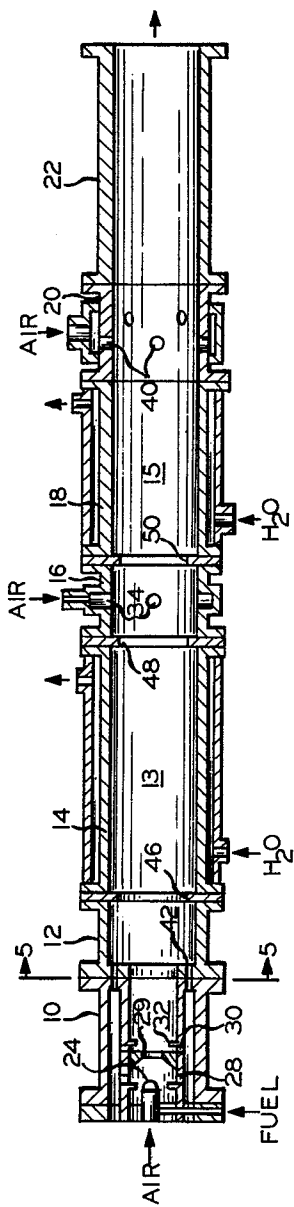


FIG. 1

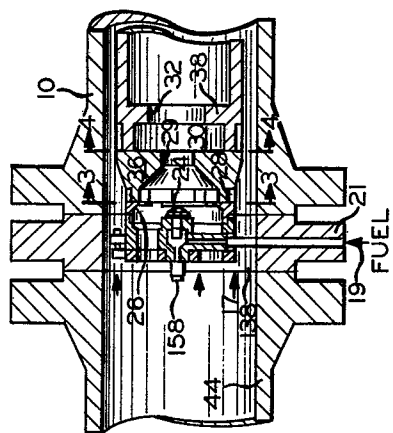


FIG. 2

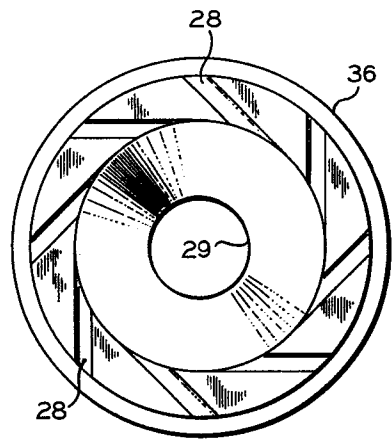


FIG. 3

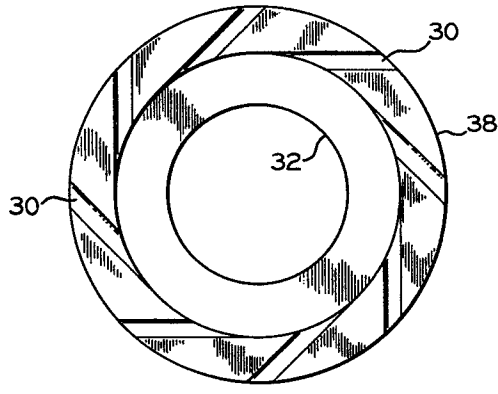


FIG. 4

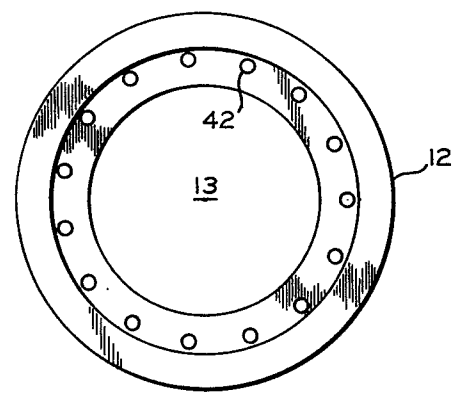
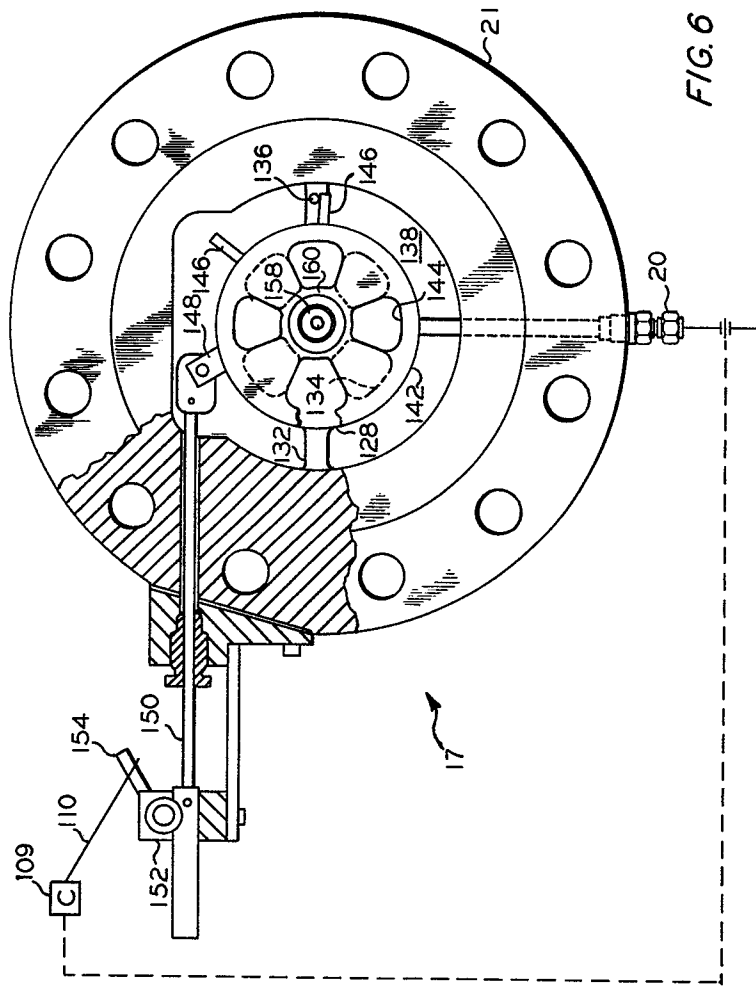


FIG. 5





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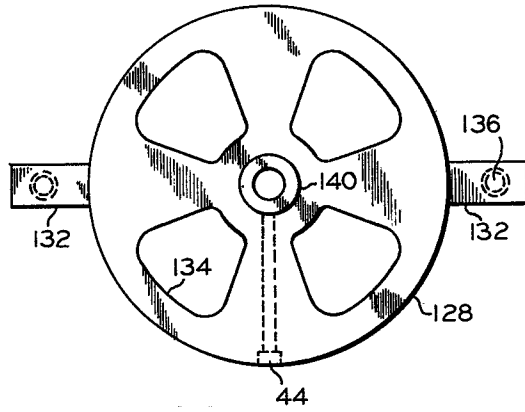


FIG. 8

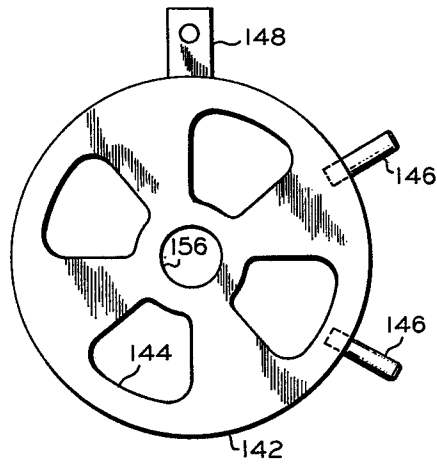


FIG. 7

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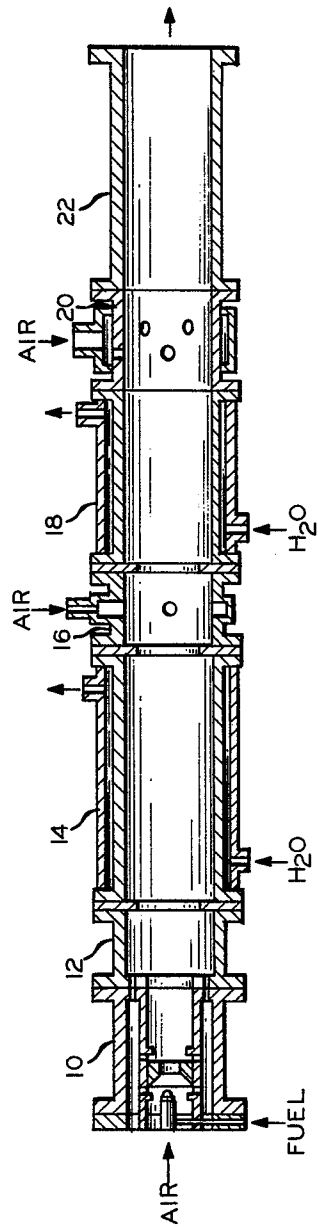
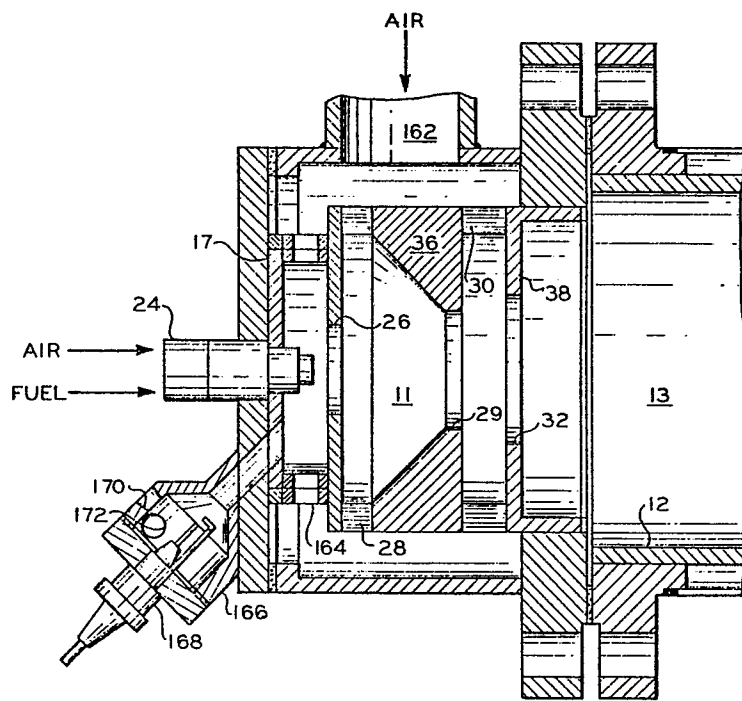


FIG. 9

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COMBUSTOR G

FIG. 10

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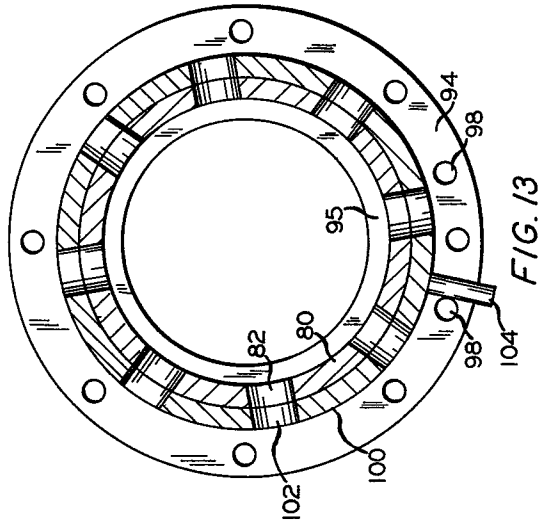


FIG. 13

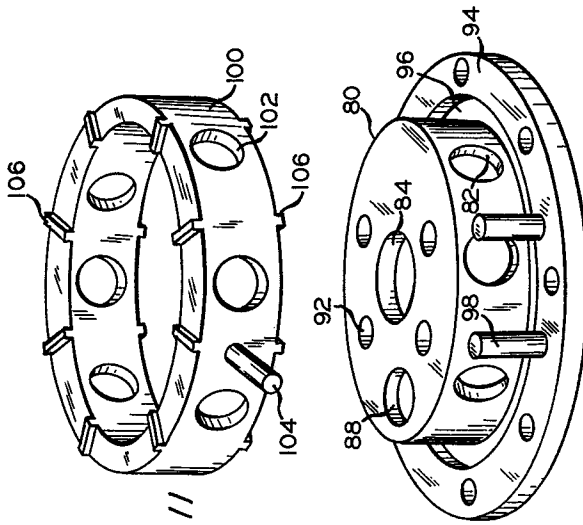
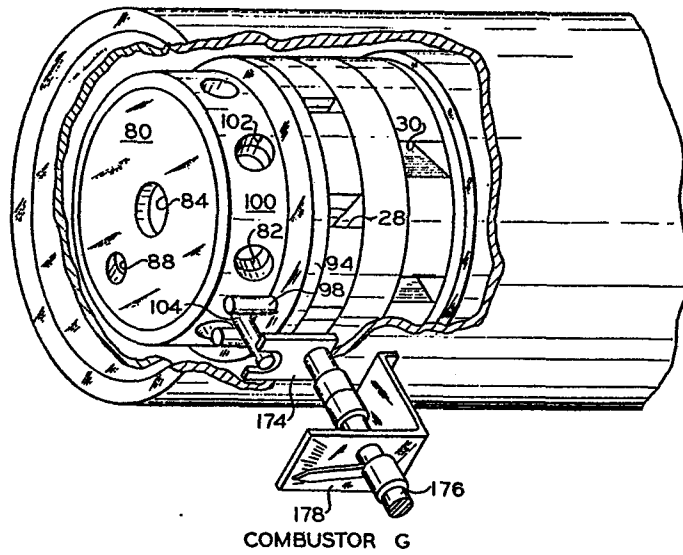


FIG. 11

FIG. 12

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COMBUSTOR G

FIG. 14

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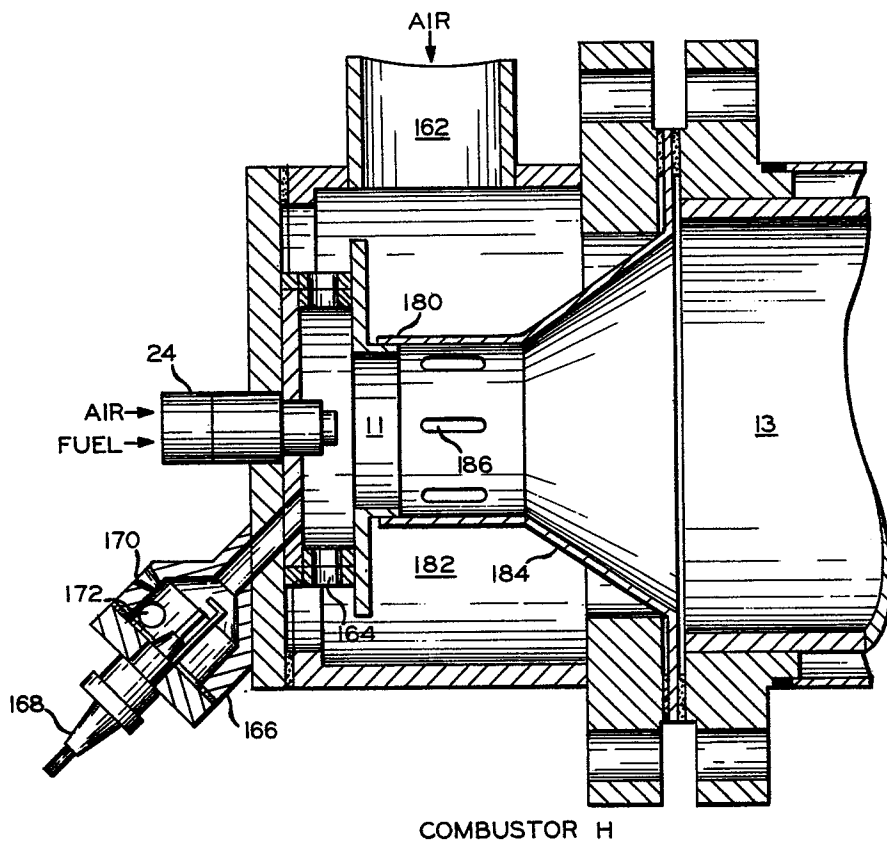


FIG. 15

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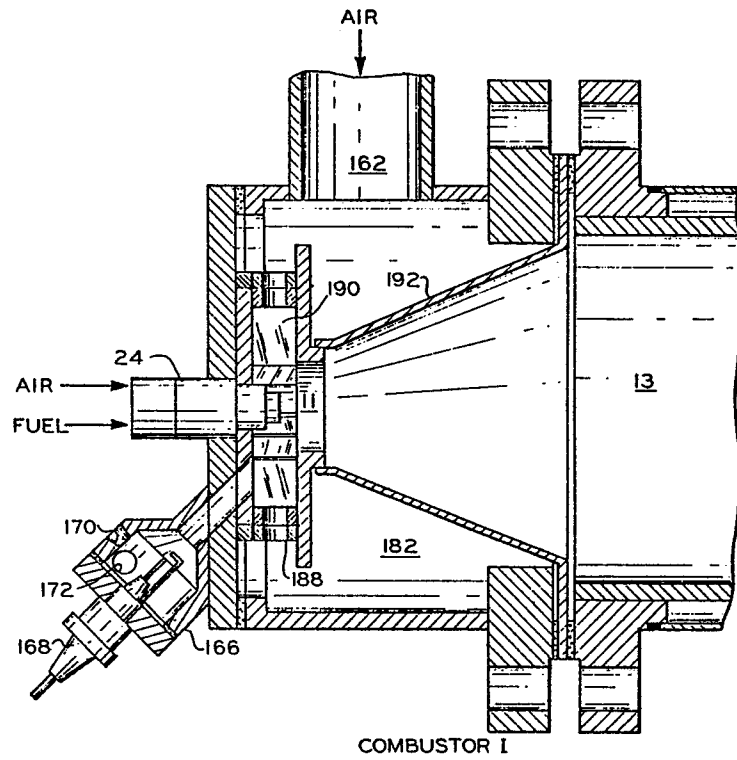
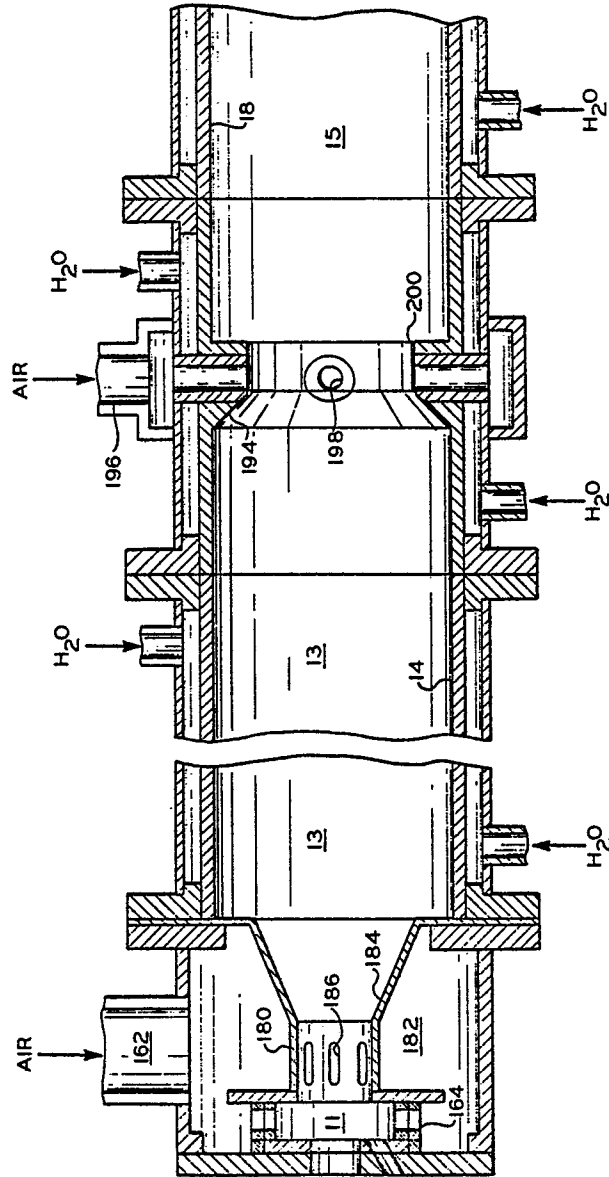


