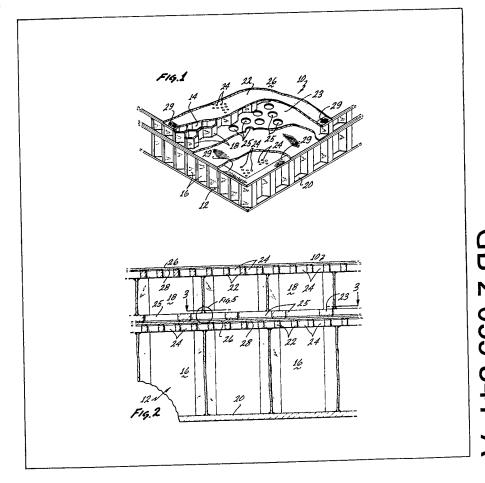
# (12) UK Patent Application (19) GB (11)

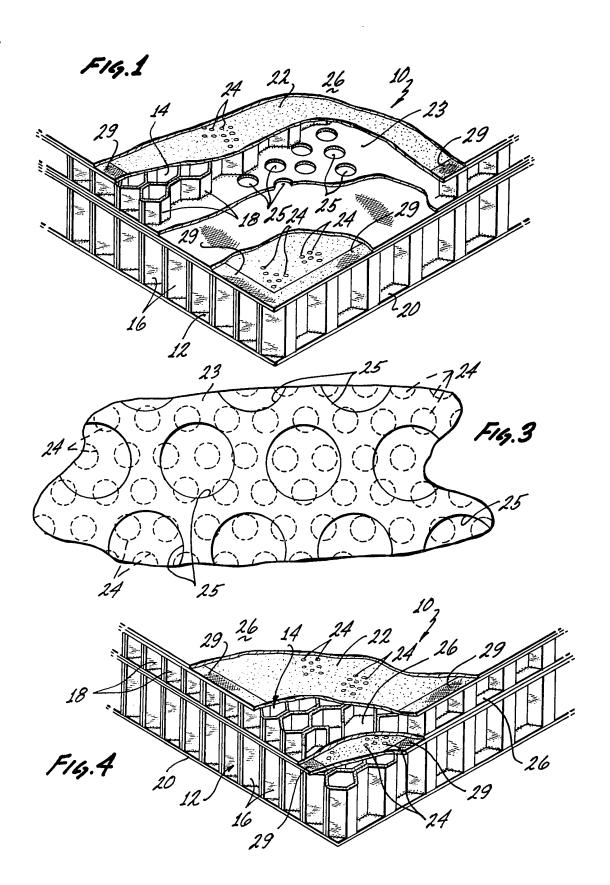
## 2 059 341 A

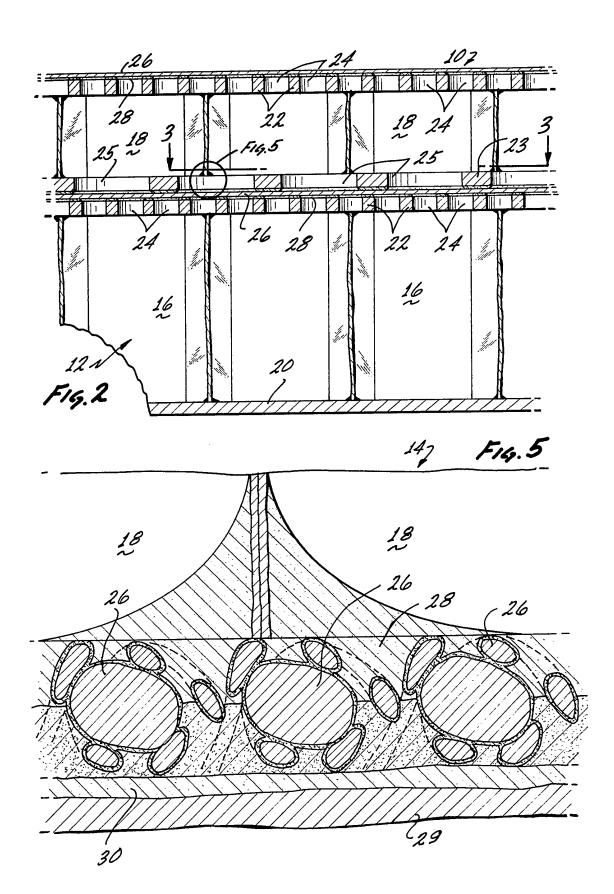
- (21) Application No 8022524
- (22) Date of filing 9 Jul 1980
- (30) Priority data
- (31) **76284 81267 103913**
- (32) 17 Sep 1979 2 Oct 1979 17 Dec 1979
- (33) United States of America
  (US)
- (43) Application published 23 Apr 1981
- (51) INT CL<sup>3</sup>
- B32B 3/12 3/24 5/22
- (52) Domestic classification B5N 0312 0324 0522
- (56) Documents cited None
- (58) Field of search **B5N**
- (71) Applicants
  Rohr Industries Inc.,
  P.O. Box 878,
  Chula Vista,
  California 92012,
  United States of America.
- (72) Inventors
  William Donald Brown,
  Philip Michael Rose,
  Frank James Riel.
- (74) Agents
  J. Owden O'Brien & Son,
  53 King Street,
  Manchester,
  M2 4LQ.