FIG. 16

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COMBUSTOR H

FIG. 17



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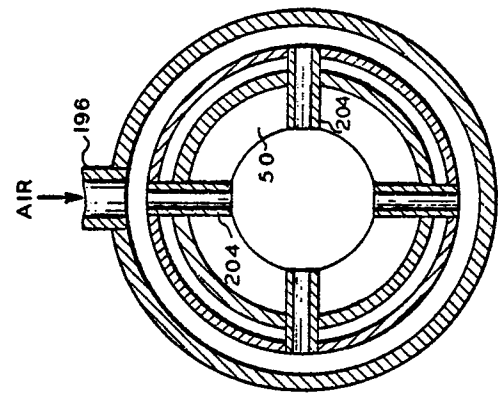


FIG. 19

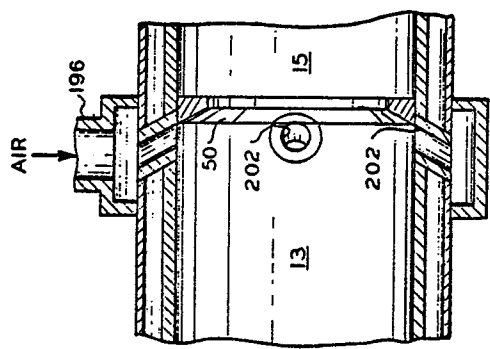


FIG. 18

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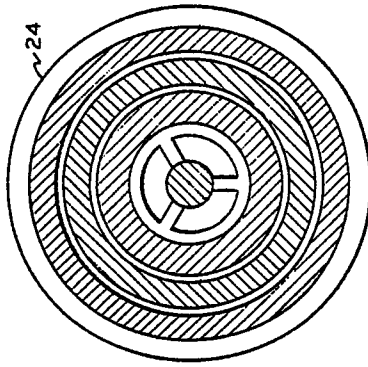


FIG. 22

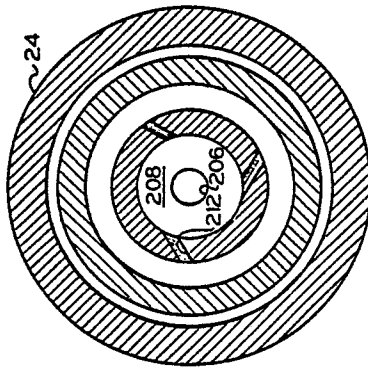


FIG. 21

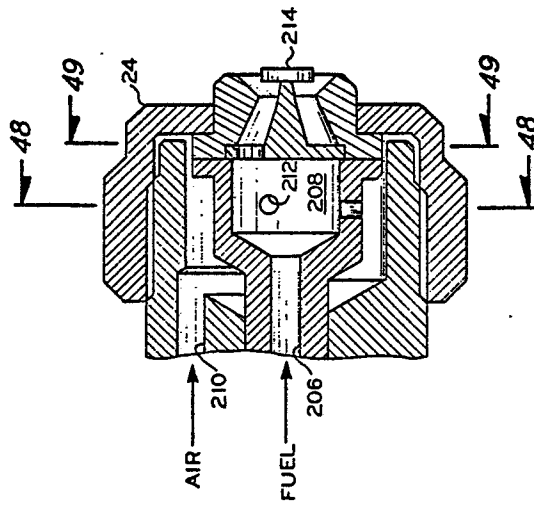


FIG. 20

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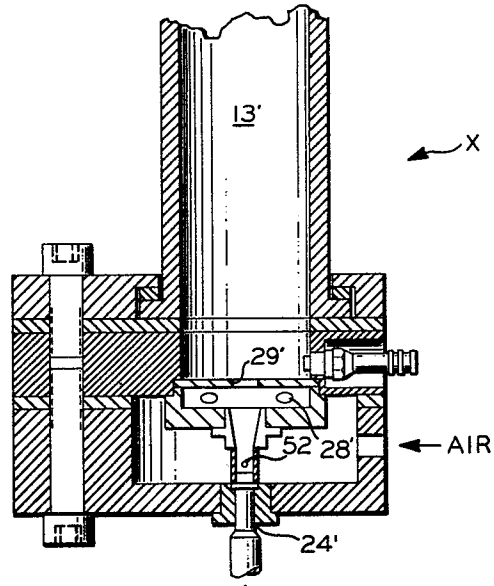


FIG. 24

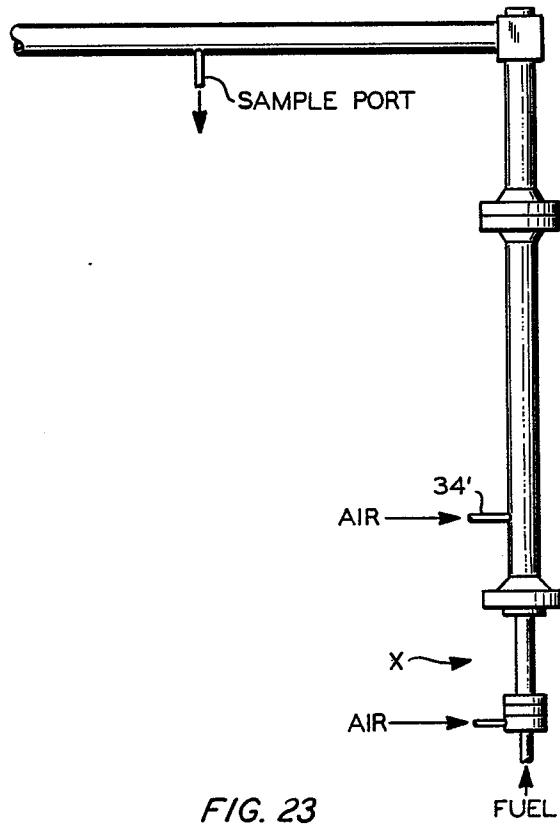
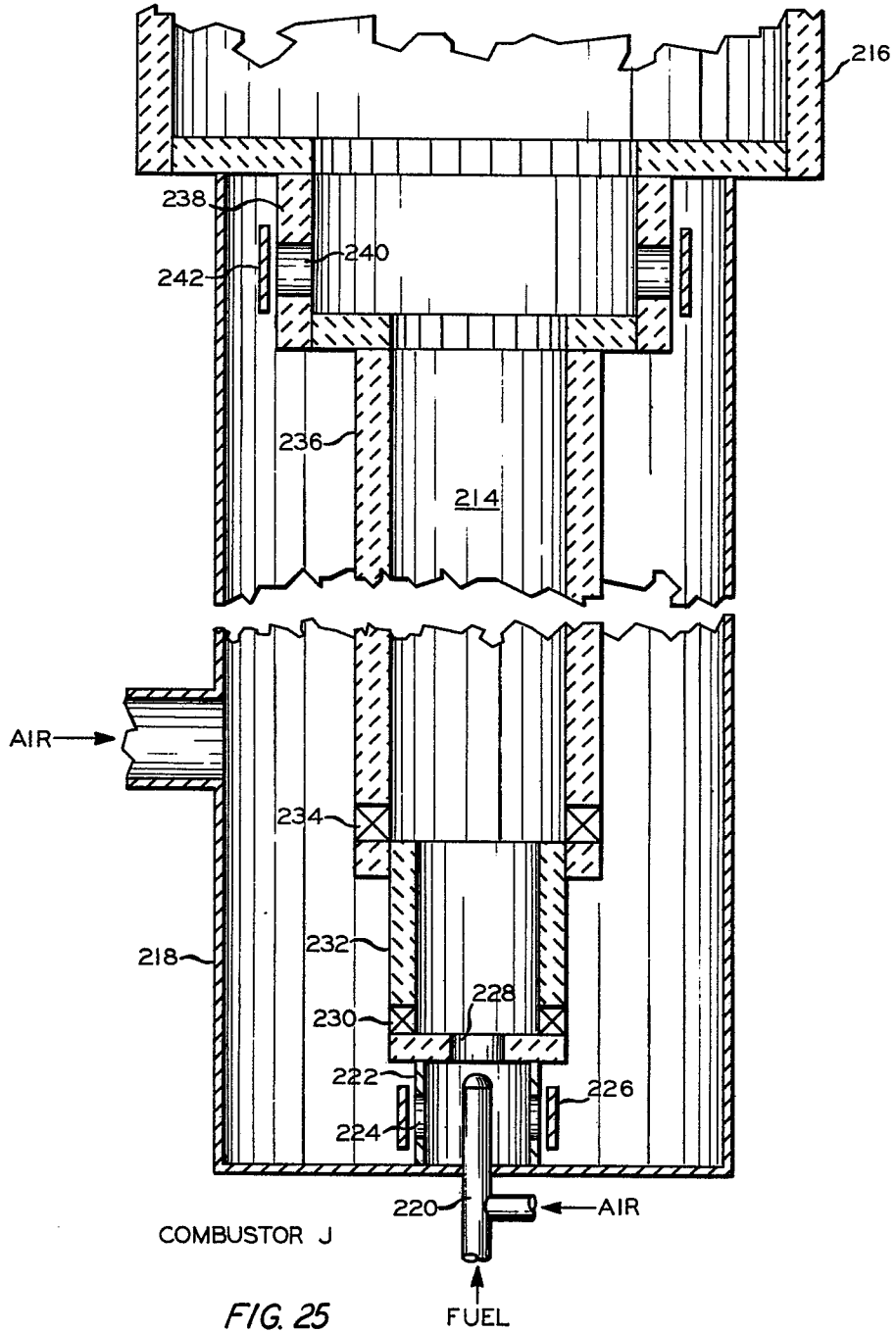


FIG. 23

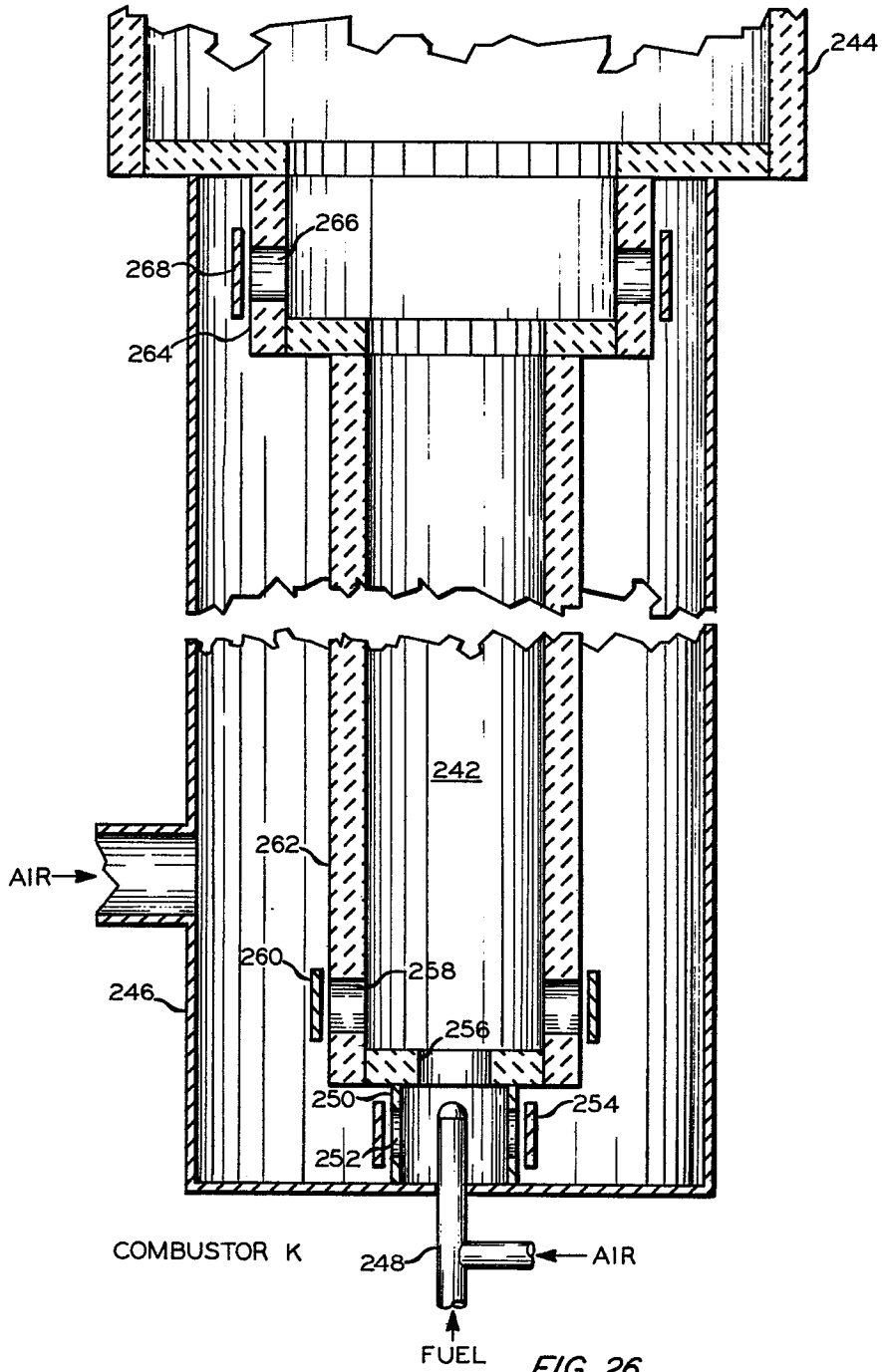
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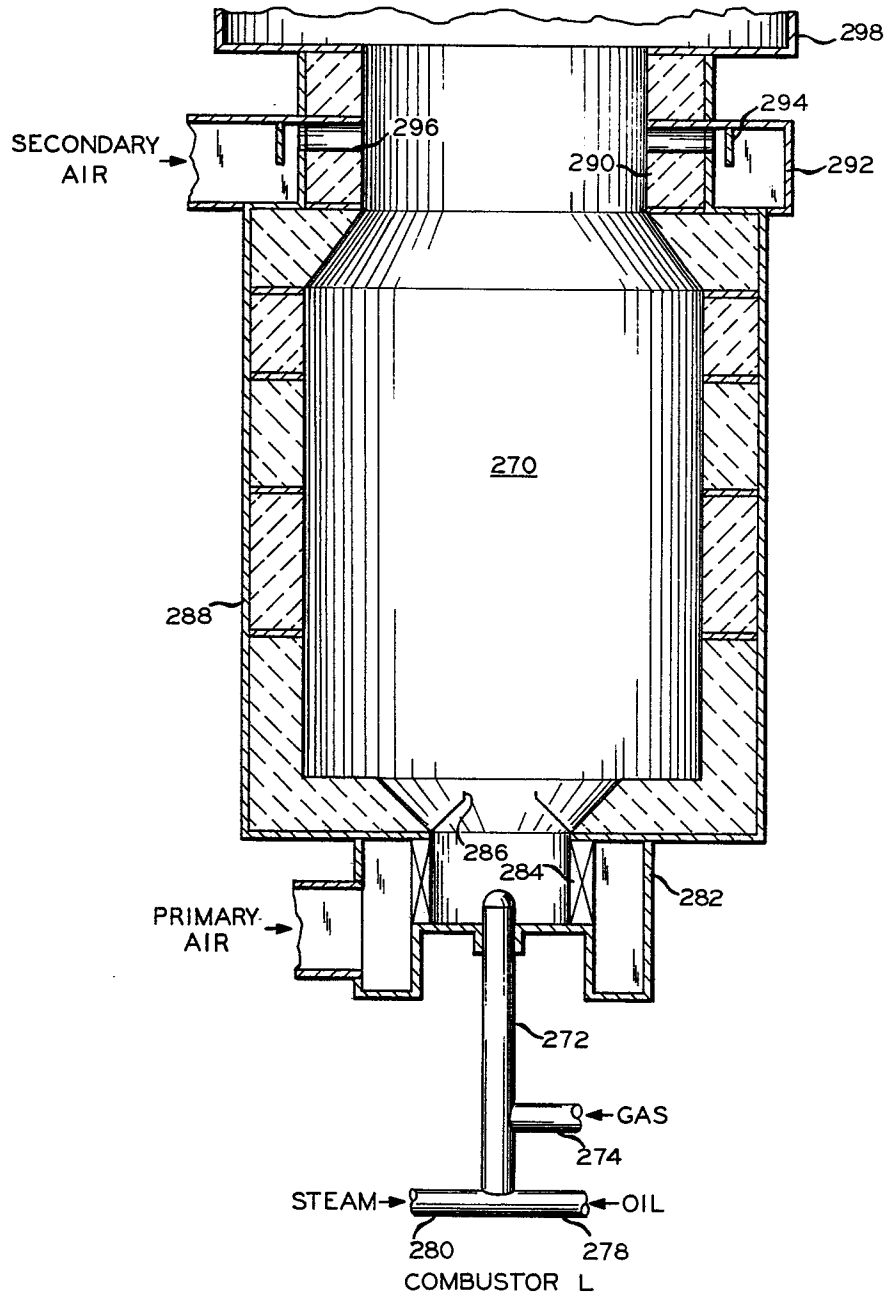
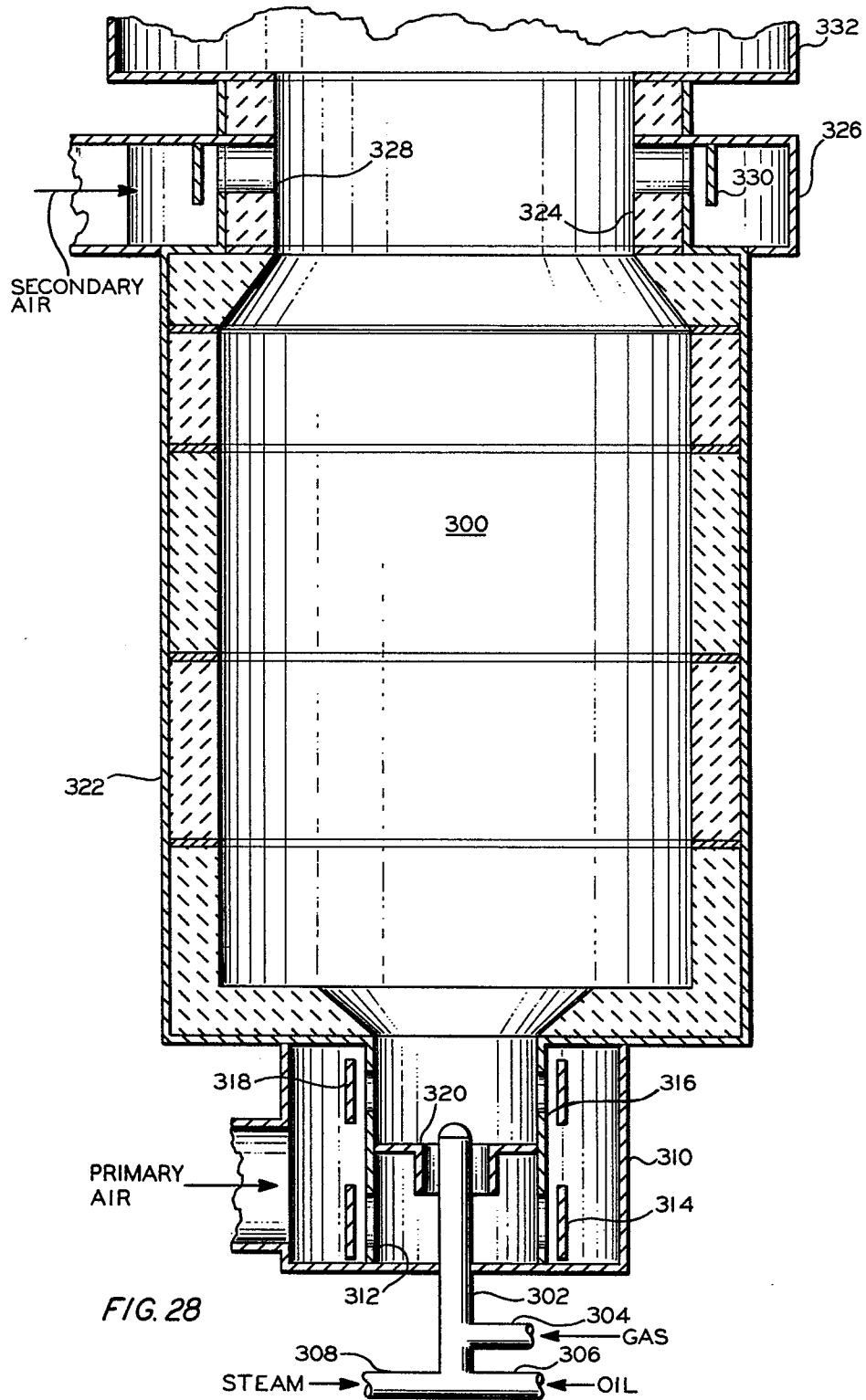


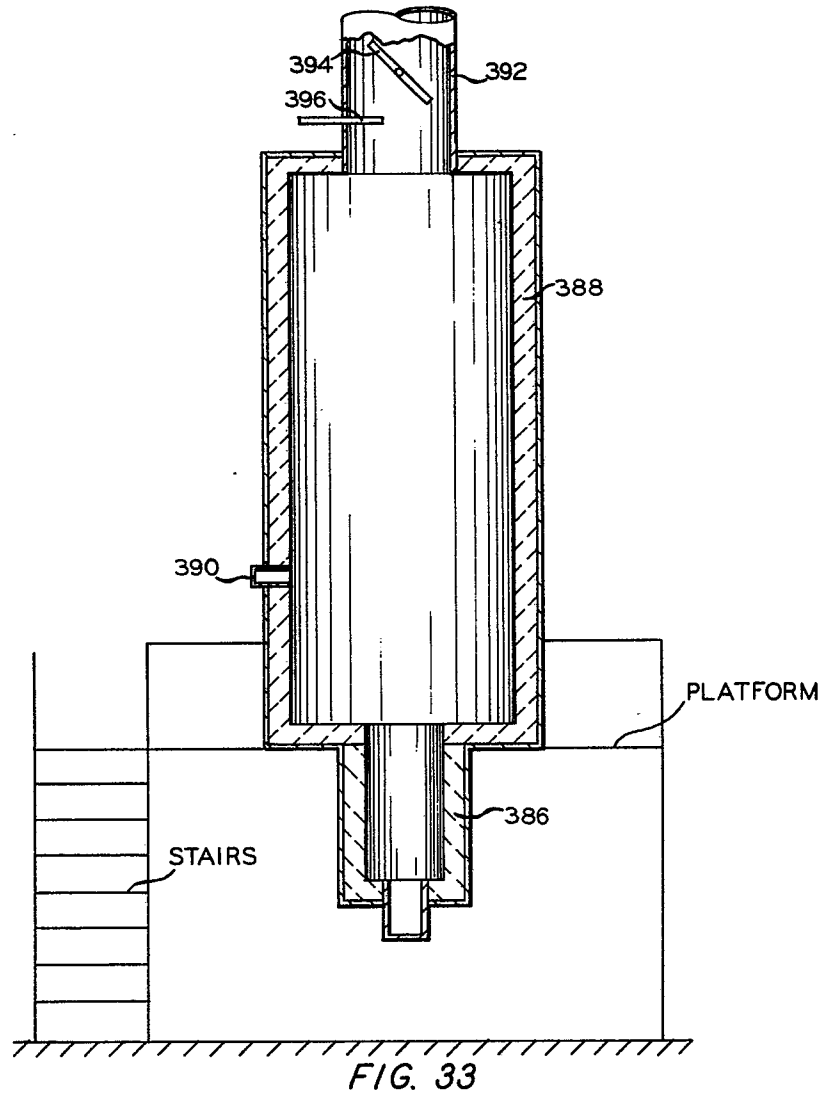
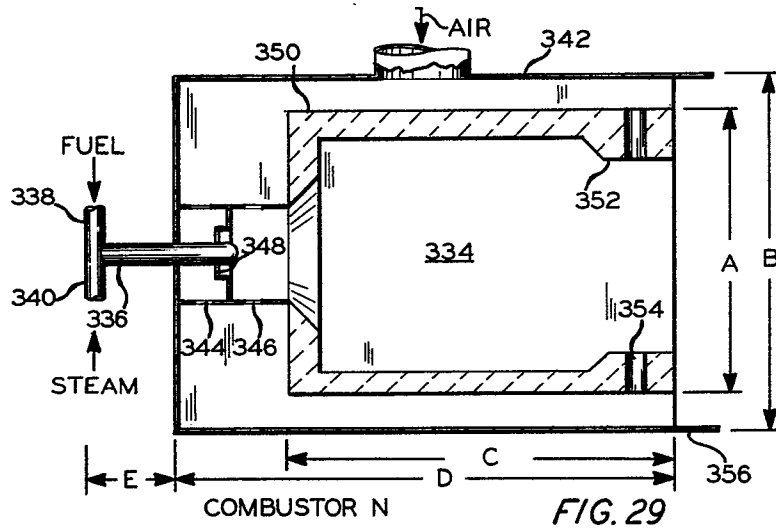
FIG. 27

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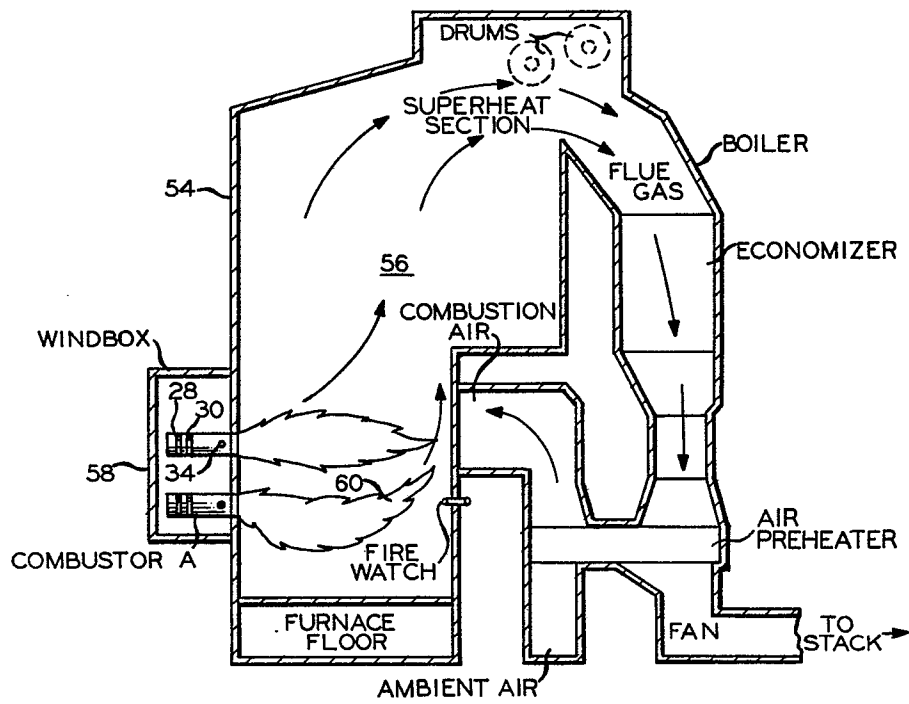


FIG. 30

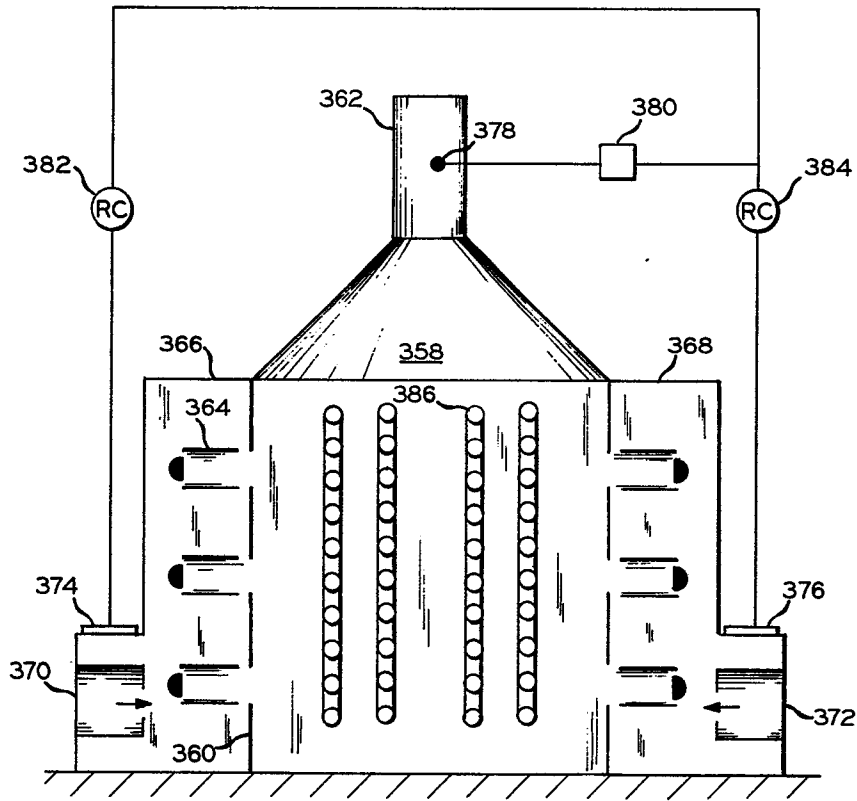


FIG. 31

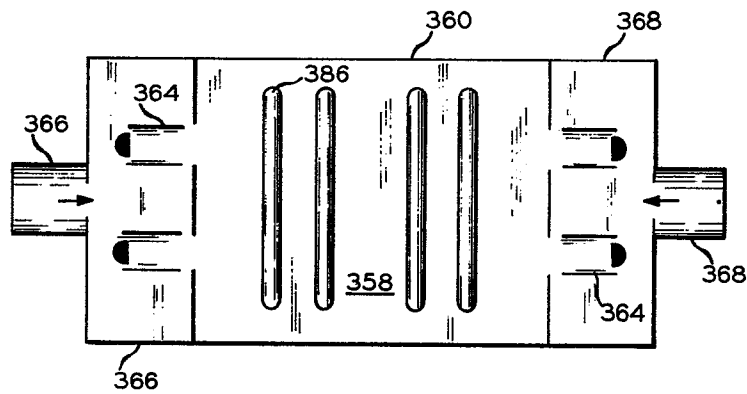


FIG. 32

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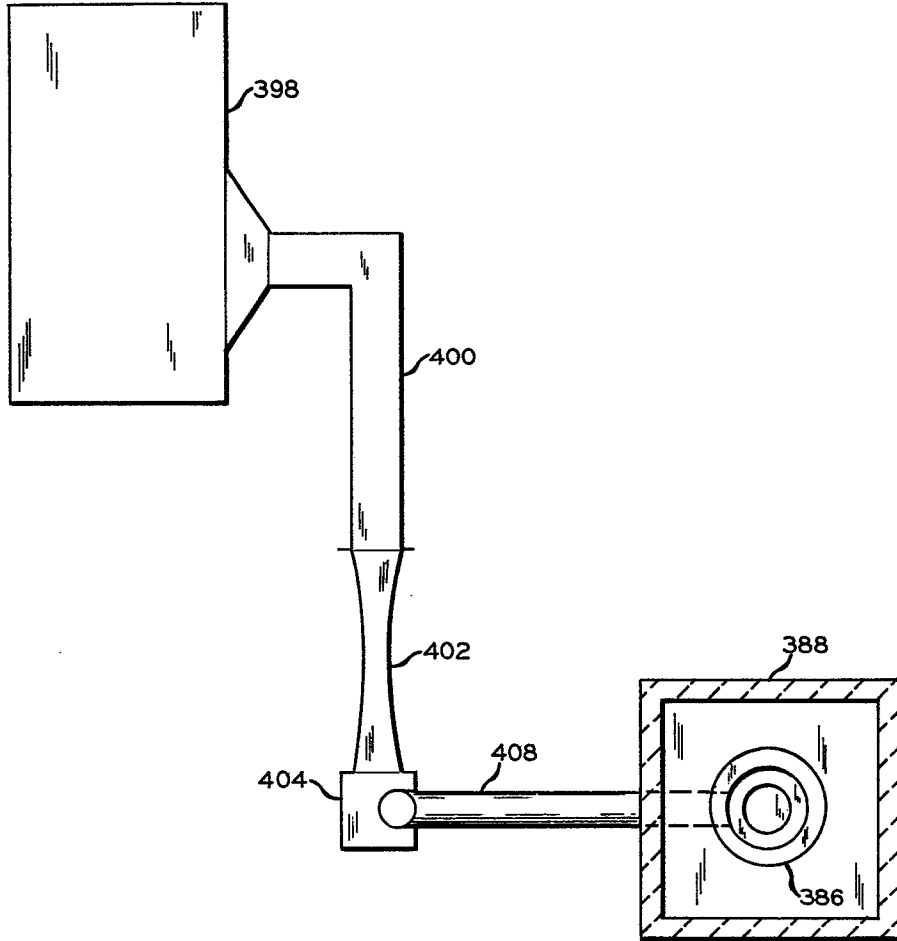


FIG. 34

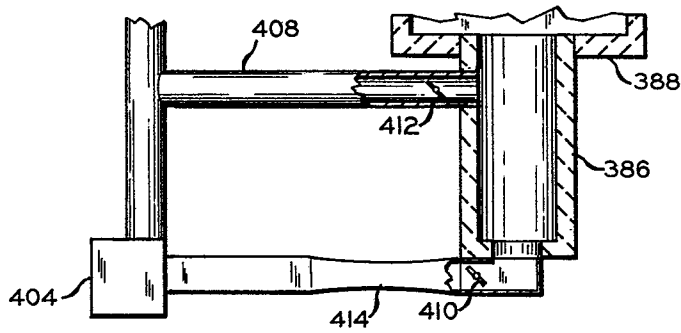


FIG. 35

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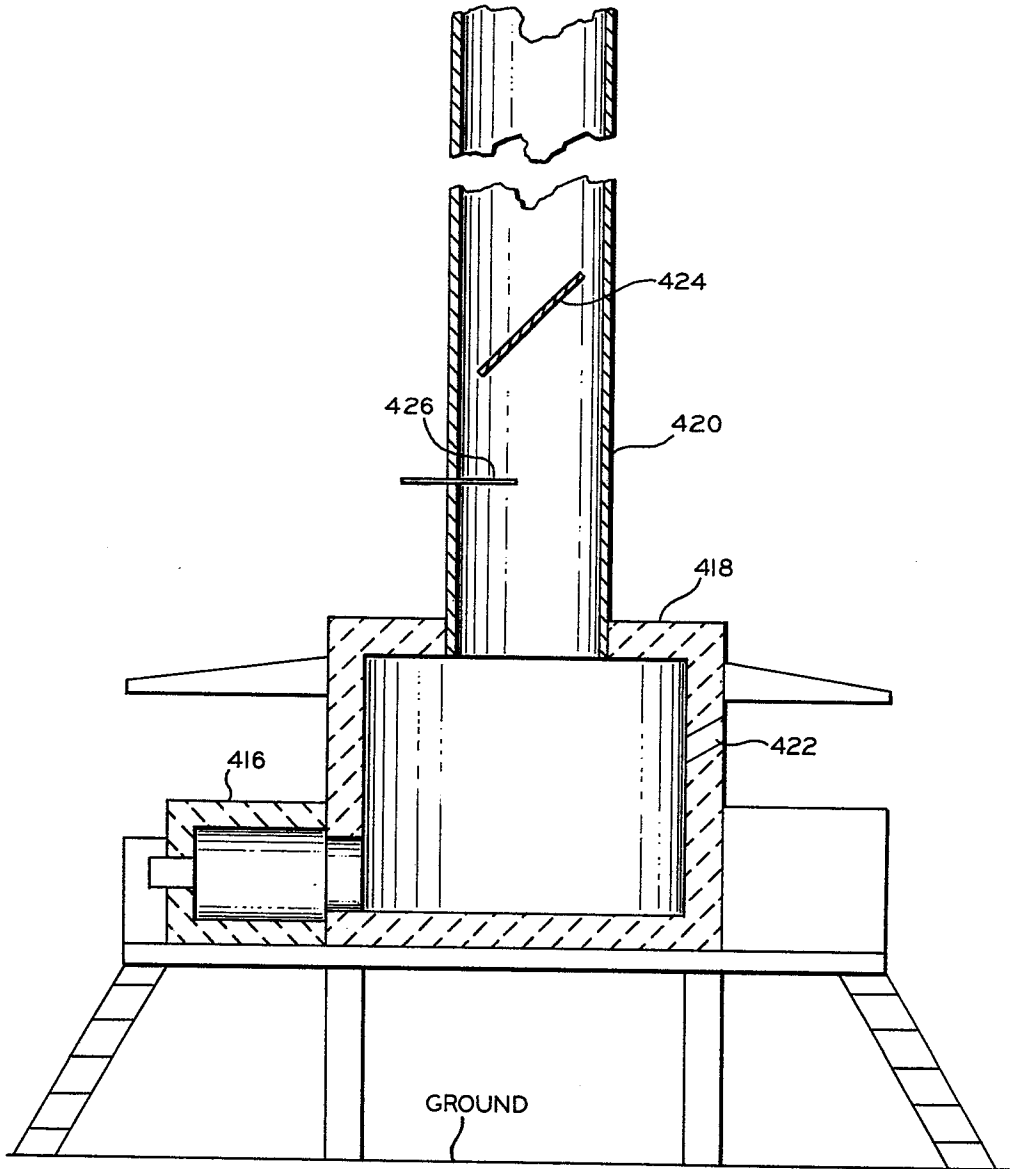