### (54) Double layer attenuation panel

(57) A double degree acoustic attenuation sandwich panel comprises in order an impervious facing 20 of thin sheet material adhesively bonded to a first honeycomb core 12 with end wise directed cells, a first perforate facing 22 of thin sheet material, a first thin layer 26 of porous fibrous material, an optional second perforated facing 22 of thin sheet material, a second honeycomb core 14 with end wise directed cells, a third perforate facing 23 of thin sheet material, the second perforated facing sheet having substantially larger perforations than the first and third sheets, and a second layer 26 of porous fibrous material. The cells of the first and second honeycomb cores may be of equal or difference volume and may be constructed of similar or dissimilar materials.







#### **SPECIFICATION**

### Double layer attenuation panels and methods of manufacture

This invention relates to a new improved noise attenuation structure, more particularly in one embodiment to a structure having a first impervious facing sheet adhered to one surface of a first 10 honeycomb core, a first perforate facing sheet with a first layer of porous fibrous material adhered to the other surface of the first honeycomb core, a second perforated facing sheet adhered to the porous fibrous woven material, the opposite surface of the 15 second perforated sheet adhered to one surface of a second honeycomb core and the other surface of the second honeycomb core adhered to a third perforated facing sheet with a second layer of porous fibrous material sedcured to the outside or opposite 20 surface of the third perforated sheet. In another embodiment of the noise attenuation structure the second perforated facing sheet is omitted. A continuous communication at a predetermined flow resistance between the cells of the first and second 25 cores and the outer surface of the second layer of porous fibrous woven material which is positioned adjacent the noise to be attenuated is maintained. The attenuation structure is specifically suitable for use in a severe environment, such as, high speed 30 gas flow surfaces of modern aircraft. The cells of the first and second cores may be of the same or different volumes to attenuate various different noise frequencies.

In manufacturing sound attenuation metal hon35 eycomb sandwich panels which are exposed to the
above mentioned extreme environment while exposed to the sound produced by modern turbine
aircraft engines, it is common practice to provide a
cellular structure utilizing the Helmholtz resonating
40 cavity principle, wherein a first thin imperforate
sheet of material is bonded to one core surface of a
sheet of cellular core material and a thin perforate
sheet of material is bonded to the opposite core
surface.

Panels of this type of construction although satisfactory for a certain degree of sound attenuation for a narrow range of sound frequencies, are found to be inefficient noise attenuation structure both for level of attenuation and for broad band noise
 frequencies customarily encountered in and around modern aircraft jet engines. Additionally it has been found that the perforations through the perforated sheet exposed directly to a high speed flow of air thereacross creates some turbulence to that air flow and the acoustic performance is generally effected by the grazing air flow.