FIG. 36

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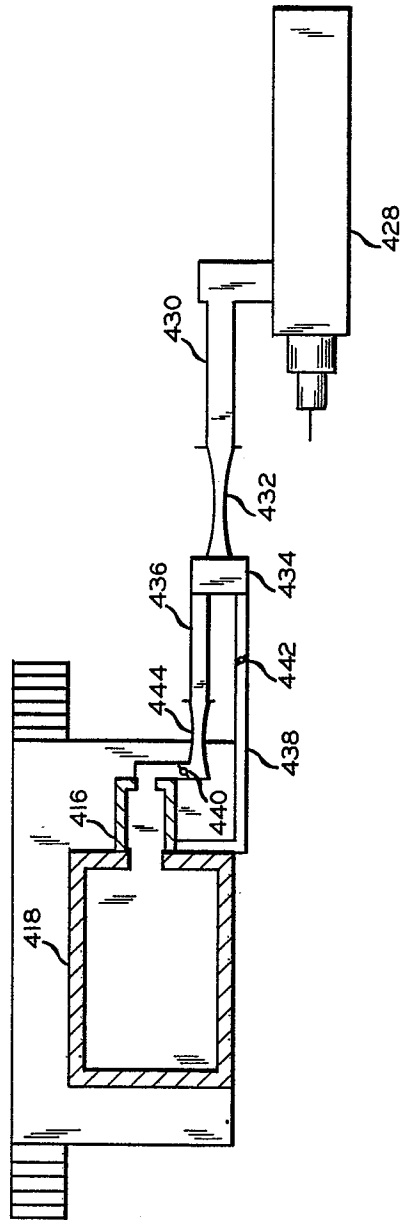


FIG. 37

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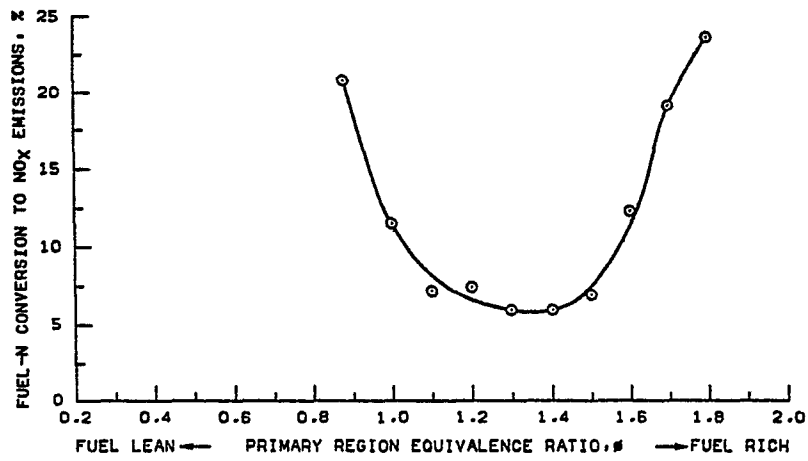


FIG. 38 (EX. I)

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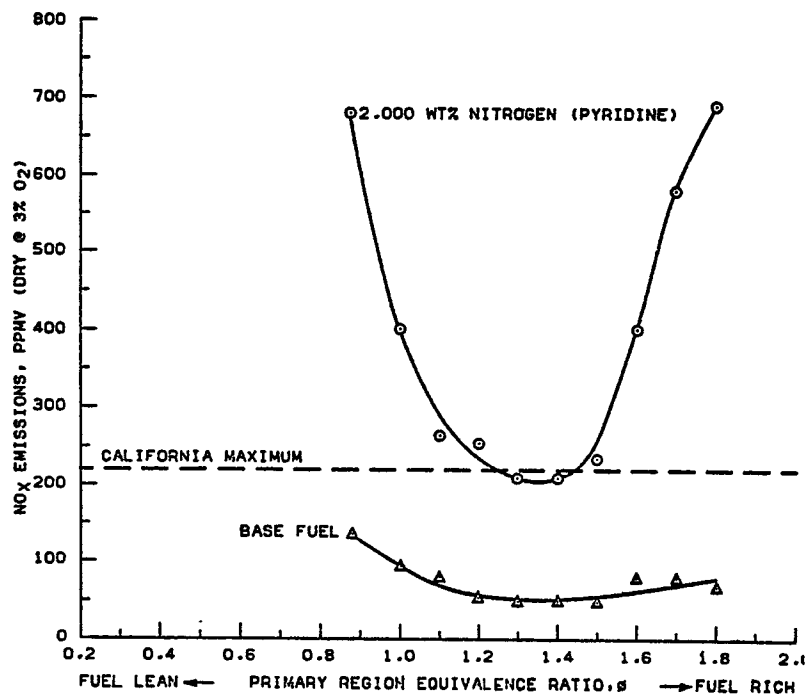


FIG. 39 (EX.I)

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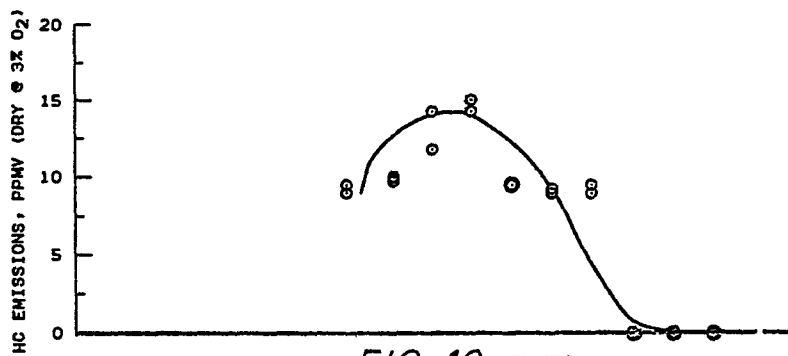


FIG. 40 (EX.I)

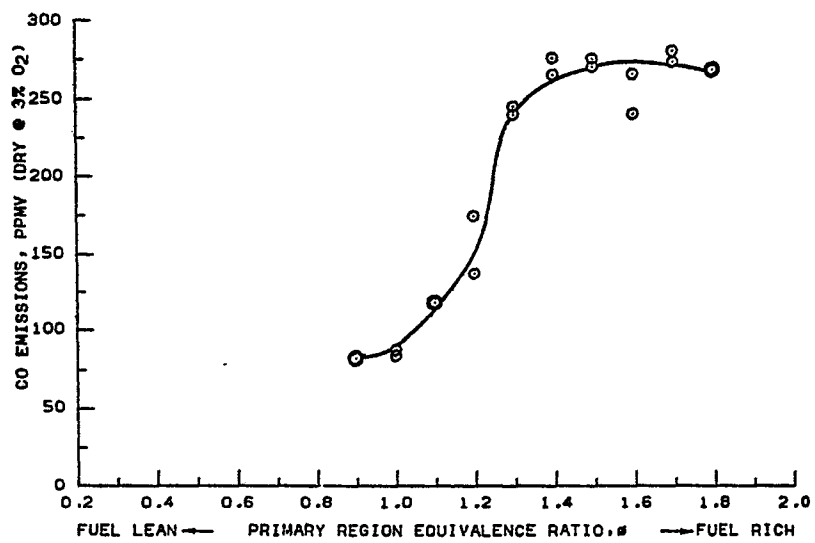


FIG. 41 (EX.I)



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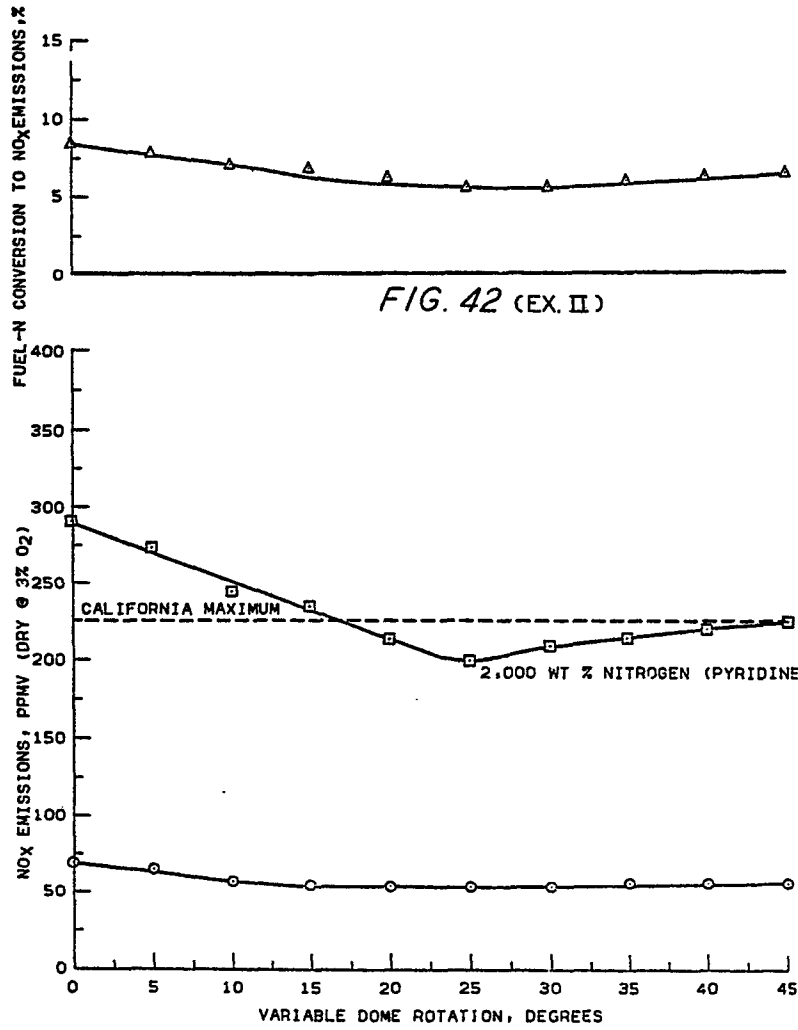


FIG. 42 (EX. II)

FIG. 43 (EX. II)

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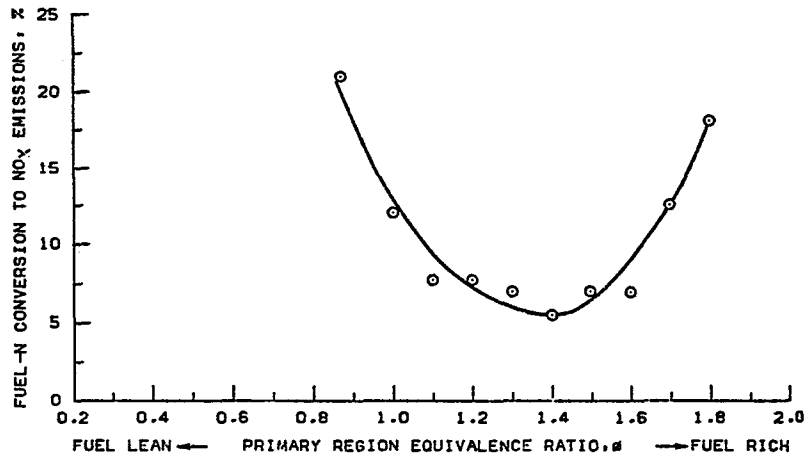
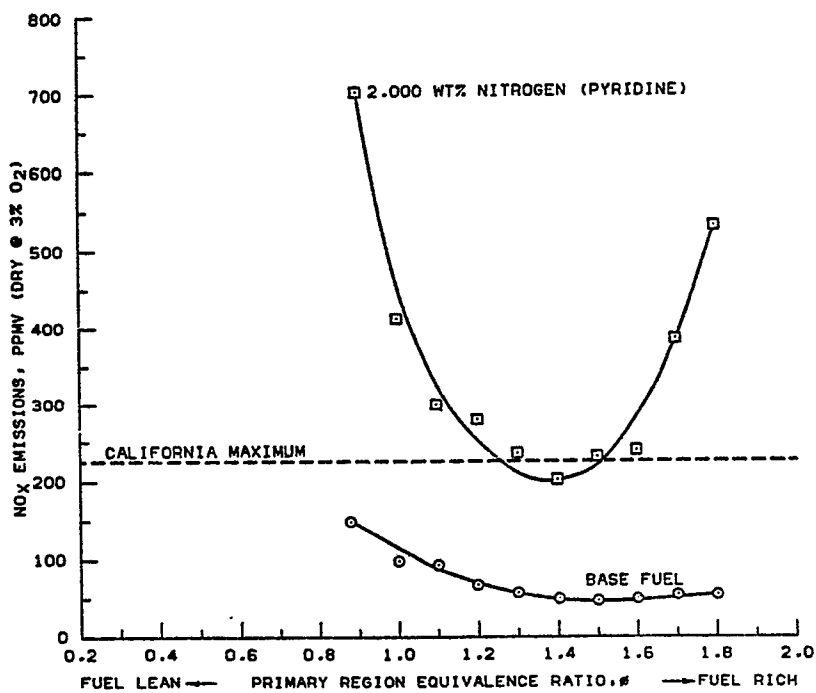


FIG. 44 (EX. II)

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(EX. II)  
FIG. 45

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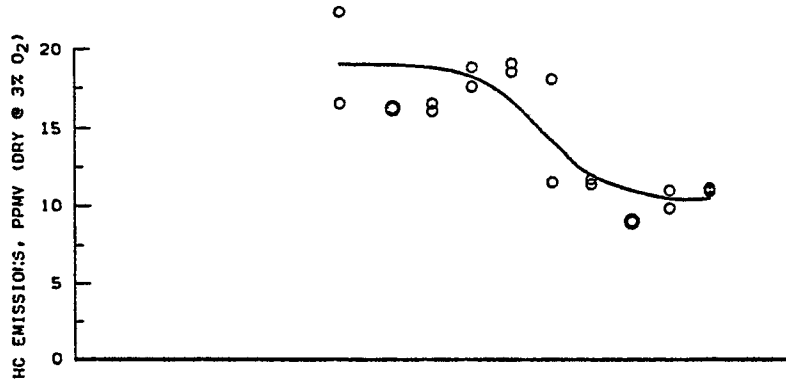
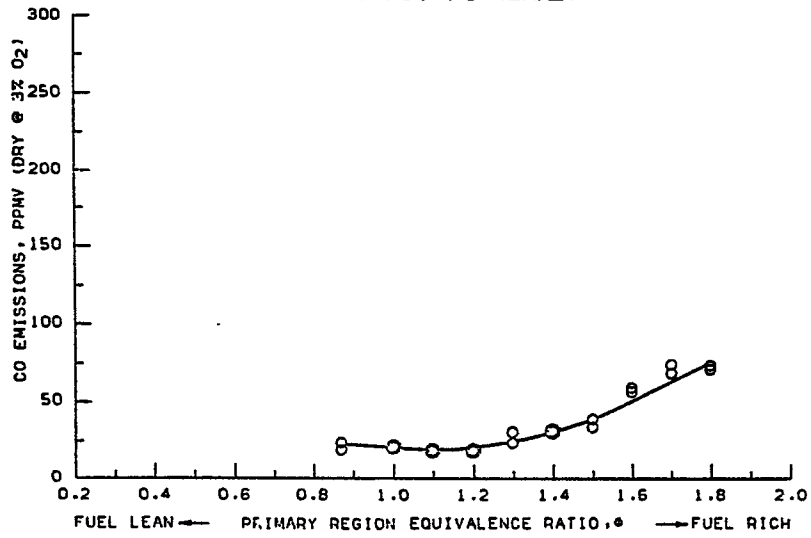


FIG. 46 (EX. II)



(EX. II)  
FIG. 47

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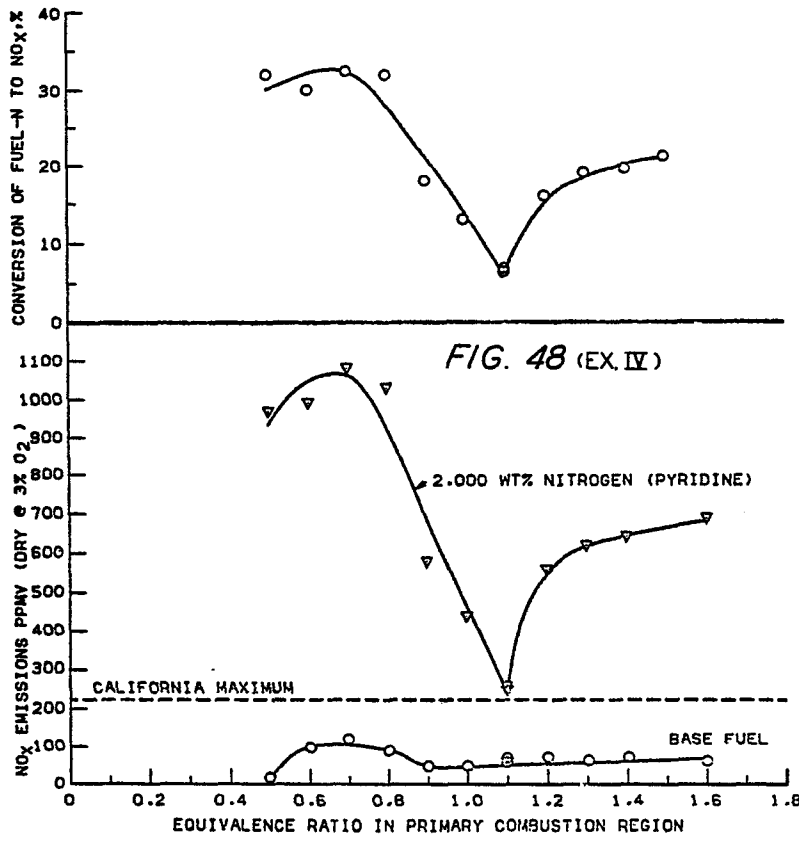


FIG. 49 (EX. IV)

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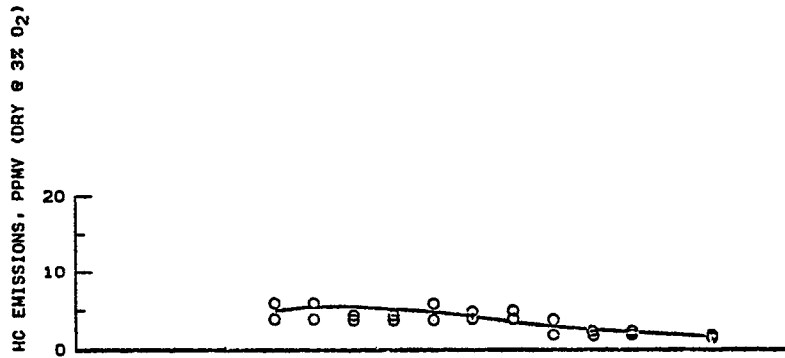


FIG. 50 (EX. IV)

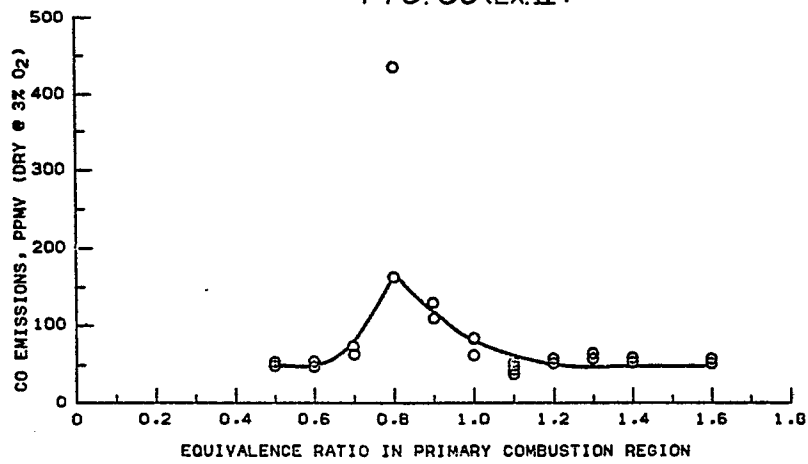


FIG. 51 (EX. IV)

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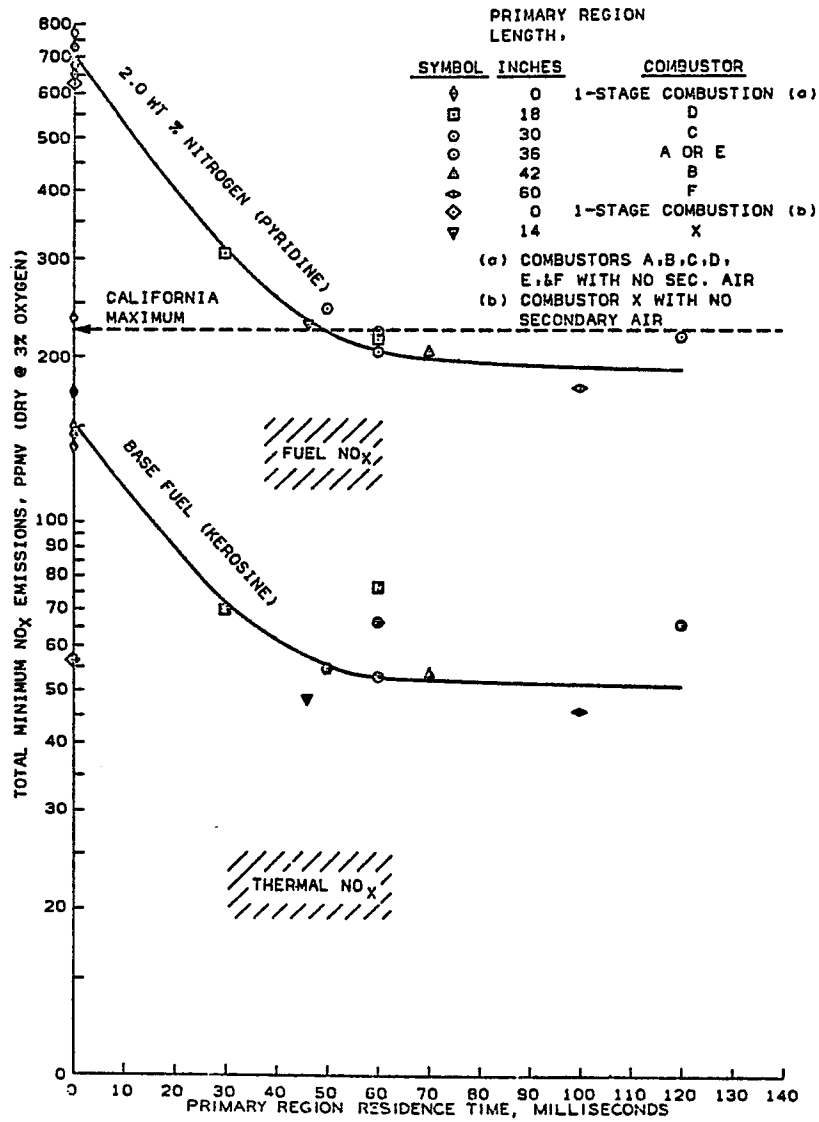


FIG. 52 (EX. V)

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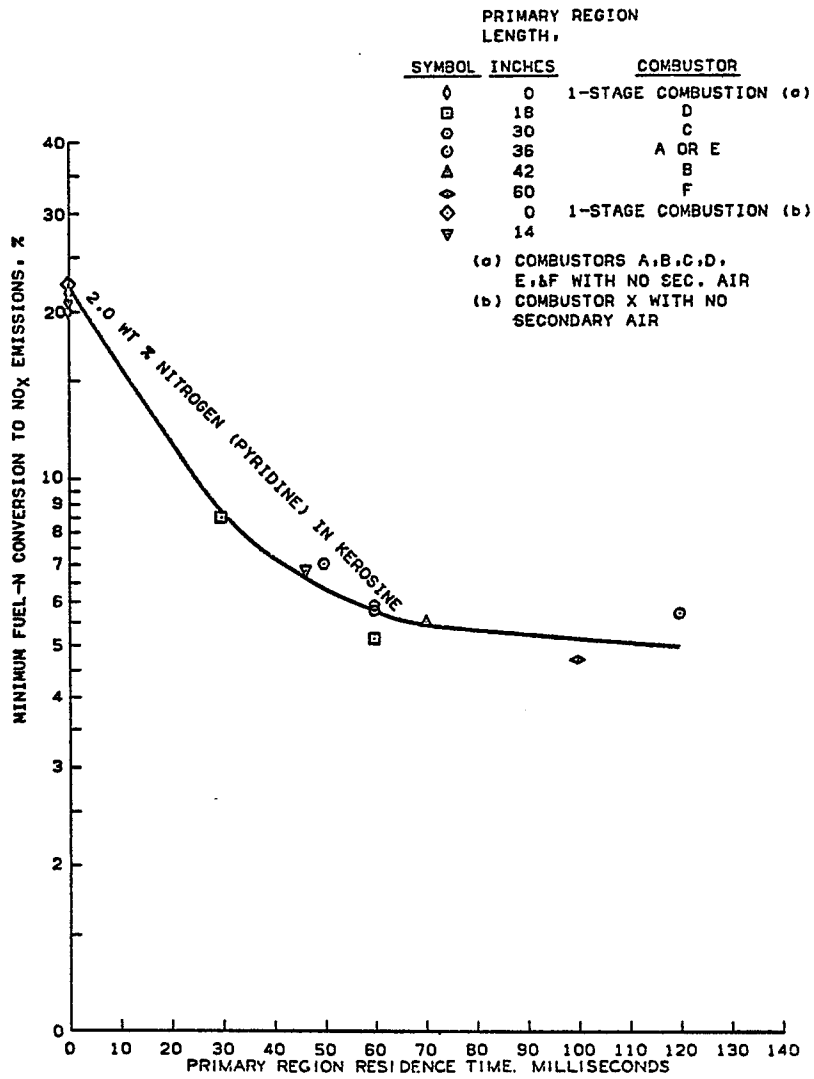


FIG. 53 (EX. V)



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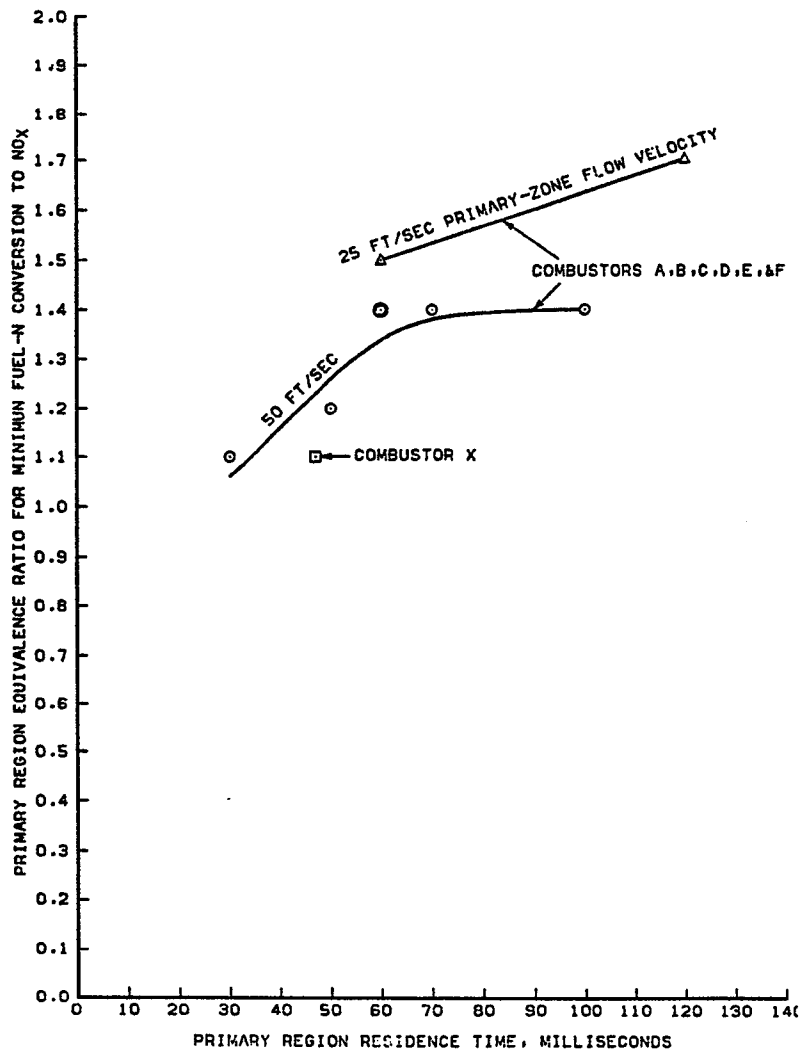


FIG. 54 (EX. V)

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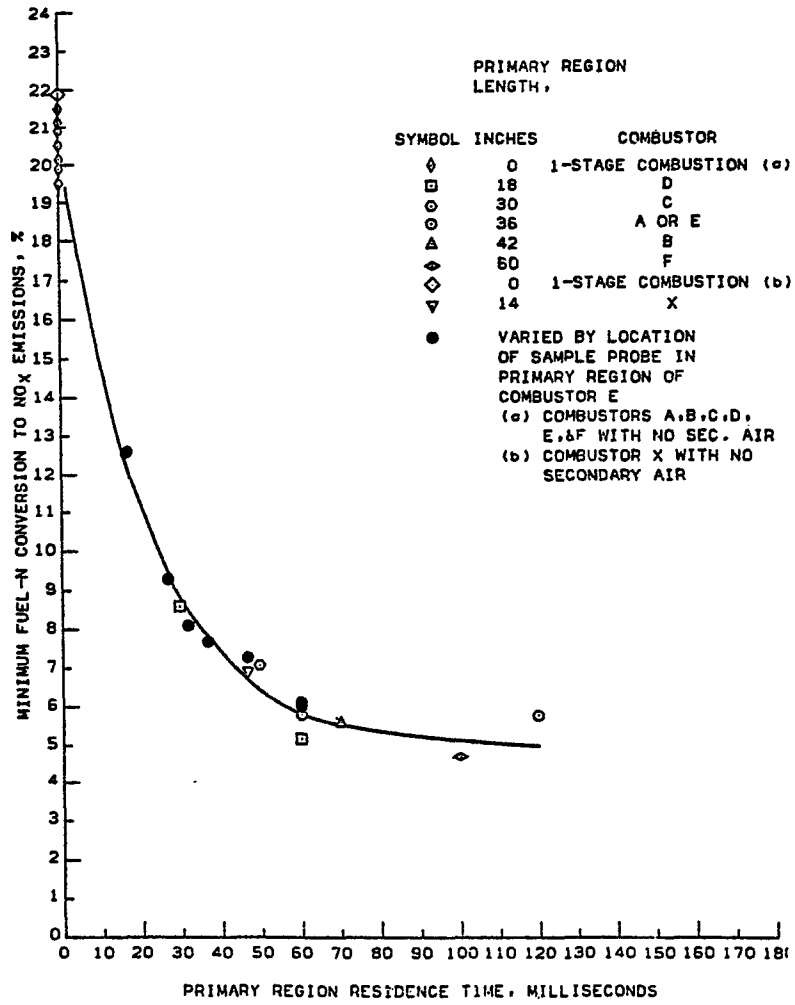


FIG. 55 (EX. V)

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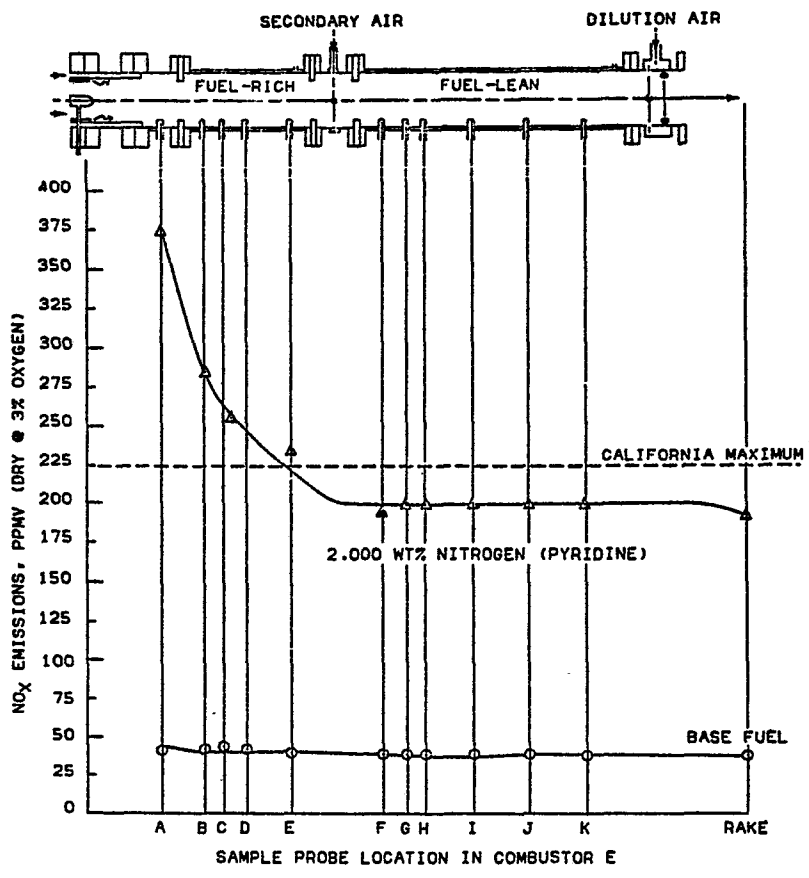


FIG. 56 (EX.V)

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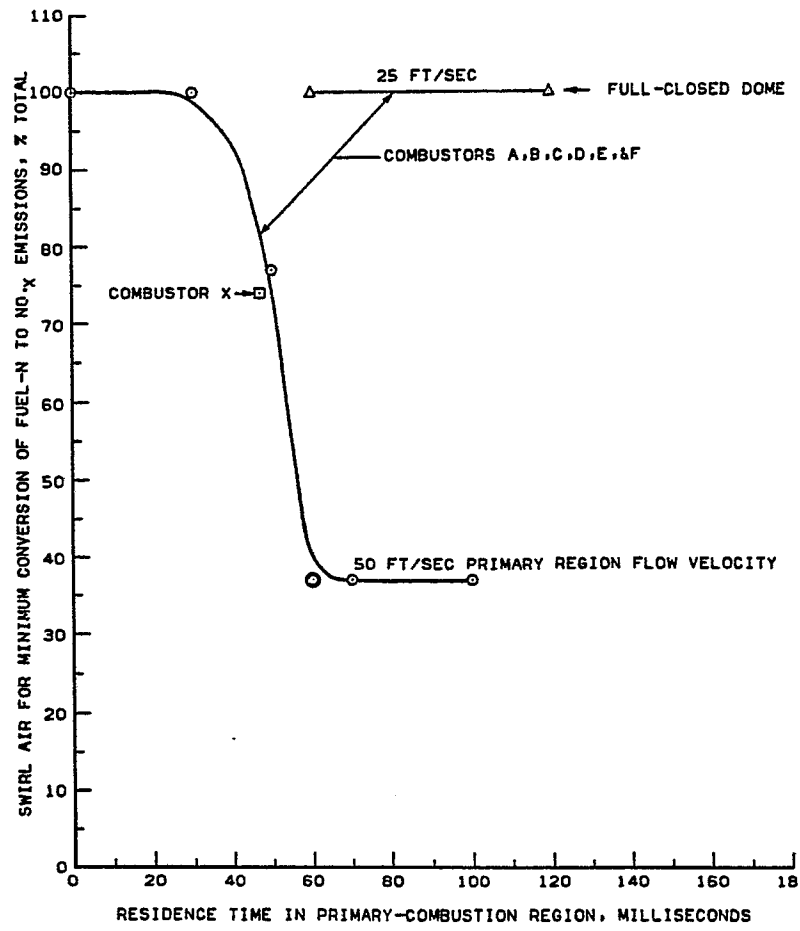


FIG. 57 (EX. VI)

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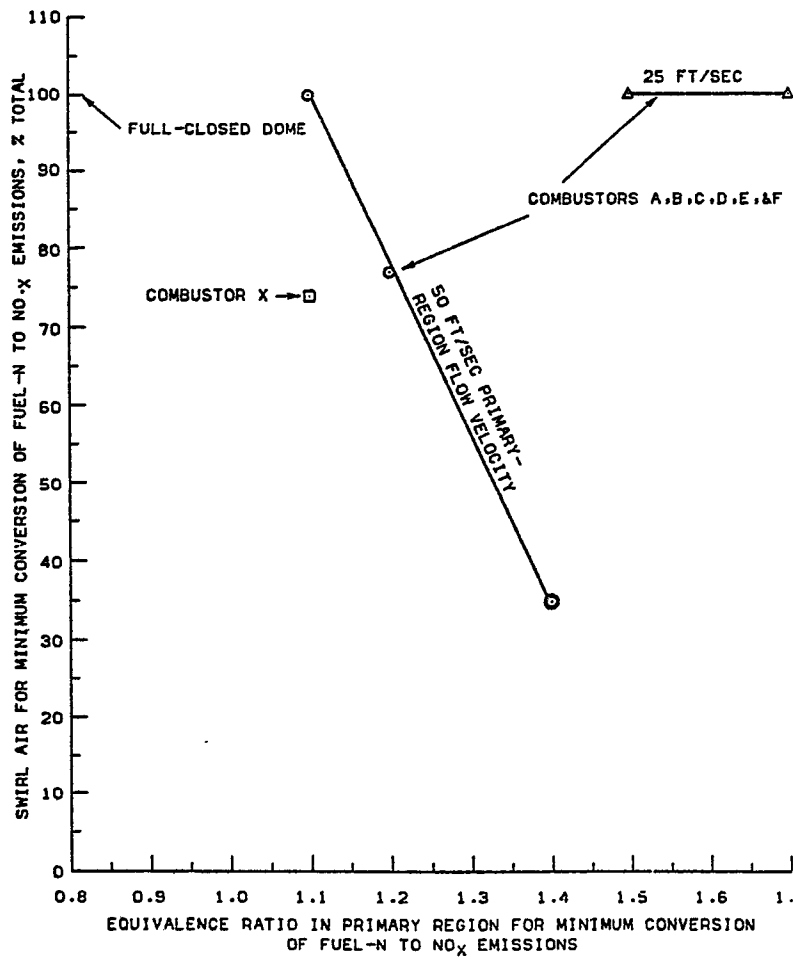


FIG. 58 (EX. VI)

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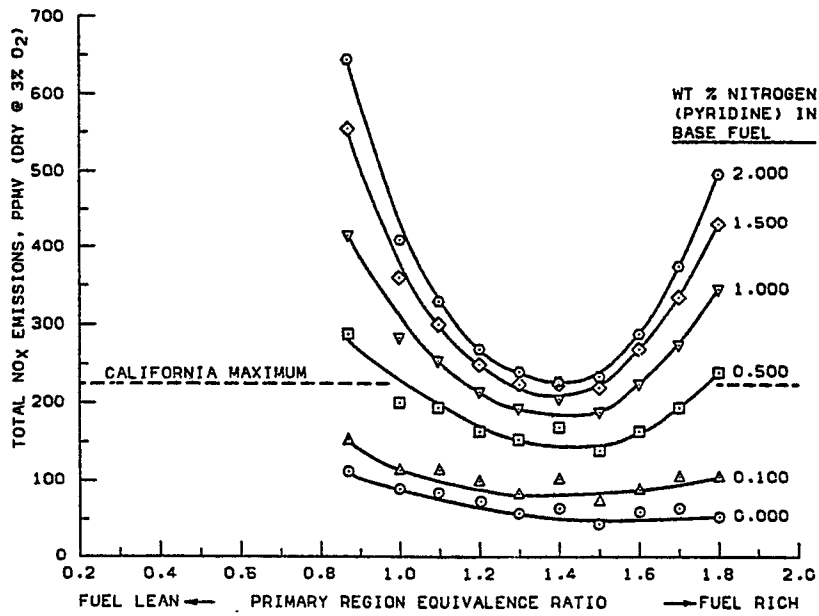