Other concepts have been included interposing a sheet of fibrous material between the perforated sheet and the core surface. This has proven to be unsound acoustically when used adjacent to high speed gas flow.

Attempts to successfully manufacture this and various other adhesive bonded sandwich sound attenuation material of this general type have re65 sulted in the adhesive used for the bonding to ooze

or wick i.e. pass by capillary induction into the perforations and at least partially filling some of these perforations reducing the effective open area which increases the flow resistance between the sound source and the resonating cavities formed by the core cells. When the number and size of the perforations are increased to overcome this deficiency, the structural strength is reduced and air flow

turbulence is increased. In those structures where
75 porous fibrous material is utilized within the sandwich between the outer perforate surface and the
central core, the adhesive is found to wick by
capillary action into the pores and around the fibers
of the porous fibrous material as well as the
80 perforations through the perforated material further

80 perforations through the perforated material further reducing the sound attenuation effectiveness of the resulting structure.

It is the primary object of this invention to produce a double degree highly efficient sound attenuation 85 structure wherein the perforations and the porous fibrous material is substantially free of any adhesive materials.

Another object of this invention is to produce a sound attenuation panel wherein the flow resistance 90 from the outer surface to the stacked honeycomb core cells can be predicted and controlled.

Still another object of this invention is to provide a double degree sound attenuation material that has structural integrity when utilized in a severe environment for sound attenuation.

A still further object is to provide an adhesive bonding medium that provides isolation between dissimilar metals and provides a funneling effect between the perforated sheets and their adjacent sheets of porous fibrous woven material which results in an effectively greater open area.

These and other objects and advantages of the invention will become better understood by reference to the following detailed description when considered together with the accompanying drawings, in which:

Figure 1 is a perspective view of an embodiment of the sound attenuation sandwich panel of the instant invention.

110 Figure 2 is a fragmentary vertical section of the sandwich panel of Figure 1.

Figure 3 is an enlarged section of Figure 1 showing the perforation size of the central perforate sheet and one of the outer perforated sheets.

115 Figure 4 is a perspective view of another embodiment of the sound attenuation sandwich panel of the instant invention.

Figure 5 is a micrograph of a section of the attenuation panel taken along line 5-5 of Figure 2 which is common with the same section of Figure 4.

Referring now to the various figures in detail, the attenuation sandwich structure 10 comprises two honeycomb cores 12, 14, each having the usual multiplicity of end wise directed cells 16, 18, a thin 125 imperforate facing sheet 20, a pair or perforated facing sheets 22, each having a multiplicity of perforations 24 of a substantial equal preselected

23 having a multiplicity of perforation 25 of substan-130 tially greater pre-selected cross-section than those of

cross sectional area, a single perforated facing sheet

Ś

sheets 22, and two sheets of porous fibrous material 26, which may be metallic fibrous felt or any other of a number of various types of fibrous material including stainless steel, graphite, nylon or the like.

5 In an alternate attenuation sandwich structure 10, the perforated facing sheet 23 is omitted (see Figure 4). For some specific application, as in the preferred embodiment, a woven material, such as Dutch Twill Weave or the like is preferred with the crossover contacts of the various strands either bonded together or left unbonded. The perforated facing sheet 23 provides strength to the assembly and yet, due to the enlarged perforations, does not significantly affect the acoustic properties.

The preferred materials for the cores and the 15 facing sheets when utilized in a severe environment, such as an aircraft engine where weight and strength are critical requirements, is generally aluminum due to its weight versus strength characteristics. Other 20 metals or materials may be used where the requirements differ. The honeycombe core may be constructed of phenolic, plastic, paper, Kevlar™, or materials having the same or similar properties. Core 12 of the double degree system may be 25 constructed of aluminum, for instance, the core bonded to the imperforate sheet 20, to provide strength to the ultimate sandwich structure 10 and the other core 14 could be of a nonstrength providing material as those materials listed above. The cell 30 size (volume) may differ between cores 12, 14 or be of equal size depending on the sound frequency of