FIG. 59 (EX. VI)

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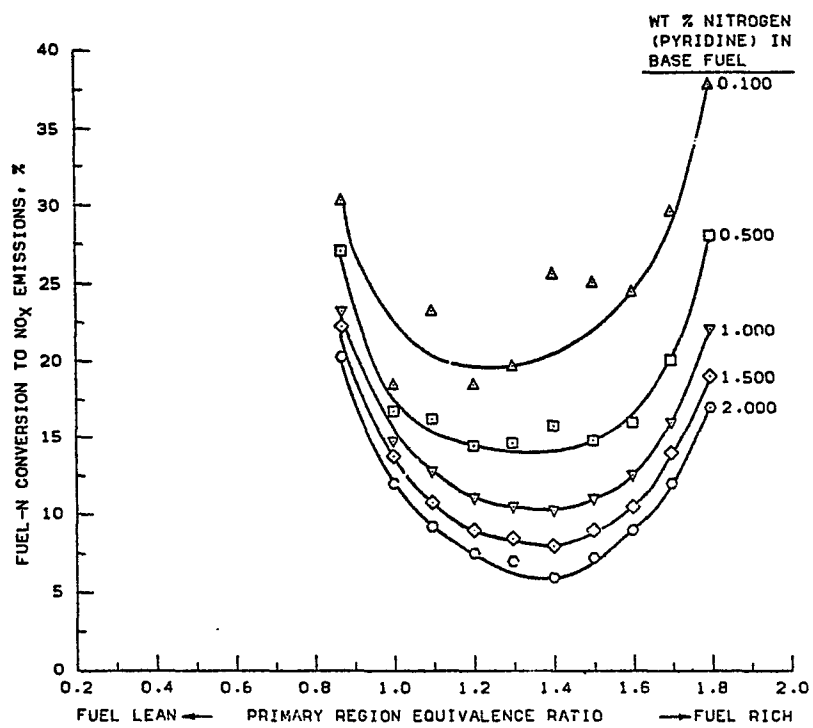


FIG. 60 (EX. VII)

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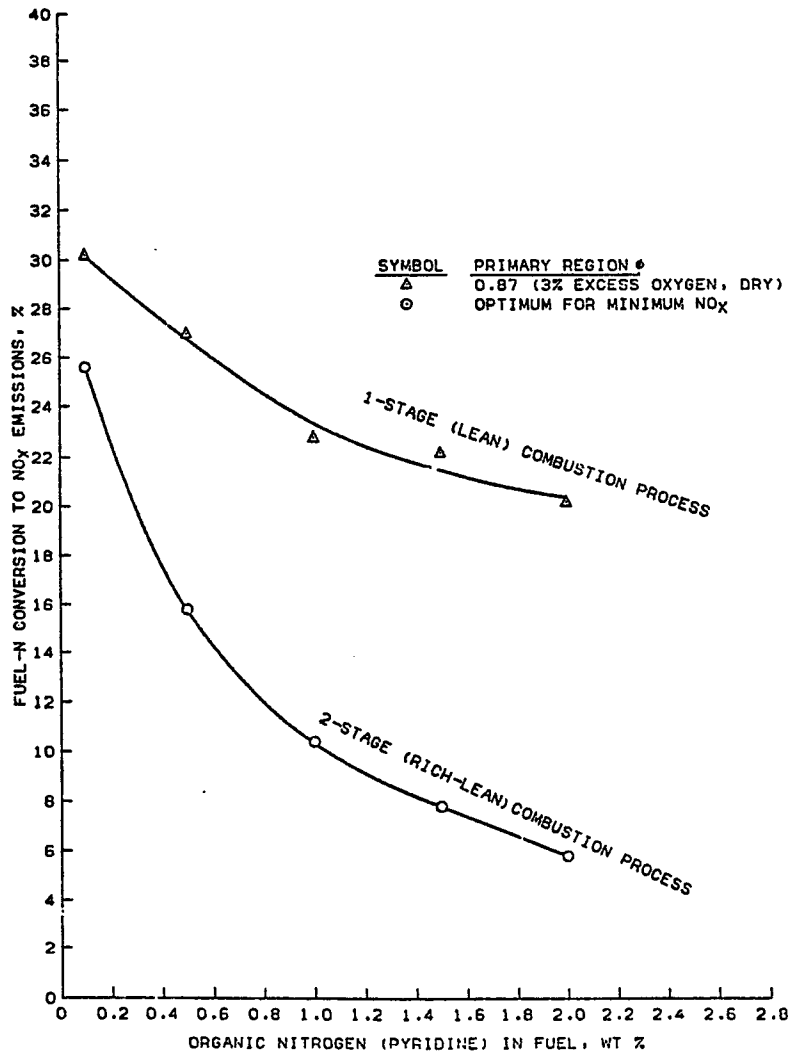


FIG. 6I (EX. VII)



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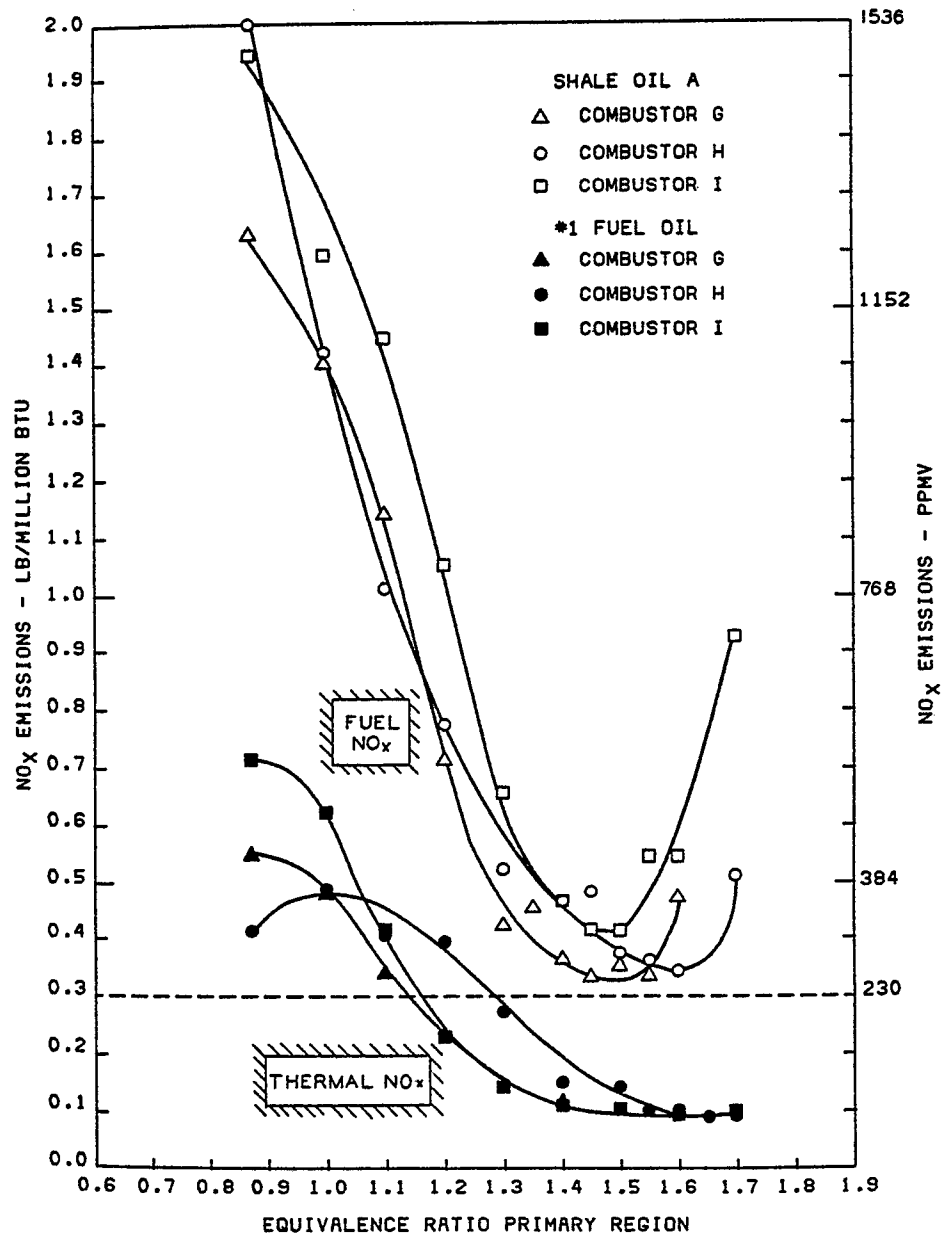


FIG. 62

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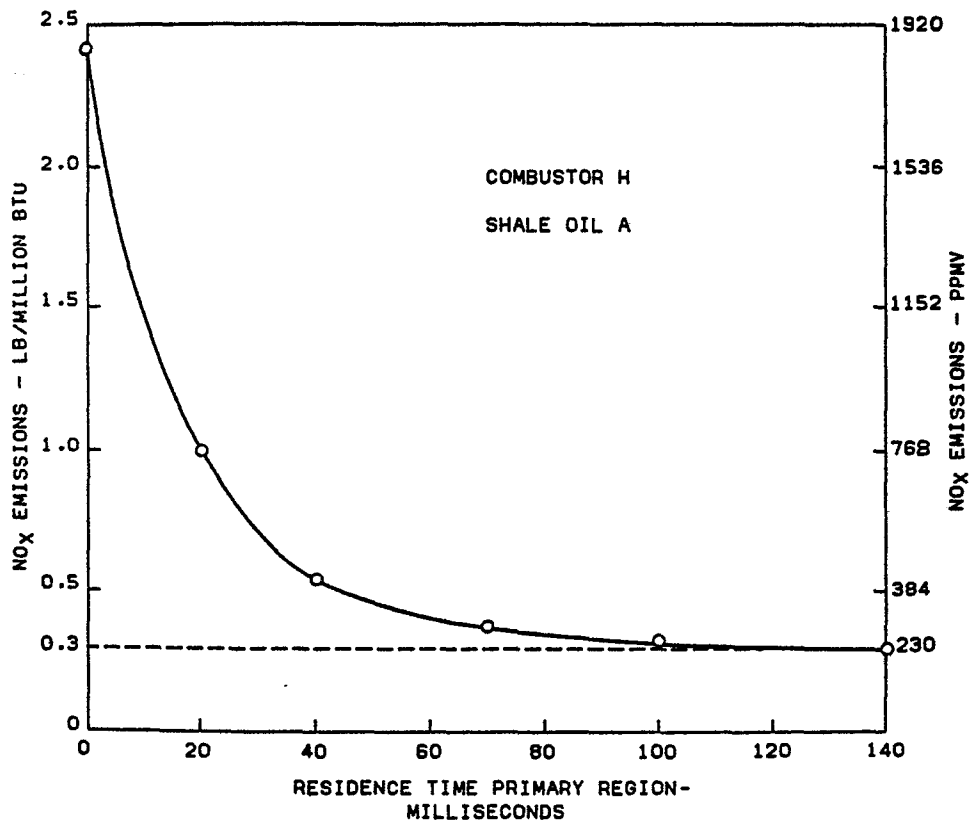


FIG. 63

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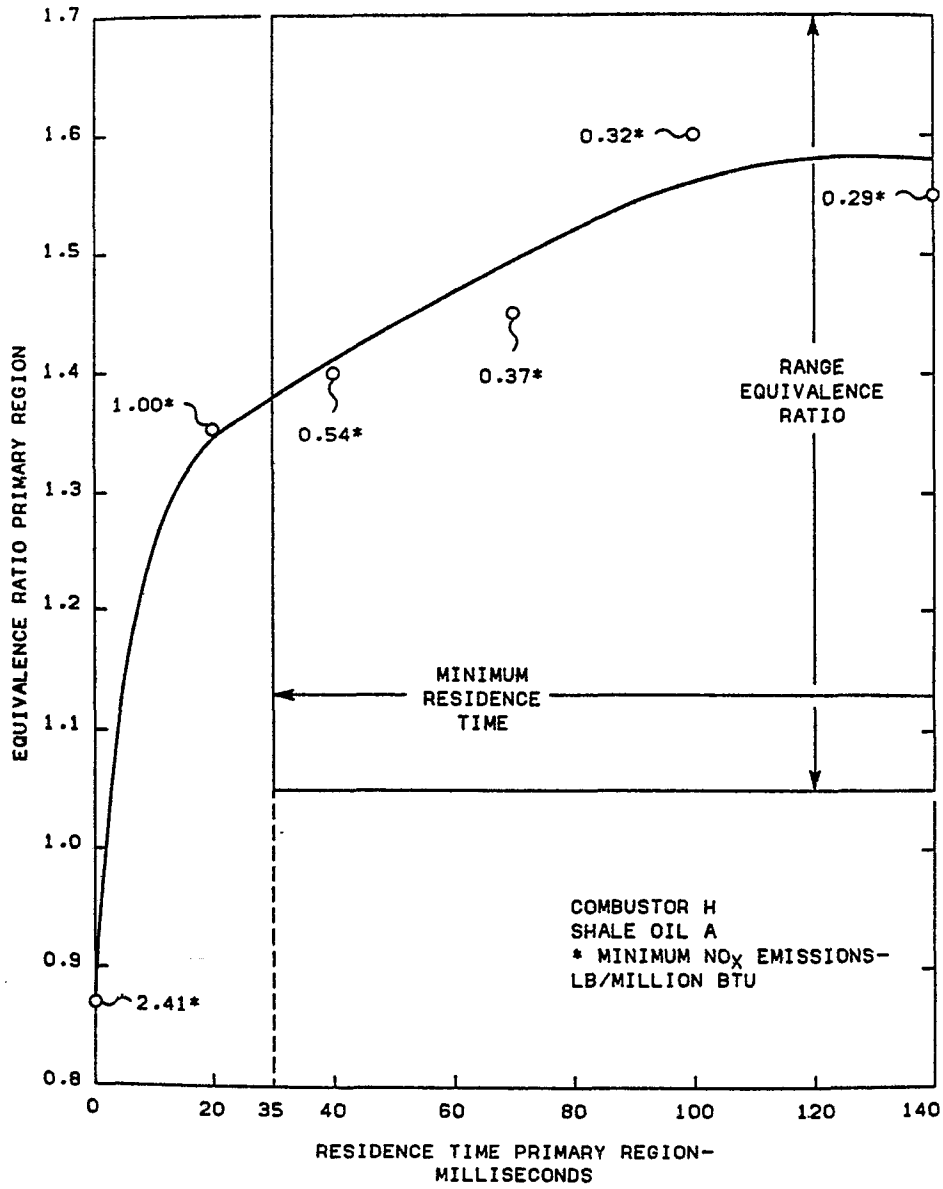


FIG. 64

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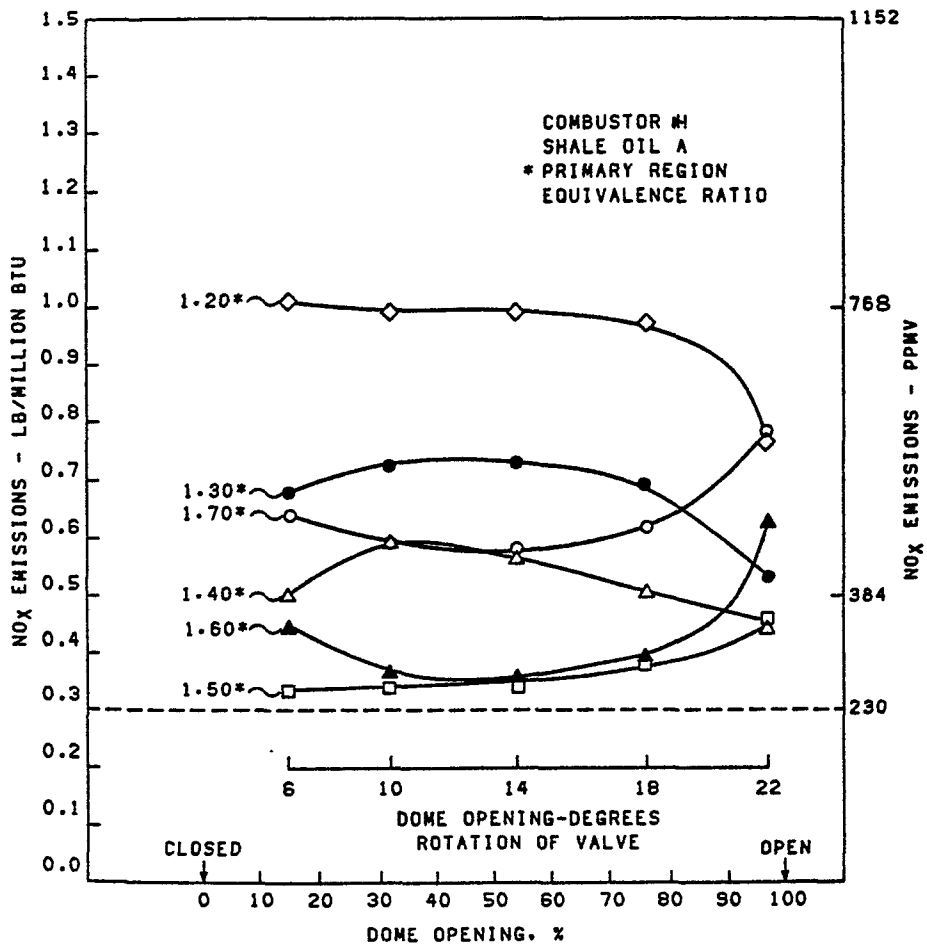


FIG 65

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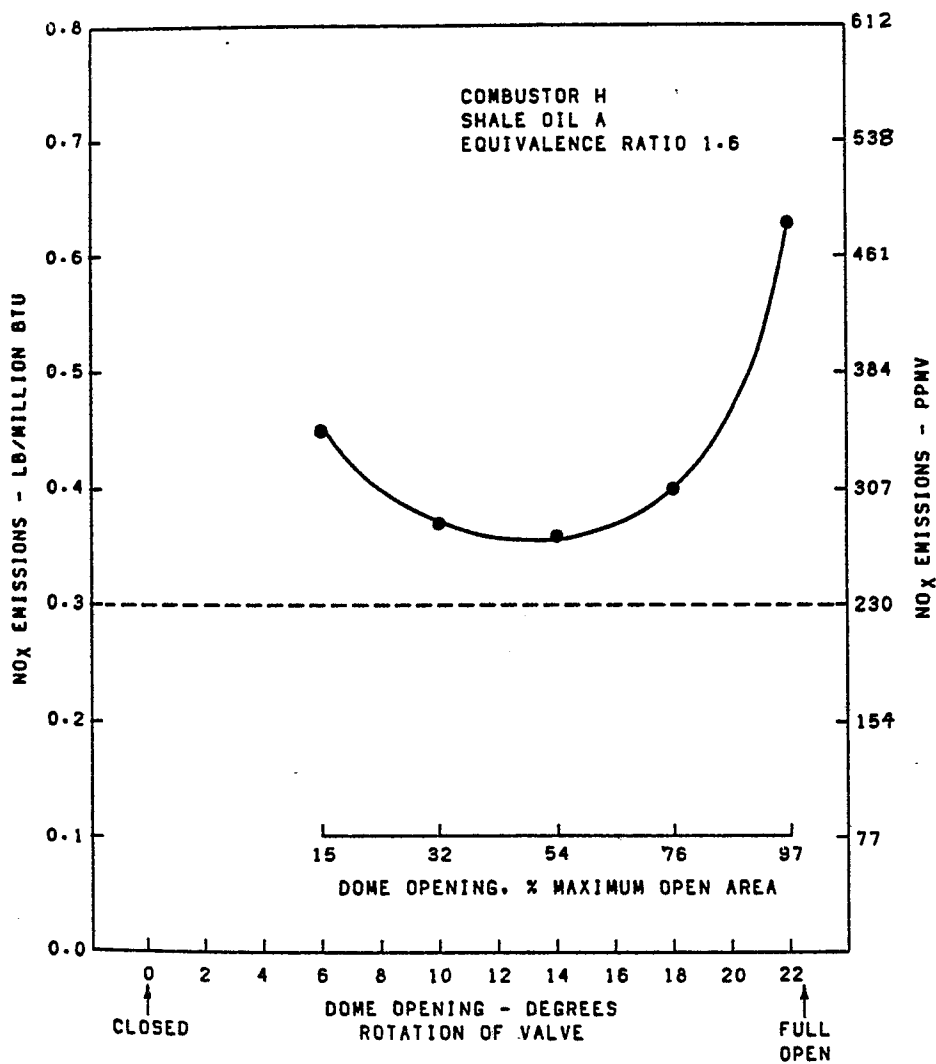


FIG. 66

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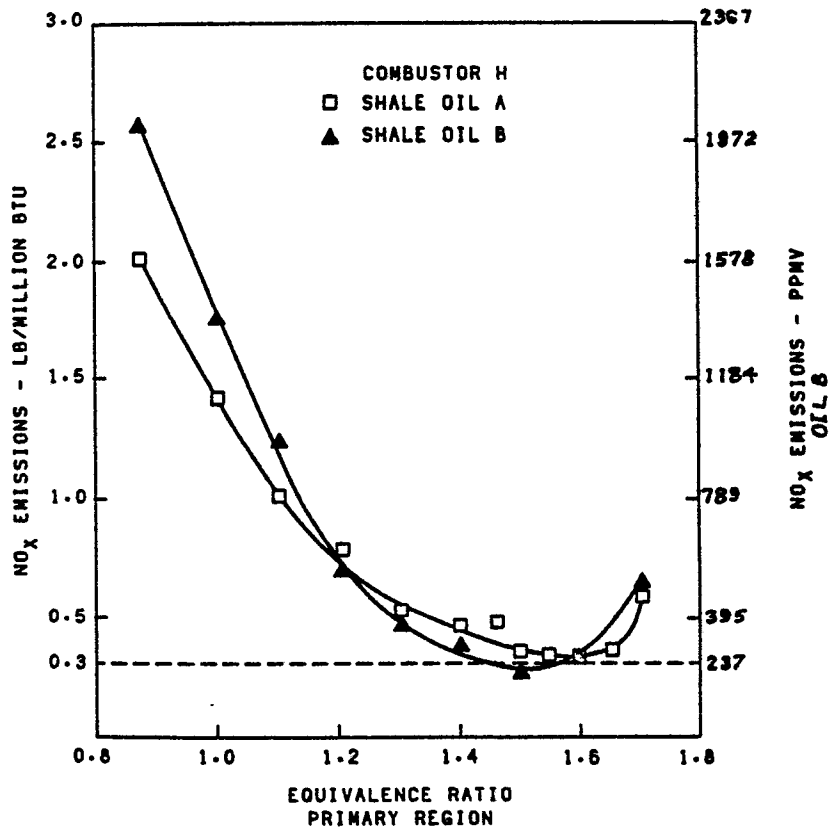


FIG. 67

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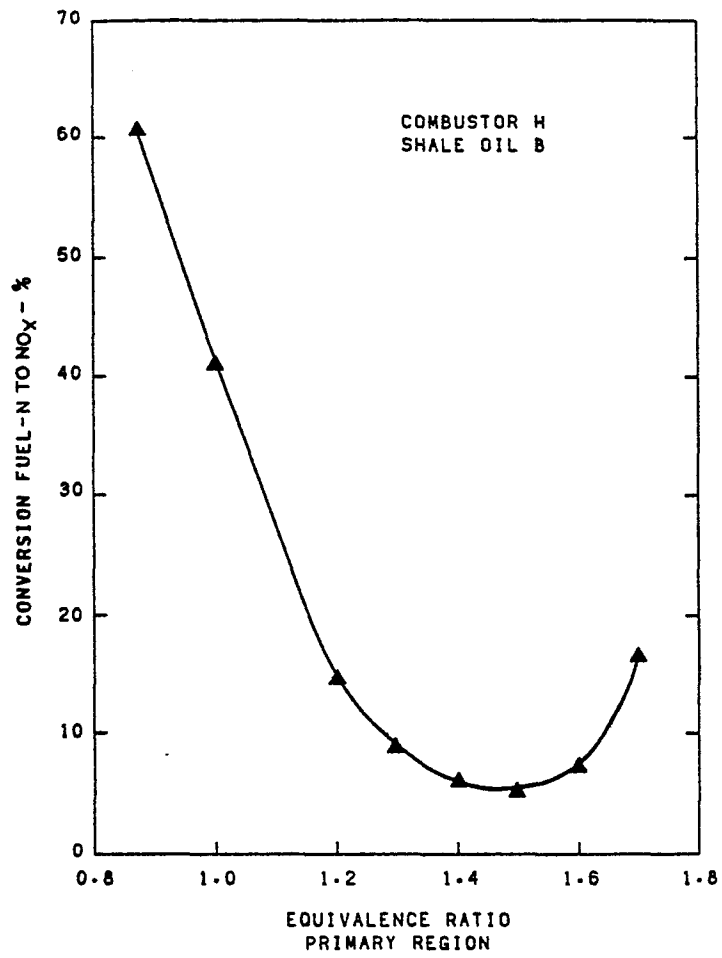


FIG. 68

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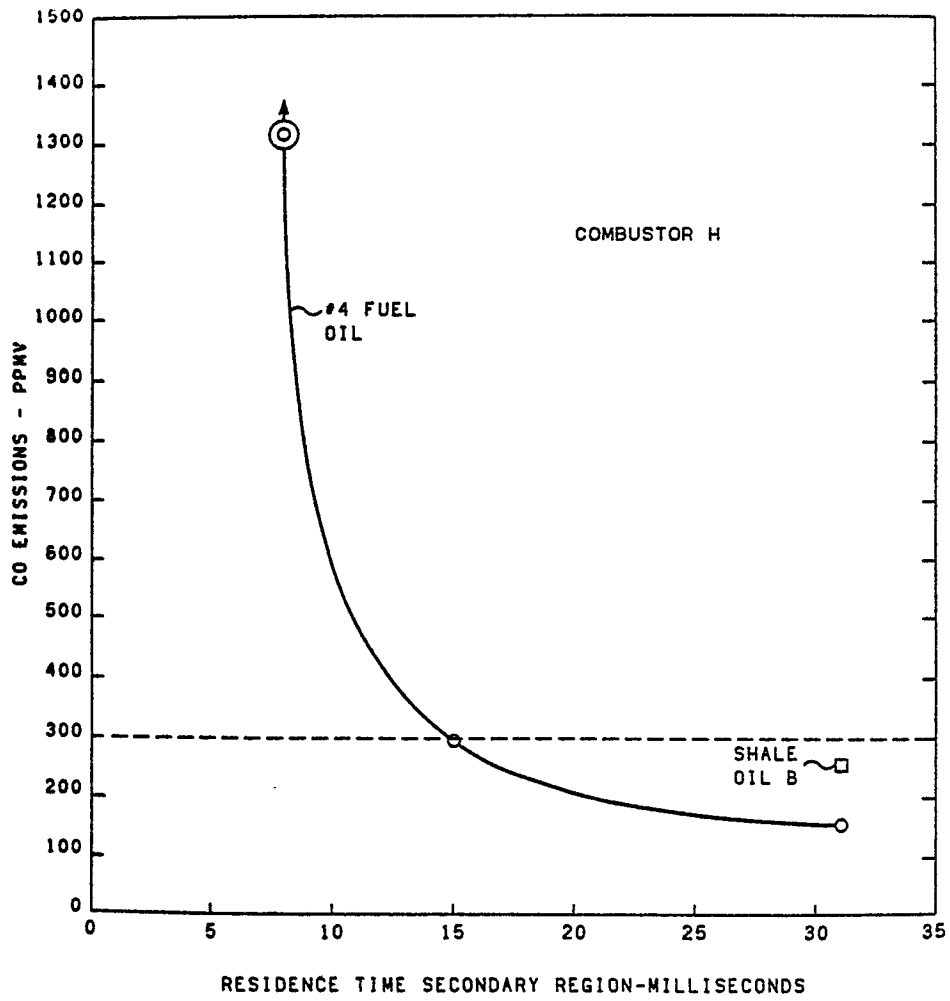


FIG. 69



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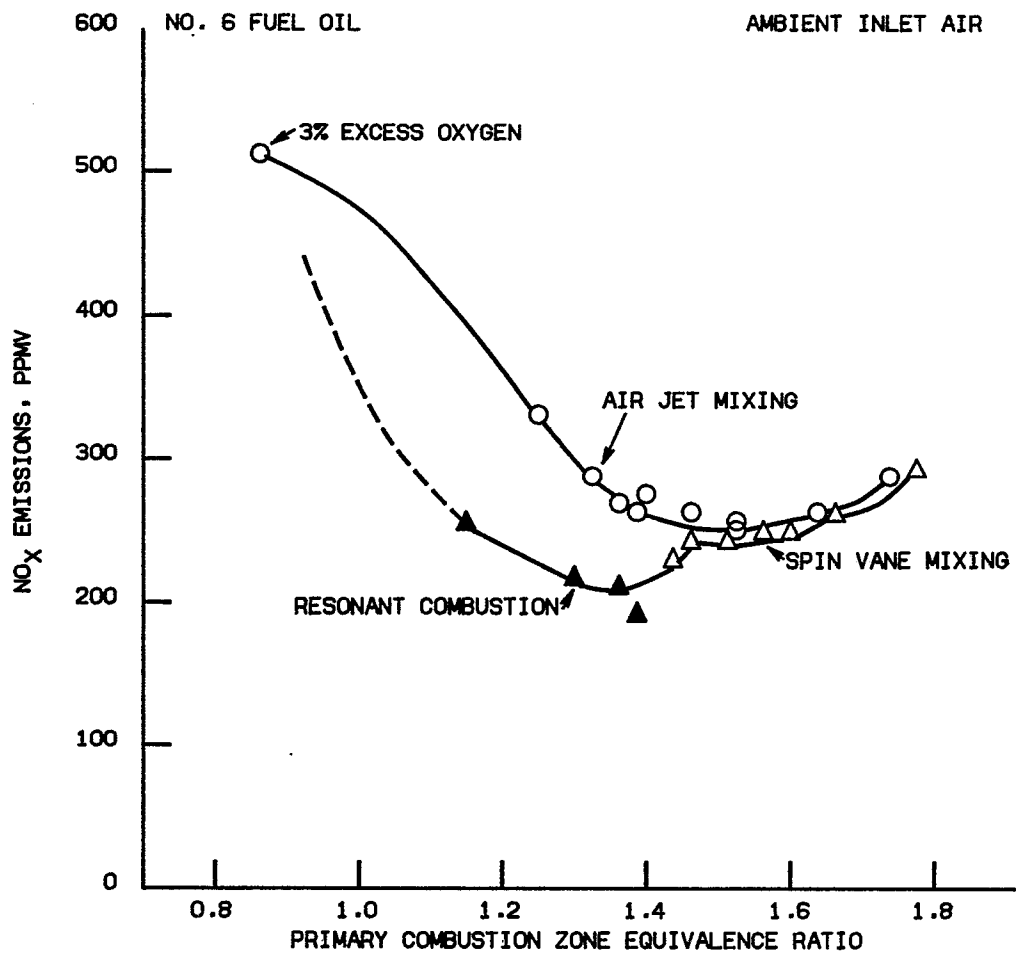


FIG. 70

5<sup>3</sup>/<sub>59</sub>

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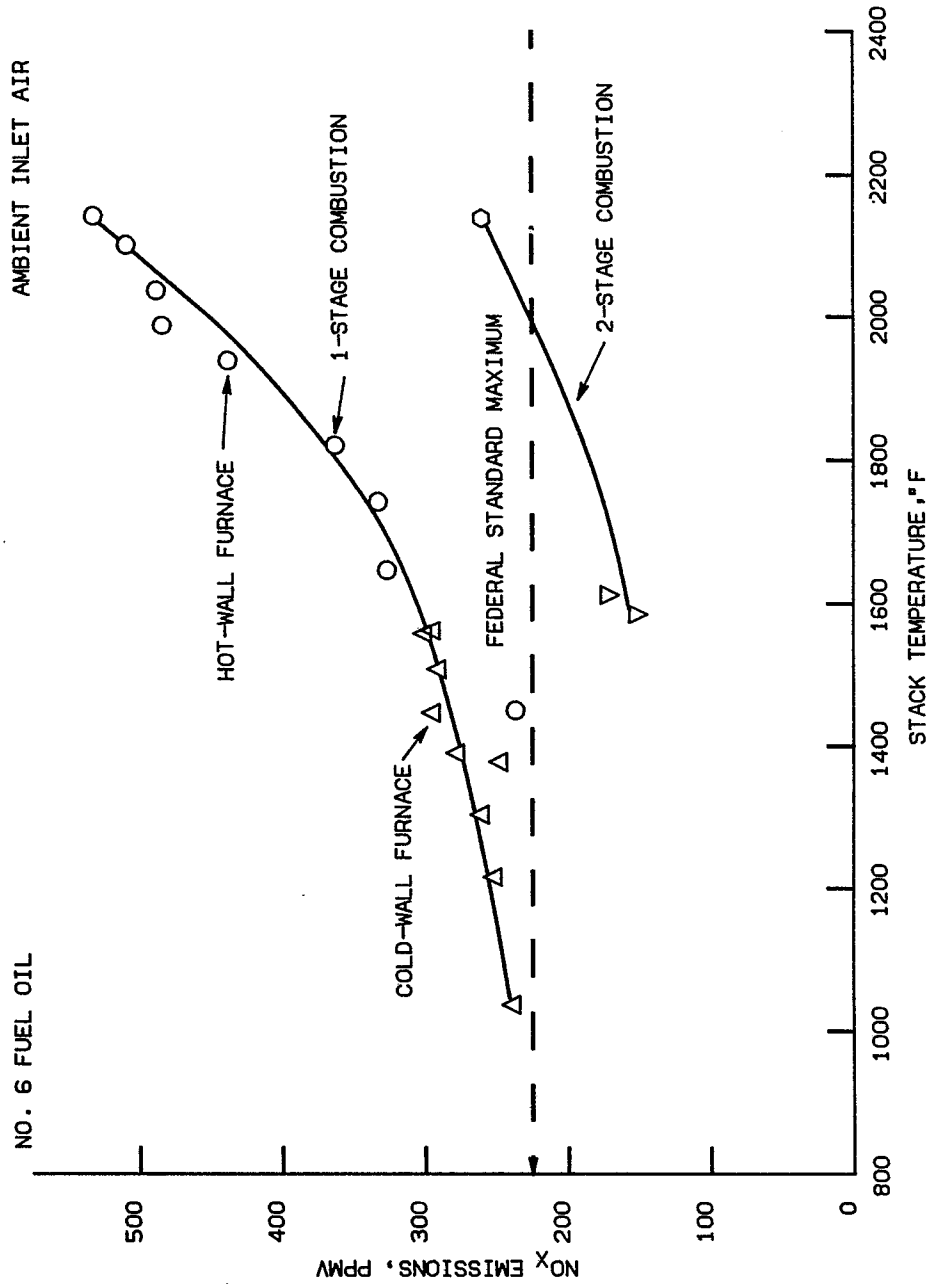


FIG 71

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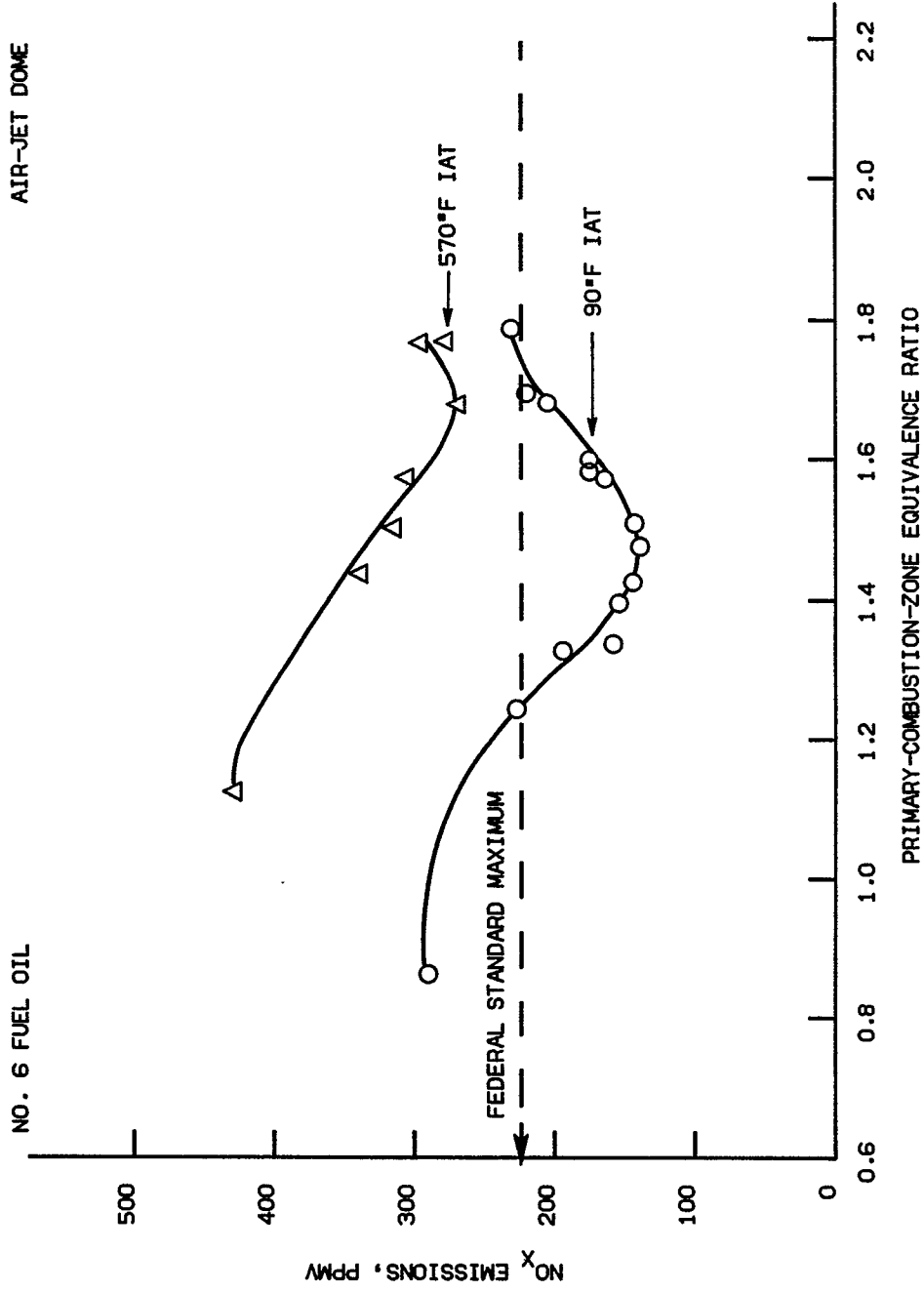