attenuation interest. The perforated facing sheets 22 are perforated with a plurality of small perforations 24, their size, 35 for example, could range from 0.035 to 0.065 inches in diameter. The perforated sheet 23 is perforated with a plurality of perforations 25 substantially larger in diameter, their size, for example, around 0.1875 inches. The perforations 24 provide a range from 40 10% to 50% actual open area to the perforated sheets 22 while the perforations 25 provide approximately 51% open area to perforated sheet 23 (see Figure 3 for comparative perforation sizes). The perforations may be punched, drilled, or chem milled through the 45 sheets. Chem milling is found to be advantageous because the finished perforation cross sectional area can be predetermined and the face surfaces do not require deburring, grinding, filing, polishing, etc., prior to their use in the sandwich structure. The 50 perforations 24 may be spaced, for example, at 0.081 inch intervals, for sheets 22 and 0.250 inch intervals for sheet 23 and as an example, perforations 24 may be spaced in rectangular patterns while perforations 25 may be spaced apart in a triangular pattern. 55 Various other spacing intervals or patterns may,

however, be used equally as well to practice this invention. The imperforate sheet 20 provides a bottom closure for the cells 16 and 18 of the adjacent honeycomb cores 12 and 14 which is the lower 60 cavity wall of the combined double degree attenua-

60 cavity wall of the combined double degree attenuation panel utilizing the above mentioned Helmholtz cavity resonant principle.

A first thin sheet of stainless steel woven mesh material 26 is adhesively bonded by a layer of 65 adhesive 28 to one surface of the perforate sheets 22 and the woven mesh surface of one sheet 22 is likewise bonded to one side of perforated sheet 23. The adhesive typically used is a moderate temperature curing Nitril/phenolic structural adhesive, a typical example being AF-31 manufactured by the 3M Company, U.S.A., Metal Bond 4021 manufactured by Narmco, U.S.A., FM-300 manufactured by Bloomingdale Aerospace Products, U.S.A., or any other adhesive having the same or similar characteristics as required to practice this invention. The adhesive 28 generally consists of a low solid to

istics as required to practice this invention. The adhesive 28 generally consists of a low solid to solvent solution. When the solvent is removed from these adhesive solutions by evaporation the viscosity index of the remaining adhesive is elevated.

The adhesive 28 for bonding the perforate sheets to the porous woven materials is preferably applied by spraying on the perforated sheets 22, 23, surface to be bonded a thin layer of the aforementioned adhesive. The solvent from the adhesive is then
 removed by evaporation. The surface attraction forces cause the adhesive layer around each perforation to take a smooth rounded, funnel like form when the solvent is removed.

The now substantially solid or highly viscous
30 adhesive remaining retains that funnel like form or
shape, and later during the final cure cycle it softens
sufficiently to adhere to the porous fibrous woven
material. However, the adhesive viscosity is sufficiently high so that it does not wick laterally through
55 the pores of the porous fibrous woven material.

This inherent funneling behavior of the selected adhesive effectively enlarges the openings leading into the perforations 24 at the entrance side of the core cells providing an enhanced acoustic open area.

100 An example is the use of a perforated sheet having an actual geometric open area of approximately 34% wherein the final product has an effective open area of approximately 42% or an increase of over 10%. This effective open area can only be attributed to the rounded funnel like effect around the opening into the perforated sheet created by the layer of adhesive treated in the manner prescribed. The thickness of this layer of adhesive is in the range of from 0.0005 to 0.005 of an inch.

This open area effect can be further enhanced by 110 applying a thicker layer of adhesive 28 in the range of 0.003 to 0.004 of an inch, removing the solvent, as described above and then curing the adhesive layer. The thickness of this adhesive layer can be further 115 increased by applying successive thin layers of adhesive 28 each with subsequent solvent removal and curing. Effective open areas of approximately 50% have been achieved using a sheet of perforate material 22 having an actual open area of approxi-120 mately 34%. In the buildup cure method, discussed above, an additional outer layer of adhesive 28 is required with the solvent removed but not cured prior to the panel assembly. the buildup described above is unnecessary on the sheet of perforated 125 material 23 because of the initially large diameter of perforation 25.