FIG. 72

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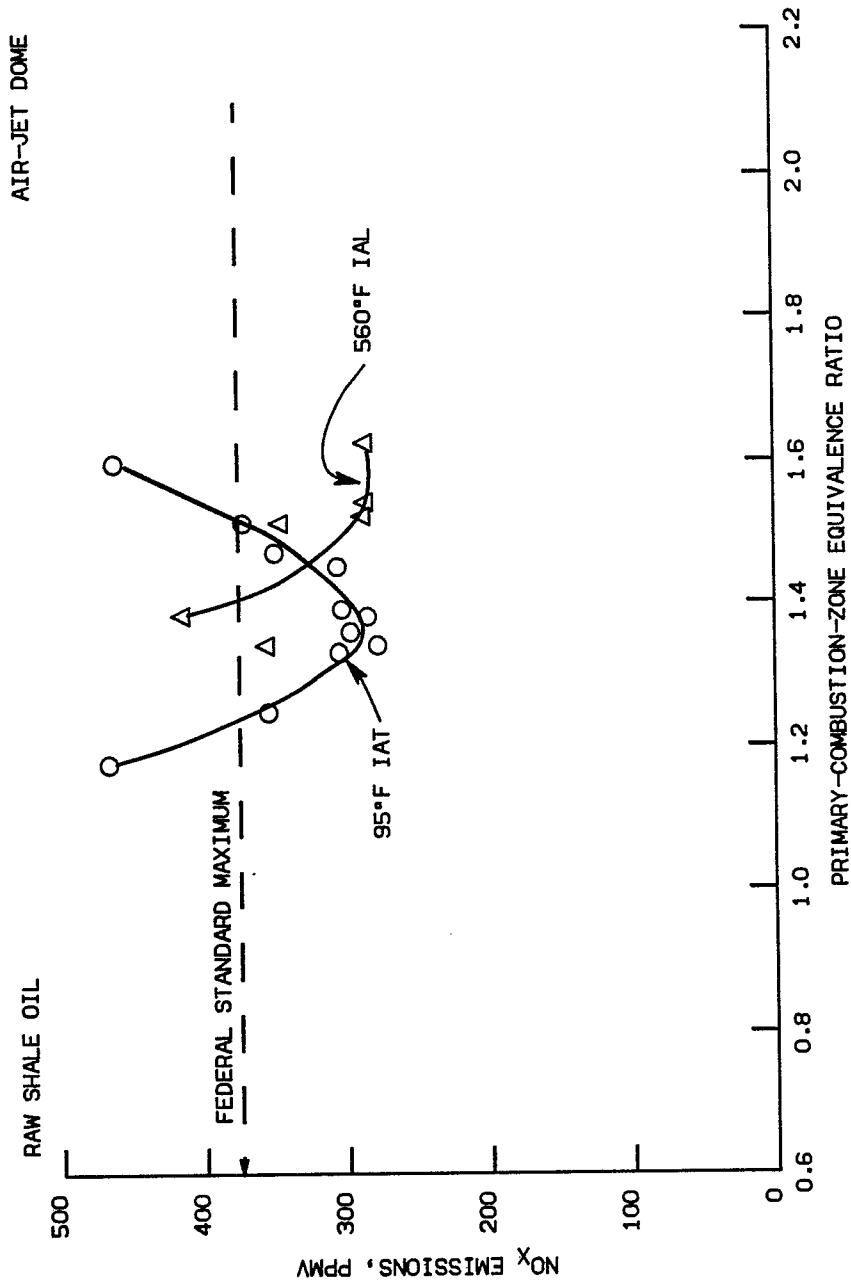


FIG. 73

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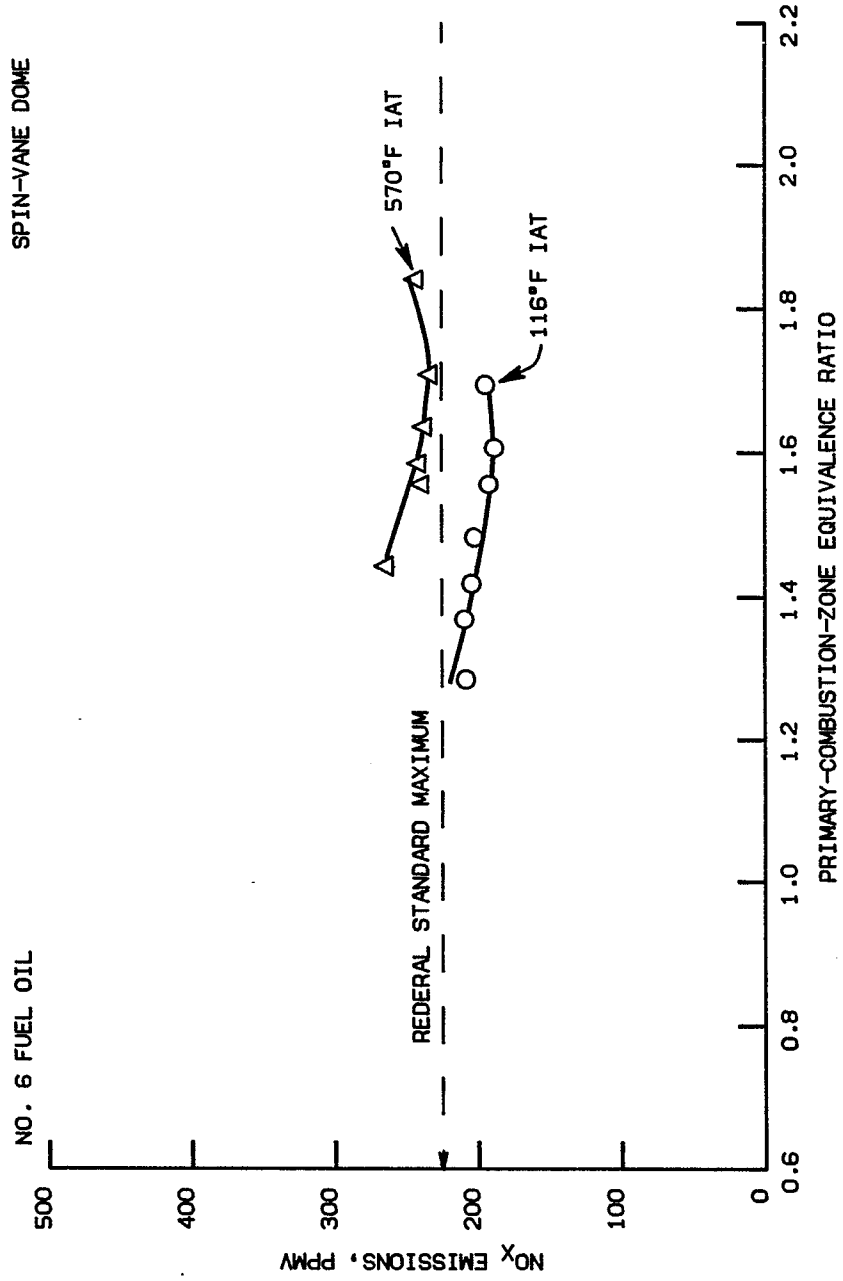


FIG. 74

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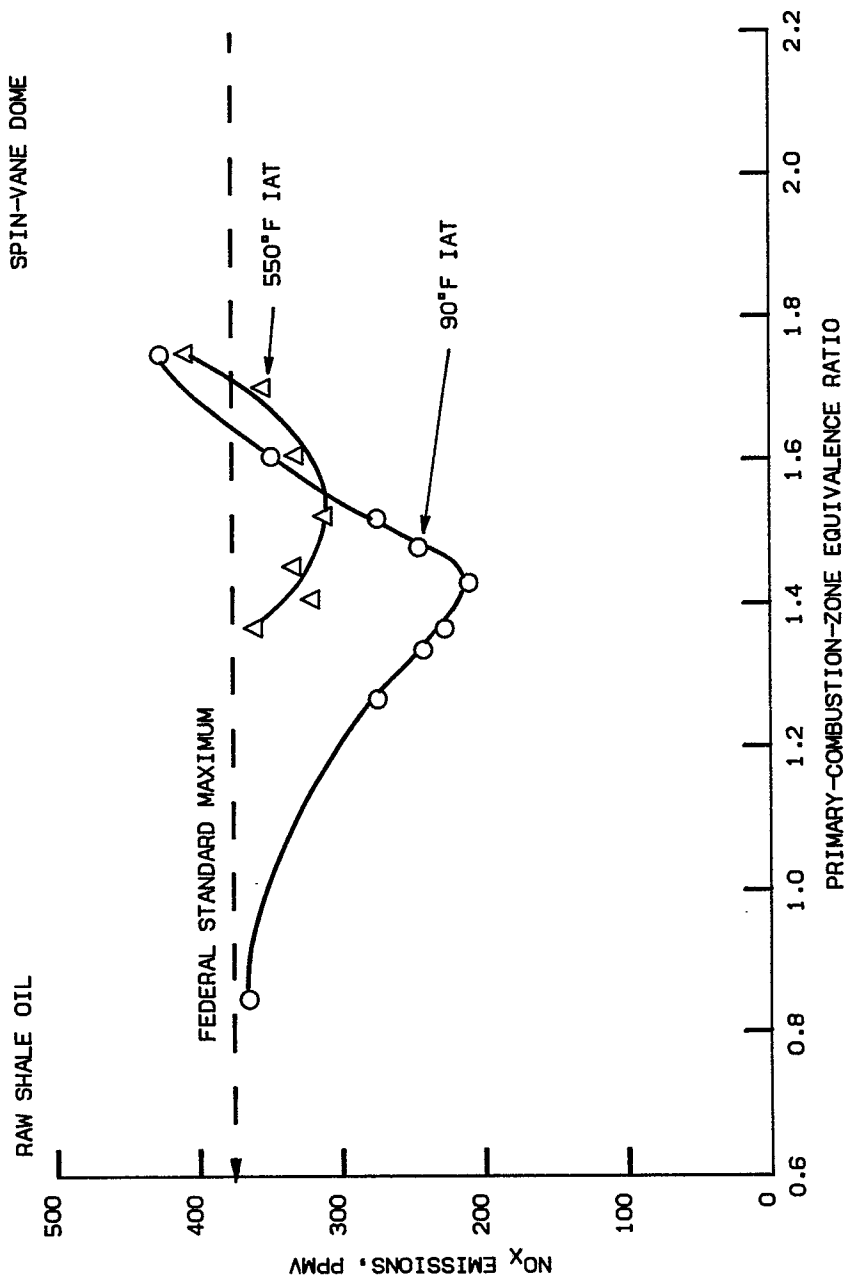


FIG. 75

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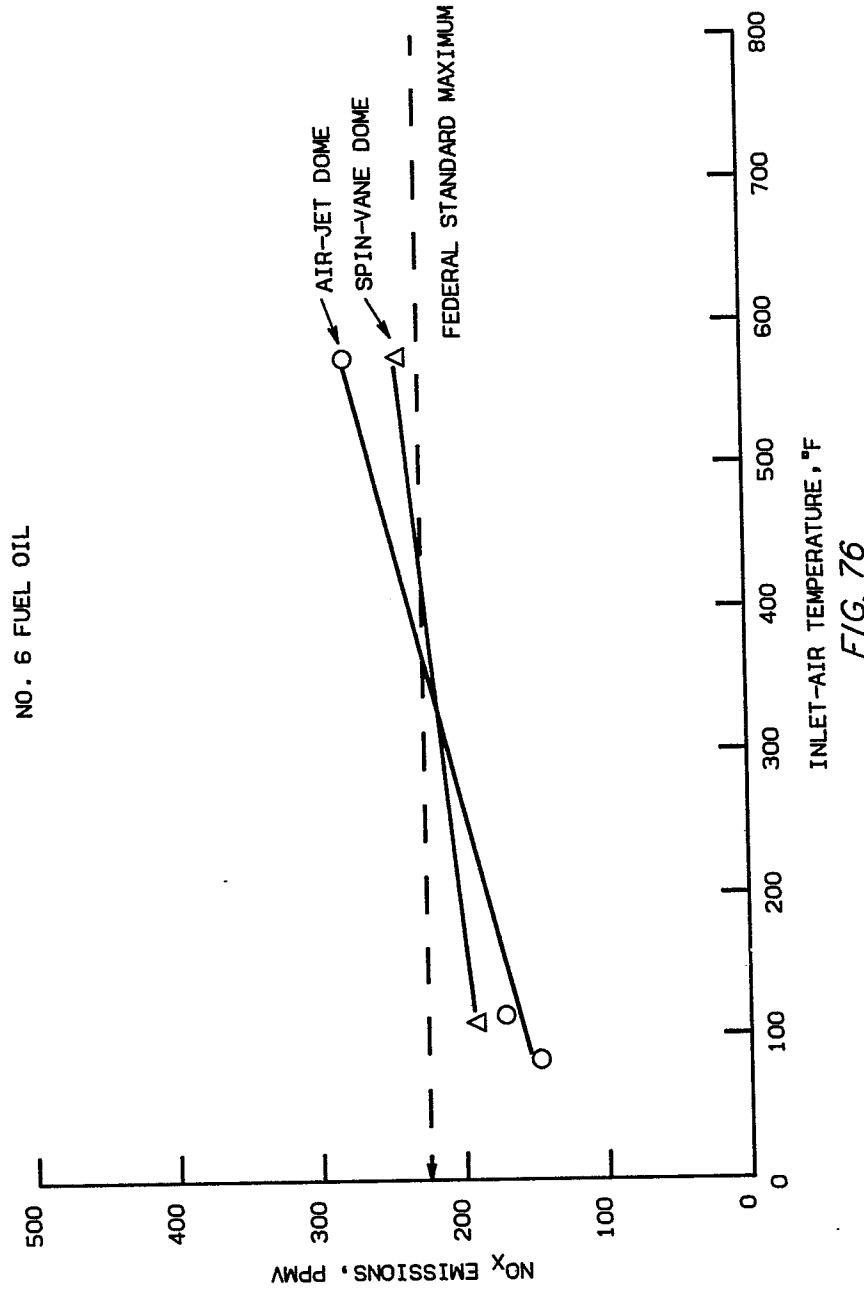
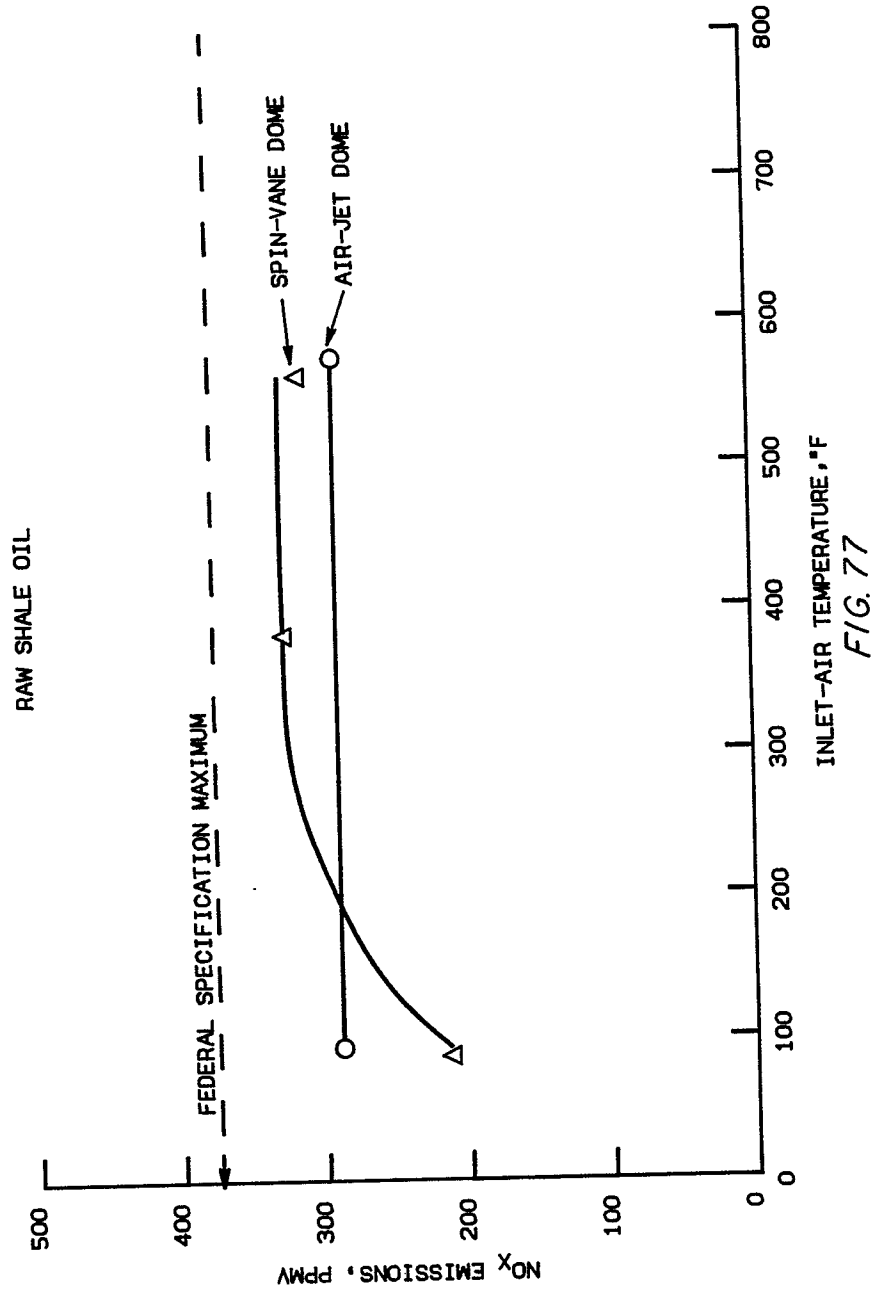


FIG. 76

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EUROPEAN SEARCH REPORT

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Application number

EP 81106796.6

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Category	Citation of document with indication, where appropriate, of relevant passages	Relevant to claim	
X	EP - A1 - 0 009 523 (EXXON RESEARCH) * Page 6, lines 17-20; page 7, lines 7-12; fig. 1 *	1,2,5-7,11,16	F 23 C 6/04 F 23 C 9/06 F 23 D 11/40
A	--	12,15	
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A	DE - B - 1 107 877 (BADISCHE ANILIN- & SODAFABRIK) * Claim 1; fig. 1 *	1,2,7,12,15	
A	CH - A5 - 591 656 (KAISHA) * Column 2, line 62 - column 4, line 2; fig. 2 * & GB-A-1 486 684 & US-A-4 054 028 --	1,2,5,7,12,15	CATEGORY OF CITED DOCUMENTS X: particularly relevant if taken alone Y: particularly relevant if combined with another document of the same category A: technological background O: non-written disclosure P: intermediate document T: theory or principle underlying the invention E: earlier patent document, but published on, or after the filing date D: document cited in the application L: document cited for other reasons
A	US - A - 3 174 530 (MEENAN) * Column 2, line 69 - column 3, line 49; fig. 1 *	1,7	&: member of the same patent family, corresponding document
X	The present search report has been drawn up for all claims		
Place of search VIENNA		Date of completion of the search 04-10-1982	Examiner TSCHÖLLITSCH



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<b>DOCUMENTS CONSIDERED TO BE RELEVANT</b>			<b>CLASSIFICATION OF THE APPLICATION (Int. Cl. 3)</b>
<b>Category</b>	<b>Citation of document with indication, where appropriate, of relevant passages</b>	<b>Relevant to claim</b>	
A	DE - A1 - 2 808 874 (NIJMEGEN B.V.)  * Claims 1, 6, 8; fig. 1, 3 * & US-A-4 255 115  -----	1,7	
			<b>TECHNICAL FIELDS SEARCHED (Int. Cl. 3)</b>