The layers of adhesive 28 between the perforate sheets 22, 23 and the sheet 26 of woven material obviously provides a layer of insulation between these sheets thus preventing any interaction (galva-

nic) between dissimilar adjacent metals.

When sizing of the finished panel is required to obtain a desired specific configuration by cutting, trimming, etc., or drill through holes are required, 5 the additional thickness of the adhesive buildup aids in maintaining a continued isolation between the perforate sheet and the porous fibrous woven mesh material.

Another method for providing continuous isola-10 tion between the perforate sheet and the porous fibrous woven material where sizing and/or drilling through the combined layers is required is the addition of thin layers of nonmetallic cloth material 29, for example, material made from fiberglass, 15 Kevlar™, or the like in those selected areas to be sized or drilled through. A thickness in the range of .003 to .007 of an inch is found to be satisfactory for the nonmetallic cloth even when used with single thin lavers of adhesive.

The initial step of the manufacturing process is to 20 clean the impervious and perforate sheets and the sheets of porous fibrous woven material to ensure that all surfactants have been removed to ensure satisfactory bonding.

The next step of the manufacturing process is to 25 bond the porous fibrous woven materials to one surface of their respective perforate sheets. As aforementioned, the adhesive coating for bonding the perforate sheet 14 to the porous fibrous material 30 is preferably applied by spraying on the surface of the perforate sheet a thin layer of one of the aforementioned adhesives 28 at a desired thickness. If multiple coats are desired, then the solvent is removed and the adhesive is cured between each 35 successive layer of adhesive. The final layer of adhesive, as aforementioned, has the solvent removed but is left uncured prior to the joining of the perforate sheet to the porous fibrous woven mesh material. When thin layers of nonmetallic cloth 40 material 29 are required at specific locations (see Figure 3), this material is applied after or during the application of the layer or layers of adhesive. The adhesive layer and/or layers substantially cover the perforate sheet and coat the strands of the nonme-45 tallic cloth to ensure proper bonding between the perforate sheet and the porous fibrous woven material. The solvent when present in the adhesive

is then removed from the last applied adhesive layer in instances where there is more than one layer as 50 aforementioned. After the solvent of the last applied coat is removed, the porous fibrous woven material is then placed on the coated surface and a positive force is applied between the two layers. Pressure may be applied by any well known means, such as, pressure used is generally in the range of 50 pounds

55 but not limited to a press, autoclave or the like. The per square inch.

A primer coat 30 may be applied to the perforate sheet prior to the application of the bonding adhe-60 sive to improve this bond.

To reduce or substantially eliminate the surface energy and resulting wicking the fibers of the porous fibrous woven material 26 and the adjacent surfaces of the perforated sheet 22, which it is attached, are 65 coated with a nonwetting substance, generally in

liquidous form, such as for purpose of example only, Frekote 33, a trademarked product of Frekote, Inc., U.S.A., (containing Methylene Chloride 30.7%, Trichorotrifluorethane 88.8%, Xylene 3.9%, VM and 70 P Naptha 25.1%, and Perchloroethylene 0.9%), as well as other materials having the same or similar nonwetting characteristics.

After the bond between the perforated sheet 22 and the porous fibrous woven material 24 is made, 75 the surface of the side of the perforated sheets with the perforations exposed (the side ultimately to be bonded to its respective cellular core or sheet of perforated material 23) is coated with a maskant material which prevents the nonwetting solution 80 from coating any portion of that surface. It is well known that any nonwetting (surface energy reducing) material present prevents a satisfactory bond. The maskant material of the preferred method is a sheet of heavy paper adhered to the perforation

85 exposed surface of the perforated sheet. The adhesive used to attach the heavy paper is nonsoluble when placed in contact with a nonwetting material and is sufficiently tacky to prevent permanent adherence to the perforated sheet. This adhesive however must be sufficiently secured to the heavy paper so that if completely removed with the paper leaving

the perforated surface substantially free of adhesive. It should be understood, however, that other types and forms of maskant material may be used equally 95 as well to practice this invention.

While the maskant material is secured to the perforated sheet, the porous fibrous woven material 26 and the exposed portion of the perforated sheets 22 are saturated with the selected nonwetting solu-100 tion, such as the aforementioned Frekote 33, wherein all of the exposed fibers of the porous fibrous woven material and the exposed surfaces, of the perforated sheets are covered. This nonwetting material is then allowed to dry leaving the contacted 105 surfaces covered.

After the nonwetting material is dry, the maskant material of the type described is then removed by peeling off the heavy paper with the adhesive attached thereto from the surface of the perforated sheet. Although the adhesive generally used is completely removed from the surface of the perforated sheet any remaining residue should be removed prior to the final assembly of the components into double layer sandwich attenuation panel.

Referring now to Figures 1, 2 and 4 specifically, the 115 various components are then stacked in assembly order. A layer of FM150 or an epoxy adhesive having the same or similar characteristics is applied between the imperforate sheet and the honeycomb 120 core 12 and the perforation exposed surface of the perforated sheet 22. A layer of the same adhesive is then applied between the outer surface of the porous fibrous woven material 26 and one surface of honeycomb core 14 or both surfaces of perforated 125 sheet 23. It should be noted that the nonwetting

substance remains on the fibers of the porous fibrous woven material 26 and the adjacent surface of the perforated material 22 may include the walls of the perforations of the combined sheet to be

130 bonded to core 14. Although it is well known that the

FM150 or the like will not adhere to a surface coated with a nonwetting material, it has been found that the cell edges force the adhesive to penetrate well into the fibers of the porous fibrous material 26 5 adjacent thereto (see Figure 5). Although the adhesive penetrates below and adjacent to the cell edges, it does not wick into the open surface areas of the porous fibrous woven material nor the perforations of the perforated sheet or sheets because of the 10 coating of nonwetting material. The adhesive selected does not adhere to the fibers, as mentioned above, however, the adhesive does flow around these coated fibers and form mechanical bonds around the fibers and with the layer of AF31 or the 15 like adhesive. The bonds are structually sufficiently strong to maintain physical integrity between the

porous fibrous woven material 22 and the cell edges of core 14. This occurs when either a perforate sheet 23 is positioned between the core 14 and porous 20 fibrous material 26 or is omitted.

It should be understood that in the manufacturing of the double layer attenuation panel of the instant invention when an adhesive bond is desired between the fibers of the porous fibrous material and 25 the cells of core 14 the steps of applying the nonwetting solution and maskant are omitted.

A layer of FM150 or the like is then applied between the other surface of the honeycomb core 14 and the perforation exposed surface of the other 30 perforated sheet 22.

Pressure in the range of 50 pounds per square inch is then applied between the imperforate sheet 20 and the outer surface of the exposed porous fibrous woven material 26 toward the center of the sandwich 35 structure. As aforementioned, this pressure may be applied by any appropriate means as herein dis-

In some instances it is preferable to cure the adhesive at an elevated temperature.

The cured attenuation panel 10 is now ready to be sized as required and placed into use.

Although the foregoing invention has been described in some detail by way of illustration and example for purposes of clarity of understanding, it 45 should be understood that certain changes and modifications may be practiced within the spirit of the invention as limited only by the scope of the appended claims.

What is claimed as new and useful is:

#### **CLAIMS**

50

1. A double degree acoustical honeycomb sandwich panel for use in the severe environment in and 55 around high velocity fluid flow paths comprising:

a first and second honeycomb core having a multiplicity of endwise directed cells with walls of thin sheet material disposed transversely to the panel;

an imperforate facing of thin sheet material; 60 a first and second perforated facing of thin sheet material; and

a first and second thin sheet of porous fibrous material;

said impervious facing of thin sheet material is

adhesively bonded to one end wise cell face of said first honevcomb core;

said first perforated facing of thin sheet material is adhesively bonded to said first honeycomb core on 70 the surface opposite said impervious facing of thin sheet material;

said first thin sheet of porous fibrous material is adhesively bonded to the other surface of said first perforated facing of thin sheet material;

one end wise cell surface of said second hon-75 eycomb core is bonded to the other surface of said first thin sheet of porous fibrous material;

said second perforated facing of thin sheet material is adhesively bonded to the opposite surface of 80 said second honeycomb core;

said second thin sheet of porous fibrous material is adhesively bonded to the opposite surface of said second perforated facing of thin sheet material.

2. A panel as defined in claim 1, wherein a third perforated facing of thin sheet material is positioned between said end wise cell surface of said second honeycomb core and the outer surface of said first thin sheet of porous fibrous material.

3. The invention as defined in claim 1 or 2, 90 wherein the adhesive material for bonding together said perforated facings of thin sheet material and porous fibrous material is of sufficient thickness to form an insulation therebetween.

4. The invention as defined in claim 1 or 2, wherein nonmetallic material is positioned between said first and second perforated facings of thin sheet material and porous fibrous material in selected areas where sizing and drill through holes are required for enhancing the insulation therebetween.

5. The invention as defined in claim 2, wherein 100 said first and second perforated facings of thin sheet material have perforations of substantially equal cross sectional size, said third perforated facing sheet has perforations of greater cross sectional size 105 than said first and second perforated facing sheet.

6. A method of manufacturing adhesive bonded acoustical attenuation structure, the structural components comprising a first and second honeycomb core with a multiplicity of end wise directed cells, an 110 imperforate facing sheet, a first and second perforated sheet, a first and second thin sheet of porous fibrous material, said method comprising the steps of:

(a) cleaning and degreasing the components to 115 be assembled into said acoustical attenuation struc- .

(b) applying a solvent base first adhesive coating to one surface of each of the perforated sheets for bonding to one of the first and second thin sheets of 120 porous fibrous material;

(c) removing the solvent from said solvent base first adhesive;

(d) stacking said first perforated sheet with said first thin sheet of porous fibrous material and said 125 second perforated sheet with said second thin sheet of porous fibrous material applying a positive pressure between the components of each stack and curing said first adhesive;

(e) applying a layer of a second adhesive be-130 tween each core and its adjacent components; and

5

- (f) stacking the various components in the following sequence, first said imperforate facing sheet, then said first honeycomb core, then the now combined first perforated sheet and the first thin
  5 sheet of porous fibrous material with the perforated surface adjacent the open core of said first honeycomb core, said second honeycomb core positioned on said first thin sheet of porous fibrous material and then the combined second perforated
  10 sheet and second thin sheet of porous fibrous material with said second thin sheet of porous fibrous material as the outermost surface, applying a positive pressure between the outer components toward the center of the stacked components and
  15 curing the first and second adhesives.
- 7. The method of claim 6, wherein there is additionally provided a third perforated sheet and said third perforated sheet is positioned between first thin sheet of porous fibrous material and said 20 second honeycomb core.
  - 8. The method as defined in claims 6 or 7 wherein the following additional steps are included between steps (d) and (e),

applying a coating of a maskant material to the 25 perforation exposed surfaces of the now combined perforated sheets and thin sheets of porous fibrous material,

applying an anti-wetting solution to the fibers of the thin sheets of porous fibrous material and the 30 adjacent attached surface of the perforated sheets, and

removing the coating of maskant material from the perforation exposed surfaces of the now combined perforated sheets and thin sheets of porous 35 fibrous material.

- 9. The method of manufacture as defined in claims 1 or 2 including additional step of adding at selected locations sections of non-metallic cloth material between steps (b) and (c).
- 40 10. A method of manufacturing multi-cell acoustic attenuators having two or more intra bonded core structures, substantially as described with reference to and as illustrated in the accompanying drawngs.
- An acoustic attenuator panel manufactured
   by a method claimed in any of claims 6 through 10.
  - 12. An acoustic attenuator panel substantially as described with reference to and as illustrated in the accompanying drawings.

Printed for Her Majesty's Stationery Office by Croydon Printing Company Limited, Croydon, Surrey, 1981. Published by The Patent Office, 25 Southampton Buildings, London, WC2A 1AY, from which copies may be obtained.